

Studies on thermal-hydraulic characteristics of supercritical CO₂ flows with non-uniform heat flux in a tubular solar receiver

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Abstract

Supercritical CO₂ (SCO₂) Brayton cycle is one of the best approaches for concentrated solar power tower, with tubular solar receiver being one of the most attractive options to directly heat SCO₂. The superposition of drastically variable properties of SCO₂ and uneven distribution of solar irradiance flux poses a great challenge to the design and optimisation of solar receiver, thus the studies on thermal-hydraulic characteristics of SCO₂ in solar receiver become crucial to deal with the challenge. In this study, the thermal-hydraulic characteristics of SCO₂ in a vertical receiver tube under axially and circumferentially non-uniform heat-flux conditions are numerically studied with average heat flux ranging from 150 kW·m⁻² to 210 kW·m⁻² and mass flux ranging from 400 kg·m⁻²·s⁻¹ to 700 kg·m⁻²·s⁻¹, and the influences of buoyancy on the flow and heat transfer characteristics are evaluated. The non-uniform distribution of the axial heat flux leads to 1.2% to 20.5% increase in the overall heat transfer coefficient, while more serious local heat transfer deterioration is observed. The buoyancy effect changes the flow structure, resulting in suppression of turbulence intensity and formation of thick momentum and thermal boundary layers, which is closely related to the severe heat transfer deterioration of SCO₂ under non-uniform heat-flux conditions. Bo^* could well predict the buoyancy effect in a vertical tube under non-uniform heat-flux conditions. Compared with upward flow, downward flow could effectively alleviate heat transfer deterioration and reduce wall temperature gradient especially for axially non-uniform heat-flux condition. The present work could provide guideline for the design and optimisation of tubular solar receiver and solar power system with SCO₂ as heat transfer fluid.

Key words: Supercritical CO₂ (SCO₂); non-uniform heat-flux condition; solar tower power; thermal-hydraulic characteristics; solar receiver.

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31 Nomenclature

a	acceleration, $\text{m}\cdot\text{s}^{-2}$
Bo^*	non-dimensional buoyancy parameter,
c_p	specific heat at constant pressure, $\text{kJ}\cdot\text{kg}^{-1}\cdot\text{K}^{-1}$
D	diameter, mm
g	gravity, $\text{m}\cdot\text{s}^{-2}$
G	mass flux, $\text{kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$
Gr^*	Grashof number
h	heat transfer coefficient, $\text{kW}\cdot\text{m}^{-2}\cdot\text{K}^{-1}$
H	enthalpy $\text{kJ}\cdot\text{kg}^{-1}$
k	turbulence kinetic energy, $\text{m}^2\cdot\text{s}^{-2}$
L	heated length or distance from the position where heating starts, m
P	Pressure, MPa
P_k	production of turbulence energy, $\text{kg}\cdot\text{m}^{-1}\cdot\text{s}^{-3}$
Pr	Prandtl number
q	heat flux, $\text{kW}\cdot\text{m}^{-2}$
r	radial coordinate, mm
R	tube radius, mm
Re	inlet Reynolds number
T	temperature, K
T_{pc}	pseudocritical temperature, K
v	velocity, $\text{m}\cdot\text{s}^{-1}$
x, y, z	x, y, z directions
y^+	non-dimensional distance from the wall

Greek symbols

β	volume expansion coefficient, K^{-1}
θ	circumferential angle, degree
λ	thermal conductivity, $\text{W}\cdot\text{m}^{-1}\cdot\text{K}^{-1}$

μ	dynamic viscosity, Pa·s
μ_t	turbulent viscosity, Pa·s
ν	kinematic viscosity, m ² ·s ⁻¹
ρ	density, kg·m ⁻³
σ	standard deviation of the Gaussian distribution
τ	shear stress, Pa
ϕ	energy dissipation, W·m ⁻³

Subscripts

ave	average
b	bulk
c	critical
i, j, l	tensor index symbols
in	inlet section
m	median
max	maximum
min	minimum
w	wall

32

33 1. Introduction

34 With the increasingly serious climate problems, reducing carbon emissions, achieving carbon neutrality and
35 controlling global warming have become the consensus in today's world. In order to achieve sustainable
36 development goals (SDGs) and Net zero targets, it is of great significance to develop highly efficient,
37 environmentally friendly energy conversion technologies. The power cycle using CO₂ as a working medium
38 has received extensive attention in recent decades. The supercritical CO₂ (SCO₂) Brayton cycle system can
39 significantly reduce compressor power consumption and system size, improve system efficiency, decrease
40 investment costs, and simplify system design [1,2]. This system has promising potentials in concentrating solar
41 systems [3], geothermal energy [4], fourth-generation nuclear power systems [5], new gas turbines and coal-
42 fired power plants [6,7], etc. Heat exchanger is a key component in SCO₂ Brayton cycle system which has a
43 significant impact on the performance and cost of the entire system. However, unlike conventional fluids,

SCO₂ drastically changes in thermophysical properties near the critical or pseudocritical point, which makes flow and heat transfer of SCO₂ unique and interesting.

Experiment studies on SCO₂ have been conducted to investigate the effects of mass flux, heat flux, pipe diameter and other operating parameters on the flow and heat transfer characteristics (also known as thermal-hydraulic characteristics) of SCO₂ in the past several decades [8,9]. Kim et al. [10,11] experimentally studied the heat transfer of SCO₂ in upward and downward flows in a tube with diameter of 4.5 mm, and they found at the conditions of $G = 489 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and $q = 85 \text{ kW}\cdot\text{m}^{-2}$ (medium wall heat flux and low mass flux), an obvious peak was observed for the wall temperature in upward flow while this peak did not appear in downward flow. Lei et al. [12] experimentally investigated the heat transfer characteristics of SCO₂ in a vertical small channel with mass flux ranging from 50 to 200 $\text{kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and heat flux ranging from 38 to 234 $\text{kW}\cdot\text{m}^{-2}$. They reported that significant difference was observed in heat transfer characteristics at different heat-flux levels in small tubes. Heat transfer of SCO₂ may deteriorate in vertical heating tubes, a criterion combined with Grashof number was proposed by Jackson et al. [13] to determine the occurrence of heat transfer deterioration. Kline et al. [14] stated that the minimum heat flux that triggered heat transfer deterioration followed the power law of mass flux in vertical tubes. In addition, Ershan et al. [15] and Xie et al. [16] reviewed the empirical correlations for heat transfer of SCO₂ in horizontal and vertical tubes under heating and cooling conditions, but no one could predict the heat transfer performance accurately under various working conditions.

Due to the difficulty of obtaining detailed information about the internal heat transfer and flow features through experiment, numerical methods have also been widely used to explore the complex heat transfer mechanism of supercritical fluids. Xiang et al. [17] and Li et al. [18] found that heat transfer deterioration was observed at the bottom region of horizontal cooled flows, while heat transfer deterioration occurred in the top region under horizontal heating conditions. Wang et al. [19] studied the buoyancy effect of SCO₂ in large horizontal tubes, and they concluded that buoyancy influenced the flow structure and turbulence levels mainly via the induced secondary flow. Guo et al. [20] used the variable turbulence Prandtl number model to study the flow and heat transfer characteristics of SCO₂ in a vertical tube, with results showing that the local heat transfer deterioration of SCO₂ was alleviated by reducing the heat flux density and pipe diameter. Zhao et al. [21] compared the heat transfer behaviour and mechanism of SCO₂ for upward and downward flows, and they found that the buoyancy effect influenced the viscous length scale and the redistribution of the velocity profiles. Some enhanced tubes such as helically coiled tubes [22] and cylindrically concaved tubes [23] were also employed to eliminate localized heat transfer deterioration and effectively improve heat transfer performance

under uniform heating conditions.

The researches on the thermal-hydraulic characteristics of SCO_2 were generally conducted under uniformly heating or cooling conditions in existing literature. However, in practical applications such as coal-fired power generation system and solar thermal power system, the heat flux distribution on the surface of the heat absorber exhibits extremely nonuniform. The schematic diagram of the structure of the solar power tower and the solar irradiance flux on the receiver focal plane are presented in Figure 1 [24]. The solar heat collection system in solar power tower is mainly composed of solar heliostat field, tower and solar receiver [25]. From the solar irradiance flux on the focal plane as shown in Figure 1(b), it is obvious that the solar flux on the surface of the receiver is non-uniformly distributed in both the axial and circumferential directions, and this brings severe challenges to the design of receiver in SCO_2 power system.

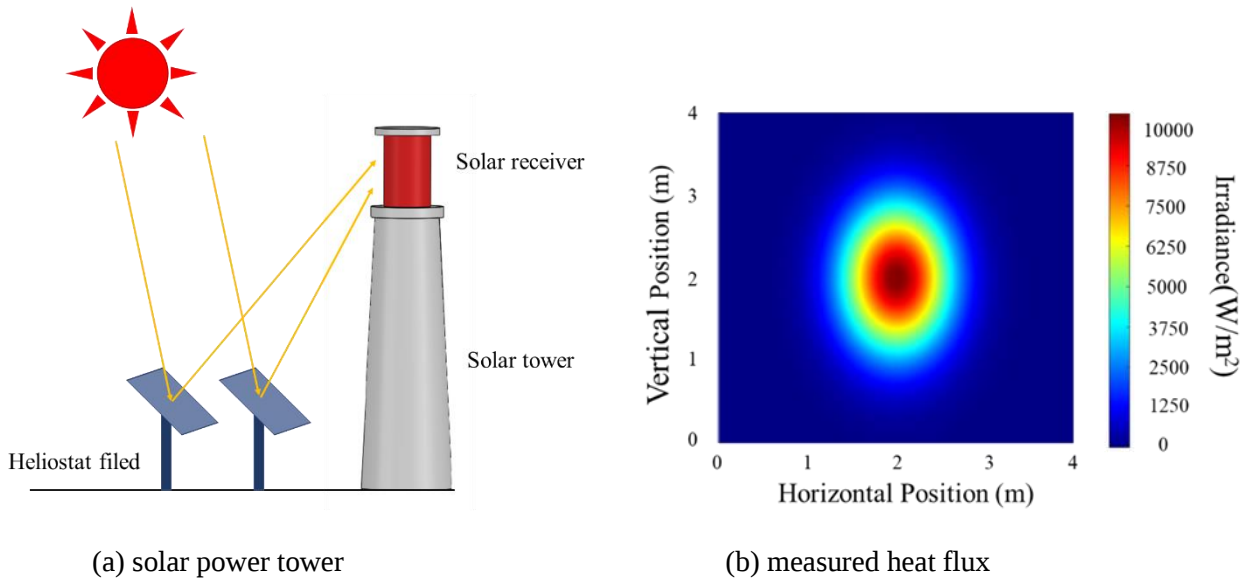


Figure 1. Schematic diagrams of (a) solar power tower and (b) heat flux on the receiver focal plane [24].

The design and optimization for future SCO_2 solar power tower requires a comprehensive and in-depth understanding of the heat transfer and flow characteristics of SCO_2 with various heat-flux distributions. The heat transfer behaviours of supercritical fluids are complex due to the drastically variable thermophysical properties near the pseudocritical temperature (T_{pc}). Although the near-pseudocritical region of CO_2 is not the main heating condition in the Brayton power cycle, the dynamic oscillation of the operating conditions during transitions such as from start-up to full-load [26] may lead to CO_2 near the pseudocritical region in the solar receiver [27,28]. Therefore, it is necessary to study thermal-hydraulic characteristics of SCO_2 near the pseudocritical region in vertical tubes with axially and circumferentially non-uniform heat-flux distribution.

The non-uniform heat-flux distribution would lead to a great difference to the heat transfer performance of SCO_2 in tubular solar receiver [29]. However, the reports on the thermal-hydraulic characteristics of SCO_2 under non-uniform heat-flux conditions are very few in existing literature, especially for vertical flows under circumferentially and axially non-uniform heat-flux conditions.

To bridge the gap between the existing researches and the practical applications in solar energy, the thermal-hydraulic characteristics of SCO_2 in a vertical tubular solar receiver under axially and circumferentially non-uniform heat-flux distributions for a solar power tower system are numerically studied in the present work, the effects of heat flux, mass flux and flow directions on the thermal-hydraulic characteristics of SCO_2 under four types of non-uniform heating conditions are investigated and analysed, and the buoyancy effect under the various heat-flux distributions is discussed in detail as well. The conclusions of this work provide insight of thermohydraulic characteristics of supercritical CO_2 flows as well as useful guidance for design and optimisation of tubular solar receiver and solar power system using supercritical CO_2 as heat transfer fluid.

2. Numerical methodology

2.1. Physical model and boundary conditions

The influence of the solid domain on the heat transfer is considered under the non-uniform heating condition in this study. The 316L stainless steel is selected as the material of the tube. The inner diameter of the tube is 6 mm and the wall thickness is set as 1 mm. We mainly focus on the flow and heat transfer characteristics of SCO_2 in the near-pseudocritical region under non-uniform heat-flux conditions, so the length of the heated section is 1,200 mm as shown in Figure 2, which is enough to conduct the studies within the ranges considered in the present work. The adiabatic sections of 300 mm and 200 mm are adopted in entrance and exit regions to eliminate the effect of unstable flow.

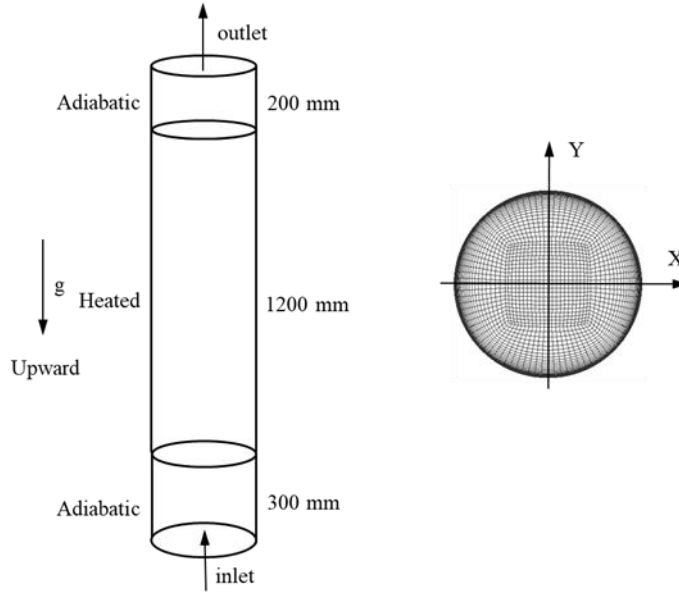


Figure 2. Physical model considered in the present work, with inner diameter of 6 mm, wall thickness of 1 mm, non-adiabatic length of 1,200 mm, and adiabatic length of 300 mm and 200 mm in the entrance and exit regions, respectively.

The heat-flux distribution on the solar thermal collector is complex in the solar power tower system, only one side of receiver wall is exposed to solar radiation. The HFLCAL (short for Heliostat Field Layout CALCulation) model [32] is one of the most widely used analytical models with a simple and closed-form expression to calculate the heat-flux distribution of the heat receiver in solar power tower system [33]. This model based on the Gaussian distribution is adopted in the present work. We only focus on the heat transfer characteristics of SCO_2 in a single receiver tube, the heat-flux distribution on receiver wall is supposed to follow the Gaussian distribution along the axial direction and cosine distribution on the semi-circumference side [34,35]. The heat-flux distribution is calculated as follows [36]:

$$q = \frac{q_{\max}}{\sqrt{2\pi}\sigma} \exp\left(-\frac{(z - z_m)^2}{2\sigma^2}\right) \quad (1)$$

$$q = q_{\text{ave}} \cdot \pi \cdot \cos \theta \quad (2)$$

where σ is the standard deviation of the Gaussian distribution, $q_{\max}$ and q_{ave} represent the maximum and average heat flux of the Gaussian distribution, z represents the coordinate on axial direction, z_m is the median of z .

According to the heat-flux distribution of solar power tower experimentally measured in Ref. [37], the value of standard deviation of the Gaussian distribution is set as 0.3 when the heating length of the solar receiver tube is 1.2 m in the present work, and the axially heat-flux distribution is shown in Figure 3.

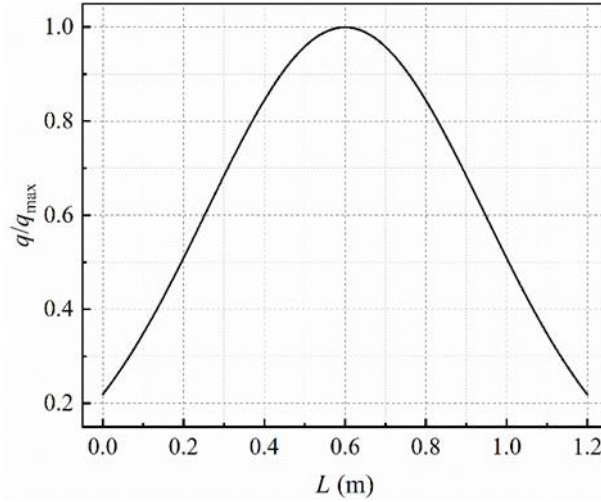


Figure 3. Distribution of the axial heat flux at the condition of axially non-uniform heat flux.

Four types of heat-flux conditions are considered to study the effect of heat-flux distribution on the thermal-hydraulic characteristics of SCO_2 , which are presented in Figure 4. They are circumferentially and axially uniform distribution (Case 1), semi-circumferentially and axially uniform distribution (Case 2), semi-circumferentially non-uniform and axially uniform distribution (Case 3), semi-circumferentially and axially non-uniform distribution (Case 4) respectively. In order to locate the position on the transverse sections, the lowest point of the tube in the negative y -direction is set to be 180° , the highest point is set to be 0° , and the leftmost and rightmost point are set to be -90° and 90° respectively. For semi-circumferential cases, the heat flux is only imposed on the wall of $-90^\circ \leq \theta \leq 90^\circ$ (top half of the tube wall shown in Figure 4(b)). To further describe the four heat fluxes quantitatively, the expressions for the four types of heat-flux distributions are listed in Table 1. It should be noted that the total heat transfer rate under four heat-flux conditions is the same when the average heat flux q_{ave} is fixed. The inlet temperature and outlet pressure are fixed as 304 K and 8 MPa ($T_{\text{pc}} \approx 307.8$ K at $P = 8$ Mpa) in all Cases. The velocity distribution and Reynolds number at the end of the inlet adiabatic section are carefully checked to ensure that the turbulence is fully developed before being heated in all cases. The Reynolds number ranges from 20,000 to 120,000 in the simulations, which conforms to the operating conditions of solar energy plants [31,38].

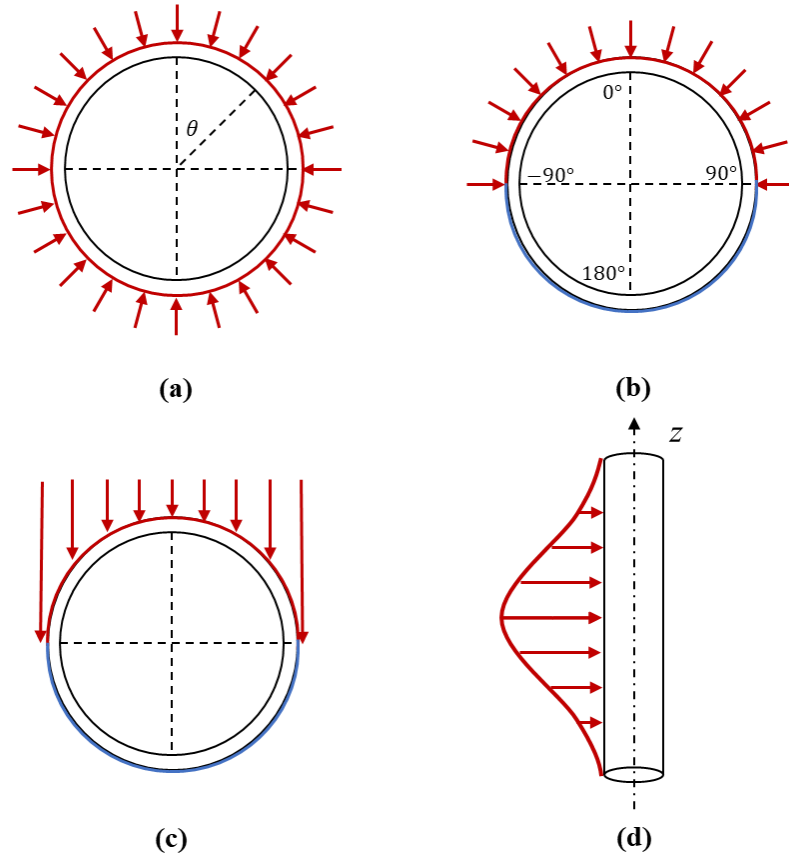


Figure 4. Four types of heat-flux conditions: (a) circumferentially uniform, (b) semi-circumferentially uniform, (c) semi-circumferentially non-uniform, (d) axially non-uniform in the vertical tube. The average equivalent heat flux remains constant in the four types of heat flux conditions.

Table 1. Expressions for the four types of heat-flux conditions with the same total heat transfer rate.

	Heating condition	Heat flux expression/ $\text{kW}\cdot\text{m}^{-2}$
Case 1	circumferentially uniform	$q = q_{\text{ave}}$
	axially uniform	
Case 2	semi-circumferentially uniform	$q = 2 \cdot q_{\text{ave}}$
	axially uniform	
Case 3	semi-circumferentially non-uniform	$q = q_{\text{ave}} \cdot \pi \cdot \cos \theta$
	axially uniform	
Case 4	semi-circumferentially non-uniform	$q = 0.76 \cdot q_{\text{ave}} \cdot \pi \cdot \exp(-4.233 \cdot (z-0.6)^2) \cdot \cos \theta$
	axially non-uniform	

2.2. Governing equations

The numerical calculation is carried out using ANSYS CFX 16.0 in the present work [39]. It's assumed that

the heat transfer remains steady state. The heat loss of the receiver in solar tower power plant could be effectively reduced to a very small percentage by adjusting cavity shapes, optimizing the surface radiation properties, etc., which has little effect on the heat-flux distribution on the wall and the thermal-hydraulic characteristics of fluid in the tube. Thus, the heat exchanged with ambient environment is neglected in the present work. The governing equations in solid domain and fluid domain are listed as follows [39].

Continuity equation:

$$\frac{\partial(\rho u_j)}{\partial x_j} = 0 \quad (3)$$

Momentum equation:

$$\frac{\partial(\rho u_i u_j)}{\partial x_j} = -\frac{\partial P_i}{\partial x_j} + \rho g_i + \frac{\partial}{\partial x_j} \left[(\mu + \mu_t) \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3} \frac{\partial u_l}{\partial x_l} \right) \right] \quad (4)$$

Energy equation for fluid domain:

$$\frac{\partial(\rho u_j c_p T)}{\partial x_j} = \frac{\partial}{\partial x_j} \left(\lambda \frac{\partial T}{\partial x_j} \right) + \phi \quad (5)$$

Energy equation for solid domain:

$$\frac{\partial}{\partial x_j} \left(\lambda_s \frac{\partial T}{\partial x_j} \right) = 0 \quad (6)$$

where ρ is density, μ is dynamic viscosity, μ_t is turbulent viscosity, λ and λ_s represents thermal conductivity of fluid and solid domains. ϕ is the energy dissipation, c_p represents the specific heat at constant pressure, g is gravitational acceleration.

The Shear-Stress Transport (SST) model [40] has advantages in robusticity near the wall and accuracy in the region apart from the wall which could result in relatively accurate simulation especially for supercritical fluids, which has been verified by lots of studies in the existing literature [29,41]. Thus, SST turbulence model is employed in the present work. The transport equations for SST turbulence model are defined as follows [42]:

Turbulence kinetic energy k :

$$\frac{\partial(\rho k)}{\partial t} + \frac{\partial(\rho k u_i)}{\partial x_i} = -\frac{\partial}{\partial x_j} \left(\Gamma_k \frac{\partial k}{\partial x_j} \right) + G_k - Y_k + S_k \quad (7)$$

Specific dissipation rate ω :

$$\frac{\partial(\rho \omega)}{\partial t} + \frac{\partial(\rho \omega u_i)}{\partial x_i} = -\frac{\partial}{\partial x_j} \left(\Gamma_\omega \frac{\partial \omega}{\partial x_j} \right) + G_\omega - Y_\omega + S_\omega \quad (8)$$

193 where G_k represents the production of k , G_ω represents the generation of ω , Γ_k and Γ_ω are the effective
 194 diffusivities of k and ω , Y_k and Y_ω represent the dissipation of k and ω respectively. S_k and S_ω are source terms.

195 A real gas property file containing the thermophysical properties of CO₂ obtained from NIST Standard
 196 Reference Database is imported to the ANSYS CFX software [43]. The governing equations are discretized by the
 197 finite volume method. High resolution is adopted for calculations in the governing equations. When the residuals
 198 for all monitored variables in Eqs. (3) to (8) are less than 1×10^{-6} , the numerical result is considered to be converged.

199 Numerous slices with equal distance of 50 mm along the flow direction are established to obtain local
 200 parameters such as wall temperature and wall heat flux. The overall parameters are obtained from local
 201 parameters. The local convective heat transfer coefficient is calculated as follows:

$$202 \quad h_b = \frac{q_w}{T_w - T_b} \quad (9)$$

203 where q_w is circumferentially average wall heat flux and T_w is circumferentially average wall temperature. T_b
 204 represents the fluid bulk temperature.

205 The heat transfer coefficient in the circumferential direction may exhibit severe non-uniformity because of
 206 the non-uniform heat-flux distribution. The local wall-circumference heat transfer coefficient is defined as follows:

$$207 \quad h_w(\theta) = \frac{q_w(\theta)}{T_w(\theta) - T_b} \quad (10)$$

208 where $q_w(\theta)$ and $T_w(\theta)$ are local wall heat flux and wall temperature based on the angular position shown in
 209 Figure 4, respectively.

210

211 2.3 Grid independence and model validation

212 To check the grid independence of numerical simulation, four types of grid systems are established by ANSYS
 213 ICEM CFD 16.0, and the number of elements is about 3.151×10^6 , 3.881×10^6 , 4.472×10^6 and 6.329×10^6
 214 respectively. The grid independence test results are shown in Figure 5 by using SST model under the condition
 215 of $G = 600 \text{ kg} \cdot \text{m}^{-2} \cdot \text{s}^{-1}$ and $q_{\text{ave}} = 180 \text{ kW} \cdot \text{m}^{-2}$. The maximum relative error of the local heat transfer coefficient
 216 between grid 3 and grid 4 is only about 2.6%. Considering the calculation time and accuracy, the grid 3 with
 217 4.472×10^6 elements is employed in the present work.

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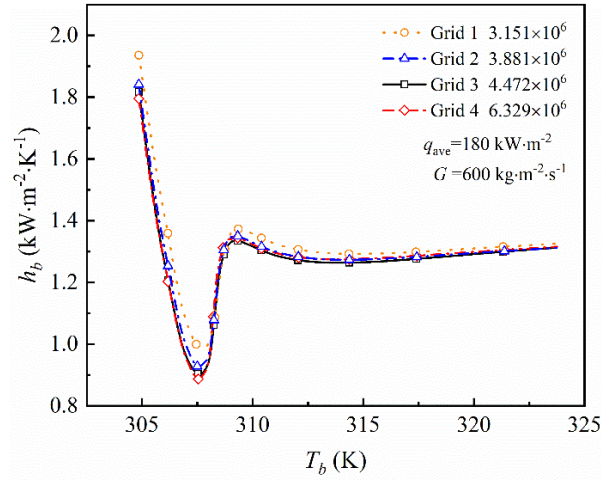


Figure 5. Local heat transfer coefficient distributions with four types of grid systems at the conditions of $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and $q_{\text{ave}} = 180 \text{ kW}\cdot\text{m}^{-2}$.

Due to the lack of experimental studies on vertical flow of SCO_2 under non-uniform heating condition, the experiment results with uniform heat-flux distribution conducted by Zhang [44] are adopted to validate the accuracy of the model in the present work. A same vertical tube model with diameter of 10 mm is established, and the mass flux of $488 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$, inlet temperature of 302.5 K and working pressure of 8.24 MPa in the experiment are adopted to validate the present model. The comparison between the local wall temperature obtained by the simulation and experiment is presented in Figure 6. The maximum relative error of wall temperature between simulation and experiment is about 6.42 %, which could be accepted in the present work. Thus, the numerical model employed in the present work is reliable and acceptable.

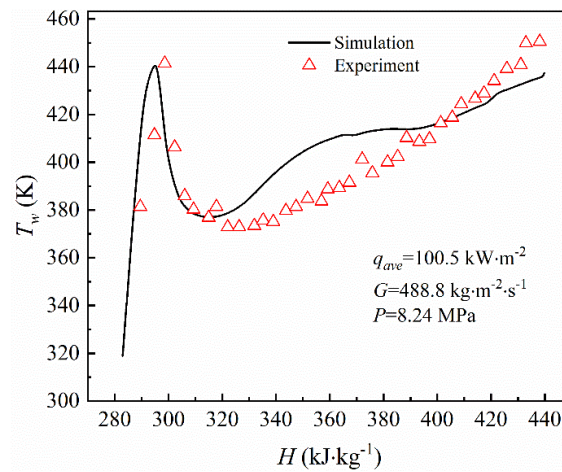


Figure 6. The comparison of wall temperature between the numerical results and experimental results in Ref. [44] at conditions of $q_{\text{ave}} = 100.5 \text{ kW}\cdot\text{m}^{-2}$, $G = 488 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and $P = 8.24 \text{ MPa}$.

236 3. Results and discussion

237 This section is arranged as follows: the flow and heat transfer characteristics of SCO₂ with vertical upward flow
 238 under four types of heat-flux conditions listed in Table 1 are studied in section 3.1; in section 3.2, the effects of
 239 buoyancy on flow and heat transfer behaviours of SCO₂ are discussed in detail; and finally, the effects of flow
 240 directions on heat transfer performance under non-uniform heat-flux conditions are investigated in section 3.3.

241

242 3.1. Influence of heat flux and mass flux

243 In this subsection, the heat transfer performance of SCO₂ in vertical upward flow will be studied under the
 244 four types of heat-flux distributions listed in Table 1 based on the model in Figure 1.

245 The relationship of the overall heat transfer coefficient h with Reynolds number Re under four types of
 246 heating conditions listed in Table 1 is presented in Figure 7. The overall heat transfer coefficient increases
 247 monotonically with the increase of Re when other conditions are fixed, and the higher the heat flux is, the
 248 lower the overall heat transfer coefficient would be. The h of axially non-uniformly heat-flux distribution (Case
 249 4) is 1.2% to 20.5% higher than that of axially uniformly heat-flux distribution (Case 3) under conditions of
 250 $q_{ave} = 150$ to $210 \text{ kW}\cdot\text{m}^{-2}$ and $Re = 4.4 \times 10^4$ to 7.8×10^4 , and this discrepancy becomes more significant as
 251 the heat flux increases. With the increase of Re , the superiority of axially non-uniformly heating condition
 252 becomes more obvious at $q_{ave} = 210 \text{ kW}\cdot\text{m}^{-2}$. However, this difference of the h between axially non-uniform
 253 and axially uniform heating firstly increases and then gradually weakens with the increase of Re at $q_{ave} = 150$
 254 $\text{ kW}\cdot\text{m}^{-2}$ and $q_{ave} = 180 \text{ kW}\cdot\text{m}^{-2}$. It can be seen that the h almost coincides together under four types of heating
 255 conditions at $q_{ave} = 150 \text{ kW}\cdot\text{m}^{-2}$ and $Re = 7.8 \times 10^4$. When the axially heat flux is uniform, the h of semi-
 256 circumferentially non-uniform heating condition (Case 3) is 0.3% to 1.3% higher than that of semi-
 257 circumferentially uniform (Case 2) and fully uniform (Case 1) heat-flux conditions. The same phenomenon is
 258 also found in the circumferentially non-uniformly vertical heated tube in the numerical study conducted by
 259 Fan et al. [27]. Therefore, the overall heat transfer of SCO₂ is enhanced under non-uniform heating conditions.

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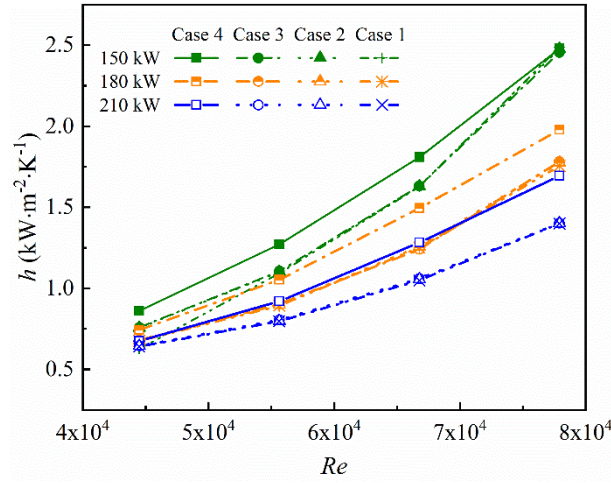
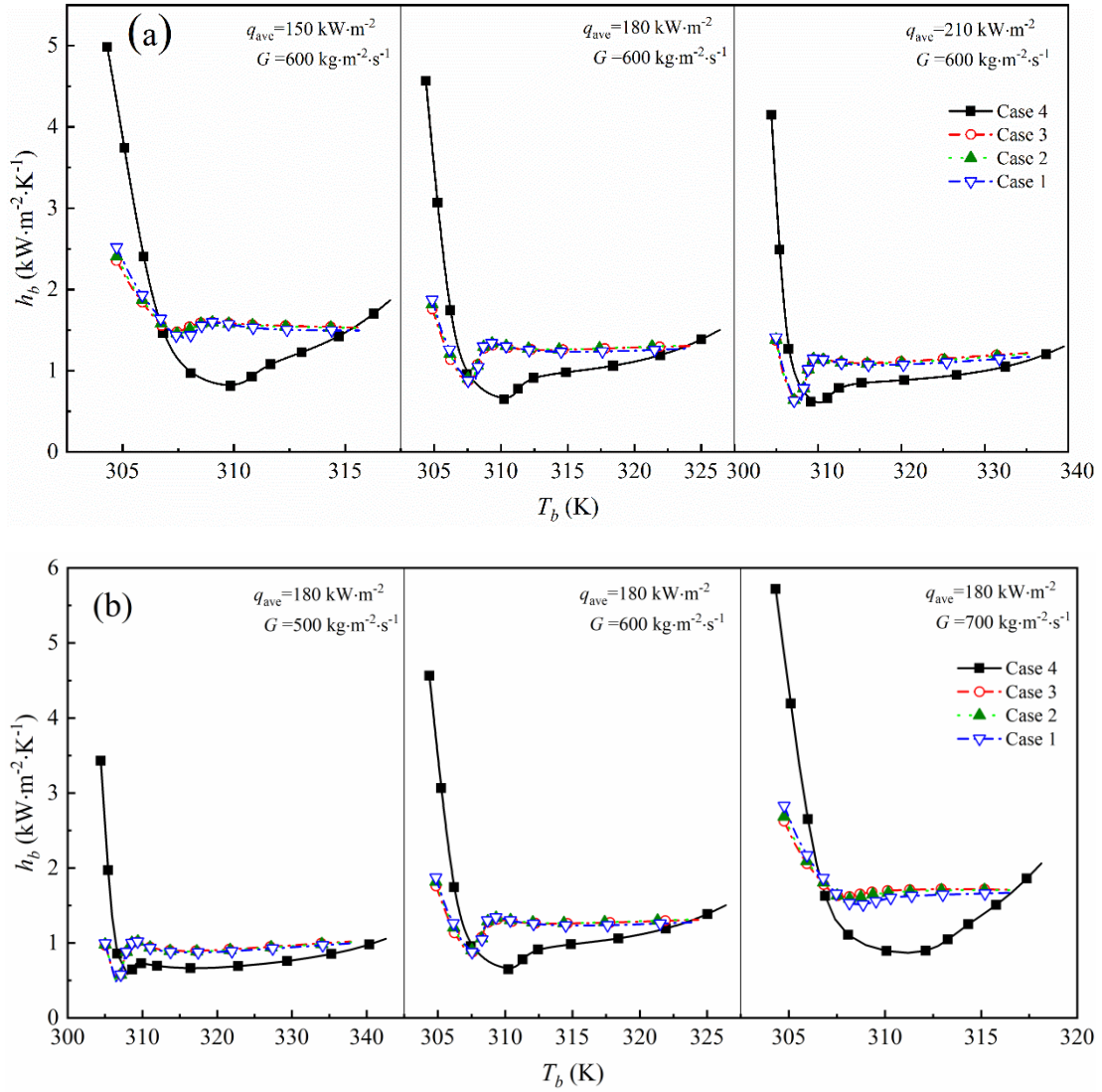


Figure 7. Relationship of overall heat transfer coefficient with inlet Re under four types of heat-flux conditions listed in Table 1 at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $q_{ave} = 150$ to 210 kW·m⁻².

The relationship of the local bulk heat transfer coefficient h_b with fluid bulk temperature T_b under four types of heat-flux conditions listed in Table 1 with average heat flux ranging from 150 kW·m⁻² to 210 kW·m⁻² and mass flux ranging from 500 kg·m⁻²·s⁻¹ to 700 kg·m⁻²·s⁻¹ is shown in Figure 8. Different heat flux distributions lead to significant discrepancies in the local heat transfer behaviour under the same mass-flux and heat-flux conditions. Figure 8(a) depicts the distribution of h_b with various heat fluxes at $G = 600$ kg·m⁻²·s⁻¹. It can be seen that there are obvious troughs in h_b under four types of heat-flux conditions at $q_{ave} = 180$ kW·m⁻² and $q_{ave} = 210$ kW·m⁻². However, at the condition of $q_{ave} = 150$ kW·m⁻², the trough of the h_b only can be found in Case 4 of axially non-uniform heat-flux distribution and there is no significant trough for axially uniform heating. The distribution of h_b with various mass fluxes under the heat flux of 180 kW·m⁻² is presented in Figure 8 (b). The h_b have significant troughs along the flow direction in most conditions except in the case of axially uniform heating at $G = 700$ kg·m⁻²·s⁻¹. At the same mass flux ($G = 500$ to 700 kg·m⁻²·s⁻¹) and average heat flux ($q_{ave} = 150$ to 210 kW·m⁻²), the T_b corresponding to the trough of the h_b under the axially non-uniform heating (Case 4) is 1.6 K to 3.3 K higher than that under the axially uniform heat-flux condition (Case 3), and this disparity is more significant at the low heat flux ($q_{ave} = 150$ kW·m⁻²) and high mass flux ($G = 700$ kg·m⁻²·s⁻¹). It is obvious that the minimum h_b is smaller under the axially non-uniform heat-flux distribution of Case 4 than under other cases. However, the h_b of Case 4 is much higher than that of other heating conditions in the initial heating section of the tube, which is the main reason why the overall h of Case 4 is higher than that of axially uniform heat-flux condition. Based on the above analysis, the overall and local heat transfer coefficients exhibit complex variations under different heating conditions, the local heat transfer characteristics of SCO₂

284 in a vertical tube are closely related to the mass flux and the local heat flux distribution.

285



286

287

288 **Figure 8.** Local bulk heat transfer coefficient distributions under (a) various heat fluxes with mass flux of
 289 $600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and (b) various mass fluxes with heat flux of $180 \text{ kW}\cdot\text{m}^{-2}$ at conditions of $T_{\text{in}} = 304 \text{ K}$,

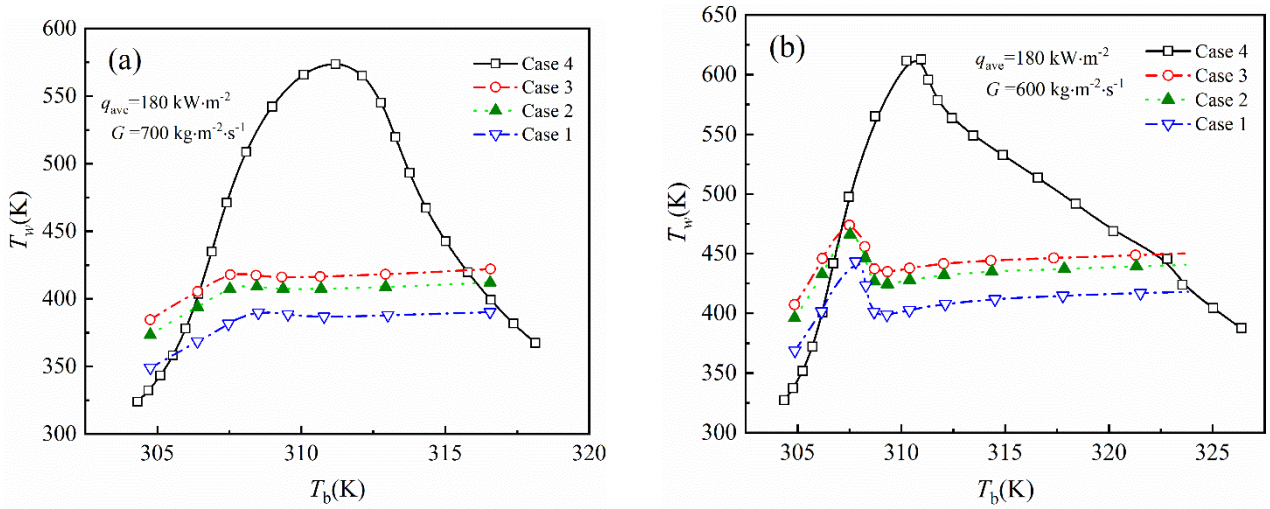
290 $P = 8 \text{ MPa}$, $q_{\text{ave}} = 150 \text{ to } 210 \text{ kW}\cdot\text{m}^{-2}$ and $G = 500 \text{ to } 700 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$.

291

292 Wall temperature is an important parameter to predict the deterioration of local heat transfer. The
 293 distributions of axially outer wall ($\theta = 0^\circ$ as shown in Figure 4) temperature T_w along the flow direction at
 294 conditions of $q_{\text{ave}} = 180 \text{ kW}\cdot\text{m}^{-2}$ and mass flux ranging from $500 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ to $700 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ are presented in
 295 Figure. 9. When the axial heat flux is uniformly distributed, there exists a peak of T_w , where T_b are about
 296 306.4 K and 307.3 K along the flow direction at $G = 500 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$. However, the T_w
 297 increases gradually without obvious inflection point along the flow direction at $G = 700 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$. According

298 to the method proposed by Urbano et al. [45], if $(dT_w/dx)_{\min} < 0$ and T_w has a significant peak under uniform
 299 heating condition, it can be considered that heat transfer deterioration has occurred. Therefore, heat transfer
 300 deterioration occurs at $G = 500 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ with axially uniform heat-flux distribution,
 301 which also corresponds to the distribution of the h_b in Figure 8(b). In addition, based on the comparison
 302 between T_w and h_b distributions, it can be concluded that this criterion is still applicable to circumferentially
 303 non-uniform heating condition. For axially non-uniform heat-flux condition (Case 4), at the condition of
 304 $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$, the peak of T_w only appears in the place at $T_b \approx 311.0 \text{ K}$, which corresponds to the minimum
 305 of h_b in Figure 8(b), and the same phenomenon is also observed at $G = 700 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$. However, at
 306 $G = 500 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$, two wall temperature inflection points along the flow direction can be found, one of which
 307 appears in the place at $T_b \approx 308.7 \text{ K}$ corresponding to the minimum value of h_b in Figure 8(b). Another
 308 inflection point is in the middle of the tube, where the heat flux is the highest. Thus, the heat transfer
 309 deterioration of SCO_2 cannot be predicted accurately only from the wall temperature inflection point under the
 310 axially non-uniform heat-flux condition.

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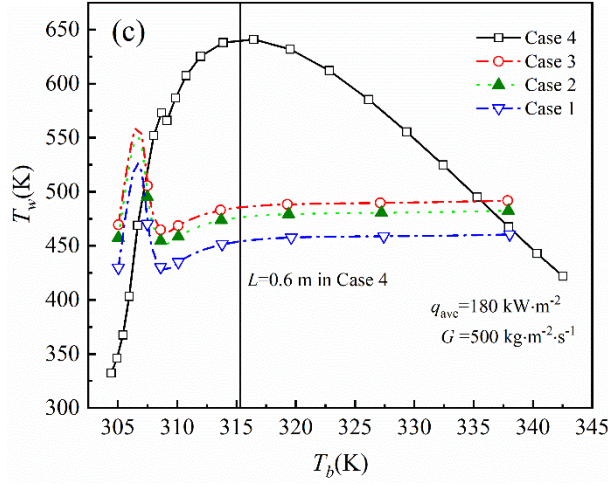


Figure 9. Axial wall temperature distributions at (a) $q_{ave} = 180 \text{ kW}\cdot\text{m}^{-2}$ and $G = 700 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$, (b)

$q_{ave} = 180 \text{ kW}\cdot\text{m}^{-2}$ and $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$, (c) $q_{ave} = 180 \text{ kW}\cdot\text{m}^{-2}$ and $G = 500 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ under the four types

of heat-flux conditions listed in Table 1 at other conditions of $T_{in} = 304 \text{ K}$ and $P = 8 \text{ MPa}$.

The circumferential wall-temperature distribution at the cross-section of where $T_b/T_{pc} = 1$ (corresponding to

$z \approx 367 \text{ mm}$ at Case 4, $z \approx 275 \text{ mm}$ at Case 3, Case 2 and Case 1) under the condition of $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ and

$q_{ave} = 180 \text{ kW}\cdot\text{m}^{-2}$ is shown in Figure 10. The distribution of T_w is approximately similar to the distribution of heat

flux. The region where the heat source is concentrated has the highest T_w in the case of unevenly circumferential

heat-flux distribution. When the heat flux is uniformly distributed along the circumference, T_w is also uniformly

distributed, and the local heat transfer deterioration of SCO_2 does not affect the distribution of T_w along the

circumference. It can be found that T_w and circumferential wall temperature gradient at the cross-section of where

$T_b/T_{pc} = 1$ are higher under axially non-uniform heat-flux condition than under axially uniform heating condition.

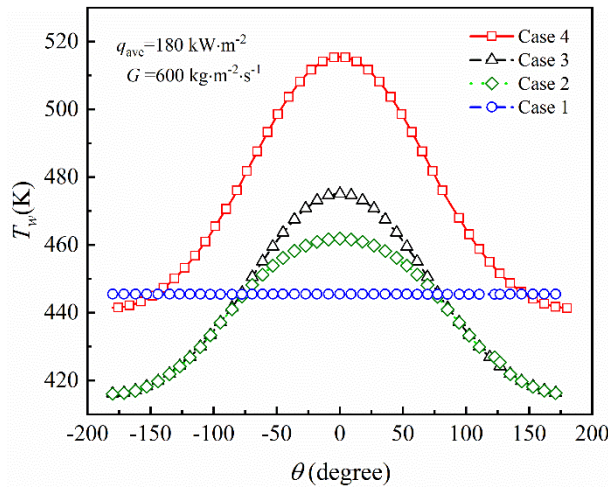


Figure 10. Circumferential wall-temperature distribution in the cross-section of where $T_b/T_{pc} = 1$

corresponding to $z \approx 367$ mm at Case 4, $z \approx 275$ mm at Case 3, Case 2 and Case 1 at conditions of

$$T_{\text{in}} = 304 \text{ K}, P = 8 \text{ MPa}, G = 600 \text{ kg} \cdot \text{m}^{-2} \cdot \text{s}^{-1} \text{ and } q_{\text{ave}} = 180 \text{ kW} \cdot \text{m}^{-2}.$$

In order to further understand the differences in local heat transfer behaviours under different heat-flux conditions listed in Table 1, the wall circumferential heat transfer coefficient h_w at positions of $\theta = 0^\circ$ and $\theta = 180^\circ$ along the flow direction under conditions of $G = 600 \text{ kg} \cdot \text{m}^{-2} \cdot \text{s}^{-1}$ and $q_{\text{ave}} = 180 \text{ kW} \cdot \text{m}^{-2}$ is demonstrated in Figure 11. The h_w follows the same trend as h_b , but h_w varies more intensely along the flow direction. When the heat flux is uniformly distributed (Case 1), the h_w is equal at locations of $\theta = 0^\circ$ and $\theta = 180^\circ$. It is seen that the heat transfer is enhanced at the unheated wall ($\theta = 180^\circ$) under circumferentially non-uniform heat-flux conditions (Case 2 and Case 3). For axially non-uniform heating, the h_w at locations of $\theta = 0^\circ$ and $\theta = 180^\circ$ is almost same on the transverse section of $z \approx 0.42$ m, although the h_w of unheated wall ($\theta = 180^\circ$) is higher than that of heated wall ($\theta = 0^\circ$) on other locations along the flow direction. Since the circumferential wall temperature changes more intensely at Case 4, and the fluid temperature in the near-wall region where the heat flux is not imposed ($\theta = 180^\circ$) is closer to the T_{pc} than that in the heated wall ($\theta = 0^\circ$). Thus, the change of h_w at $\theta = 180^\circ$ is greater than that at $\theta = 0^\circ$ on the transverse section of $z \approx 0.42$ m.

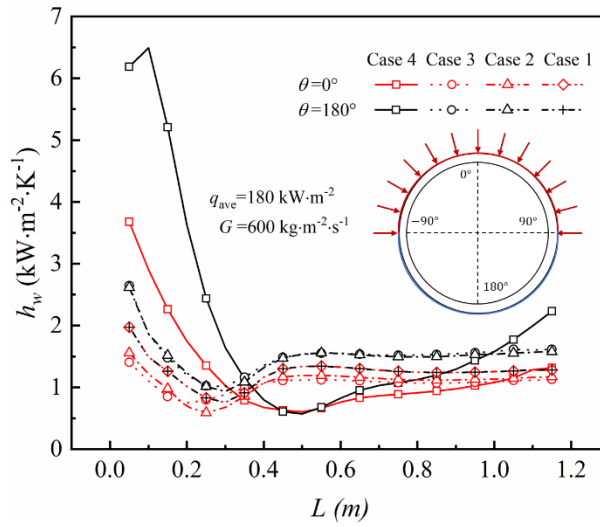


Figure 11. Distribution of local heat transfer coefficient at locations of $\theta = 0^\circ$ and $\theta = 180^\circ$ along the flow direction at conditions of $T_{\text{in}} = 304 \text{ K}$, $P = 8 \text{ MPa}$, $G = 600 \text{ kg} \cdot \text{m}^{-2} \cdot \text{s}^{-1}$ and $q_{\text{ave}} = 180 \text{ kW} \cdot \text{m}^{-2}$.

3.2. Influence of buoyancy effect

The complex heat transfer properties of SCO_2 in vertical tubes are related to buoyancy effect [46]. The buoyancy effect on the flow and heat transfer behaviours of SCO_2 in the vertical tube with different heating

352 methods will be discussed in detail in this subsection.

353 To quantify the influence of buoyancy, Jackson [47] proposed a dimensionless buoyancy force factor Bo^*
354 to determine the strength of the buoyancy force in a vertical tube. It is generally accepted that when
355 $Bo^* > 5.6 \times 10^{-7}$, the effect of the buoyancy force cannot be ignored.

356
$$Bo^* = Gr^* / (Re^{3.425} Pr^{0.8}) \quad (11)$$

357
$$Gr^* = \frac{g\beta q d^4}{\lambda_b \nu_b^2} \quad (12)$$

358
$$\beta = -\frac{1}{\rho} \left(\frac{\partial \rho}{\partial T} \right)_p \quad (13)$$

359 where β is the fluid volume expansion coefficient, Pr is Prandtl number, ν is the kinetic viscosity.

360 Bo^* is widely used and has been proved to have a good performance in predicting the effect of buoyancy
361 on the heat transfer performance with vertical tubes under uniform heating condition [46, 48]. The distribution
362 of local buoyancy factor Bo^* at conditions of $G = 600 \text{ kg} \cdot \text{m}^{-2} \cdot \text{s}^{-1}$ and $q_{ave} = 180 \text{ kW} \cdot \text{m}^{-2}$ for the four types of
363 heat-flux distributions is presented in Figure 12. The local Bo^* decreases gradually along the flow direction in
364 all cases, and the region with more significant influence on buoyancy ($Bo^* > 5.6 \times 10^{-7}$) is concentrated in the
365 first half of the tube ($z < 0.5 \text{ m}$). With the increase of T_b , the change of thermophysical properties of SCO_2
366 slows down and the buoyancy effect is gradually weakened. The variation trend of the Bo^* under the non-
367 uniform heat-flux condition (Case 4, Case 3 and Case 2) is similar to that under uniform heat-flux condition
368 (Case 1), which is in accordance with the distribution of h_b in Figure 8. Thus, Bo^* could provide a good
369 prediction of the buoyancy effect under circumferentially and axially non-uniform heat-flux conditions. In the
370 region where buoyancy force affects the heat transfer ($z < 0.5 \text{ m}$), the Bo^* at axially non-uniform heating
371 condition (Case 4) is significantly greater than that at axially uniform heat-flux conditions, indicating buoyancy
372 is more visible under axially non-uniform heating condition than under the axially uniform heating condition.

373

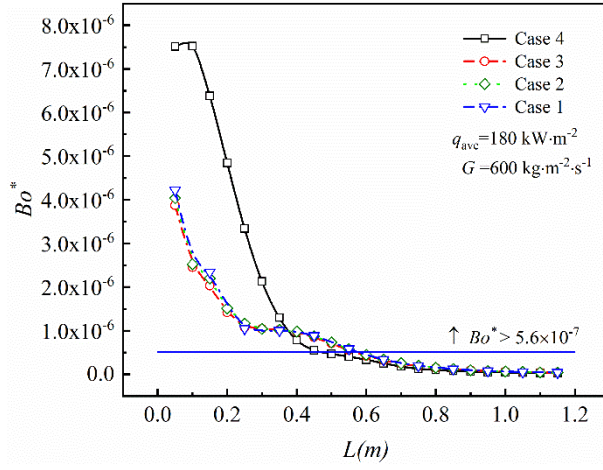
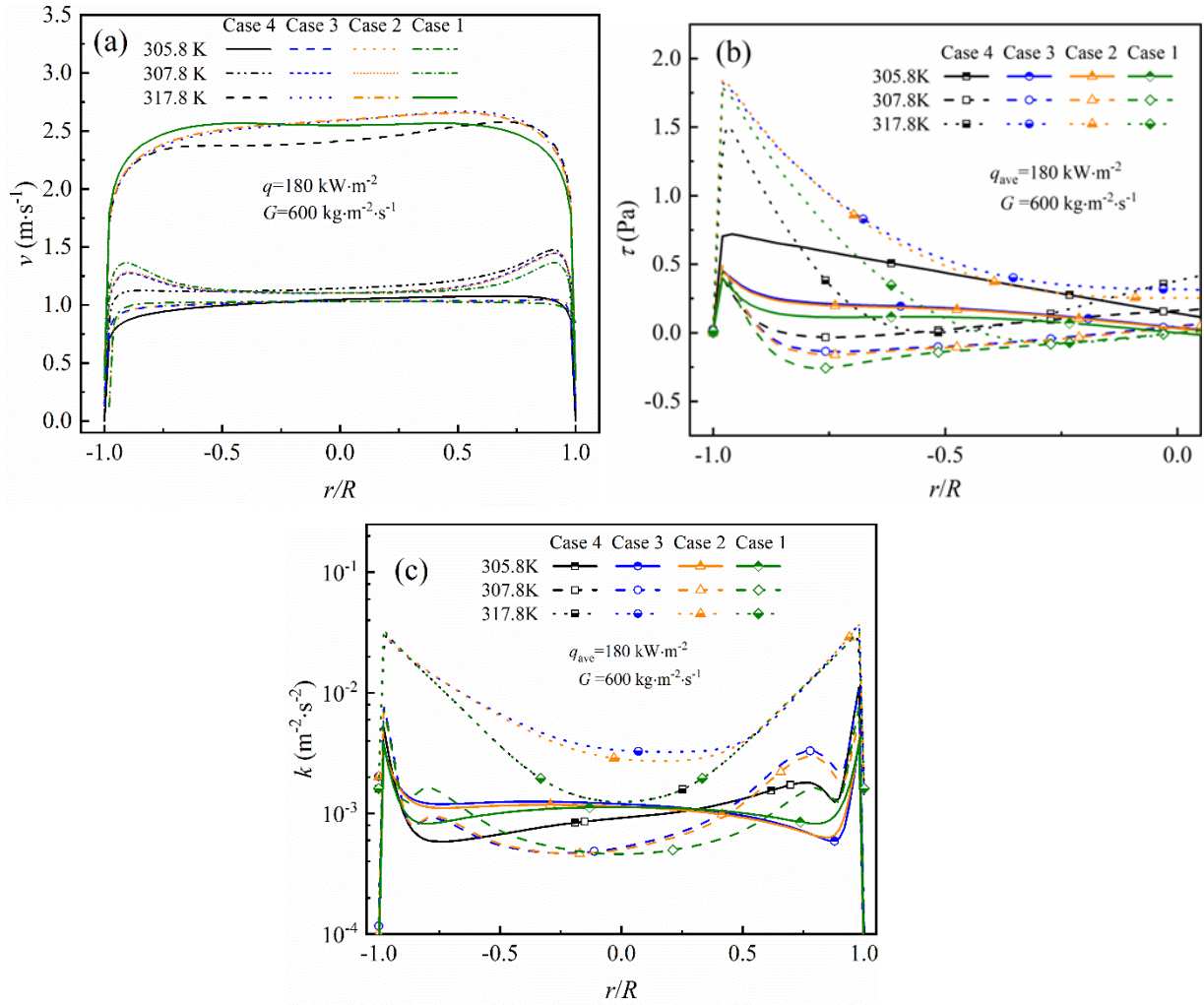


Figure 12. Distribution of local buoyancy factor Bo^* under four types of heating conditions listed in Table 1 at other conditions of $T_{in} = 304$ K, $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻².

Three transverse sections are established (at $T_b \approx 305.8$ K, $T_b \approx 307.8$ K and $T_b \approx 317.8$ K) to compare local parameters under different heat transfer performances. Heat transfer is deteriorated on the transverse section of where T_b is about 307.8 K and is restored on the cross-section of where T_b is about 317.8 K. Figure 13 shows the radial distributions of velocity v , turbulent shear stress τ and turbulent kinetic energy k on the cross-sections of where $T_b \approx 305.8$ K (corresponding to $z \approx 208$ mm at Case 4, $z \approx 120$ mm at Case 3, Case 2 and Case 1), $T_b \approx 307.8$ K (corresponding to $z \approx 367$ mm at Case 4, $z \approx 275$ mm at Case 3, Case 2 and Case 1), $T_b \approx 317.8$ K (corresponding to $z \approx 835$ mm at Case 4, $z \approx 955$ mm at Case 3, Case 2 and Case 1) under the four types of heat-flux conditions with other conditions of $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻². At the cross-section of where $T_b \approx 305.8$ K, the variation of velocity is mainly concentrated in the near-wall region, where a great velocity gradient can be observed as shown in Figure 13(a). The k is slightly suppressed at the cross-section of where $T_b \approx 305.8$ K, and for circumferentially non-uniform heat-flux conditions (Case 3 and Case 2), it can be seen the change of k is more significant in the near-wall region with heat-flux distribution ($r > 0$) than that without heat-flux distribution ($r < 0$). When bulk temperature reaches T_{pc} (≈ 307.8 K), a peak of velocity is formed in the near-wall region for the strong buoyancy in this region, and the velocity profile exhibits an M-shaped distribution. Non-uniform distribution of circumferential heat flux results in an asymmetric M-shaped velocity profile, which is more pronounced for axially non-uniform heating. Negative turbulent shear stress is formed because of the negative velocity gradient, and the turbulent kinetic energy is strongly suppressed at the cross-section of where $T_b \approx 307.8$ K as shown in Figure 13(c). With the increase of the T_b , the fluid velocity increases significantly due to the decrease of density. The peak of velocity in the near-

397 wall region disappears, and the velocity profile exhibits flat again. The turbulent kinetic energy and heat
 398 transfer capacity are restored at the cross-section of where $T_b \approx 317.8$ K.
 399



400

401

402

403 **Figure 13.** Radial distributions of (a) velocity, (b) turbulent shear stress and (c) turbulent kinetic energy in
 404 several cross-sections of where $T_b = 305.8$ K, $T_b = 307.8$ K and $T_b = 317.8$ K at conditions of $T_{in} = 304$ K,
 405 $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻².
 406

407 The turbulence production rate G_k representing the generation of turbulence kinetic energy due to mean
 408 velocity gradient can be calculated by [42]:

$$409 \quad G_k = \mu_t \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \frac{\partial u_i}{\partial x_j} \quad (13)$$

410 The distribution of turbulent production rate G_k at the cross-section of where $T_b / T_{pc} = 1$ (corresponding
 411 to $z \approx 367$ mm at Case 4, $z \approx 275$ mm at Case 3, Case 2 and Case 1) at the conditions of $G = 600$ kg·m⁻²·s⁻¹ and

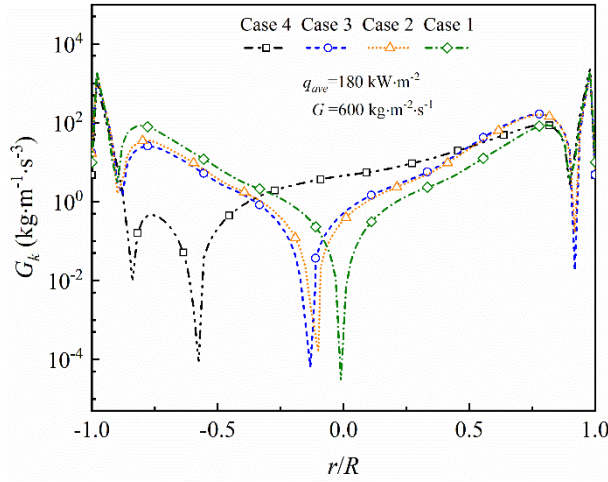
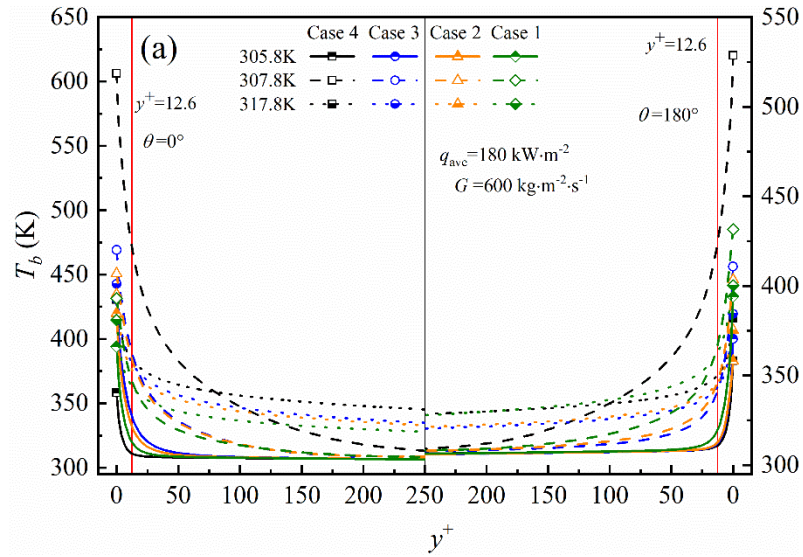


Figure 14. Radial distribution of turbulent production rate on the cross-section of where $T_b/T_{pc} = 1$ corresponding to $z \approx 367$ mm at Case 4, and $z \approx 275$ mm at Case 3, Case 2 and Case 1 respectively, at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻².

The convective heat transfer of fluid is significantly affected by flow and heat transfer in the momentum and thermal boundary layer. The Prandtl number Pr could be regarded as the ratio of momentum diffusion capacity to thermal diffusion capacity, which represents the relative thickness of the momentum boundary layer and the thermal boundary layer. The turbulent boundary layer is defined by the dimensionless distance y^+ which is the ratio of the distance from the wall to viscous length scale, and the definition proposed by Bazargan et al. [49] and Hiroaki et al. [50] is employed in the present work for SCO₂, the viscous sublayer is in the region where $y^+ < 12.6$, the buffer layer is in the region where $12.6 < y^+ < 60$ and logarithmic-law layer is in the region where $60 < y^+ < 300$.

The radial distributions of fluid temperature and Pr on the cross-sections of where $T_b \approx 305.8$ K,

435 $T_b \approx 307.8$ K and $T_b \approx 317.8$ K under four types of heat-flux conditions with other conditions of
 436 $q_{ave} = 180 \text{ kW}\cdot\text{m}^{-2}$ and $G = 600 \text{ kg}\cdot\text{m}^{-2}\cdot\text{s}^{-1}$ are displayed in Figure 15. On the cross-section of where
 437 $T_b \approx 305.8$ K, the variation of fluid temperature is mainly concentrated in the viscous sublayer region which
 438 represents a thin thermal boundary layer. Pr varies dramatically in the viscous sublayer region especially under
 439 the axially non-uniform heating condition as shown in Figure 15(b). The thermal resistance is mainly
 440 concentrated in the viscous bottom layer at the cross-section of where $T_b \approx 305.8$ K. As T_b gradually approaches
 441 the T_{pc} , the change of fluid temperature is not limited to the viscous sublayer region, and the distribution of
 442 fluid temperature resembles a parabola under the axially non-uniform heat-flux condition. The thermal
 443 boundary layer becomes much thicker at this section. The value of Pr is larger than one, indicating the
 444 momentum boundary layer is thicker than the thermal boundary layer, and the heat transfer of SCO_2 is
 445 deteriorated at the cross-section of where $T_b \approx 307.8$ K. When the fluid bulk temperature reaches 317.8 K, the
 446 temperature profile flattens out again and the thermal boundary layer becomes thinner. The Pr is near one,
 447 which indicates that the momentum layer is almost as thin as the thermal layer, and the heat transfer is
 448 recovered at this cross-section. Thus, it can be obtained that the large heat transfer resistance caused by thick
 449 momentum and thermal boundary layer should be an important reason for the heat transfer deterioration.



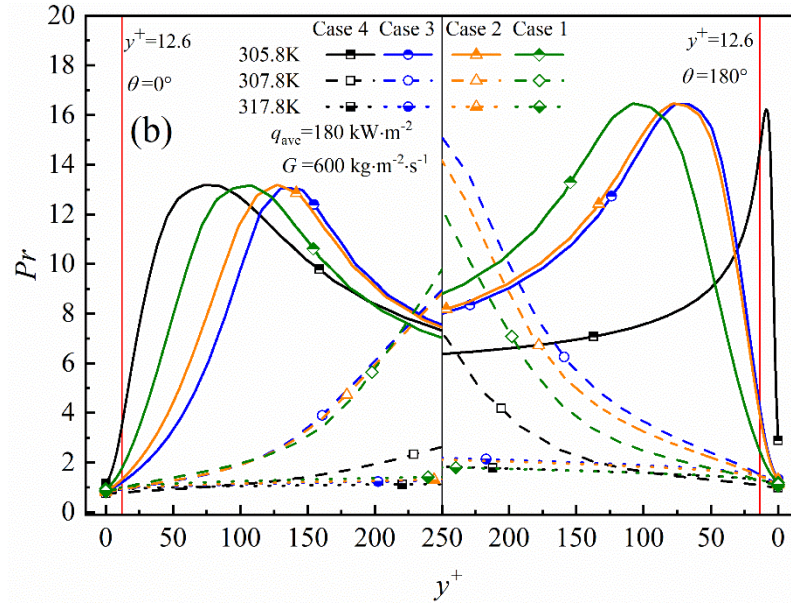


Figure 15. Radial distributions of (a) fluid temperature and (b) Prandtl number on several cross-sections of where $T_b = 305.8$ K, 307.8 K and 317.8 K at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻².

3.3. Influence of flow direction

The effects of flow direction on heat transfer performance under non-uniform heating conditions are investigated in this subsection. Besides the vertical downward and upward flows along the z -axis, the thermal-hydraulic characteristics of flow without gravity is also considered as a baseline case.

The overall heat transfer coefficient h under the two non-uniform heating conditions of Case 4 and Case 3 for the three flows is presented in Figure 16. The vertical downward flow has the best heat transfer performance and the heat transfer performance of vertical upward flow is the worst among the three flows. The h of downward flow is 62.0% to 75.4% and 55.4% to 76.1% higher than that in upward flow under the axially non-uniform heat-flux condition (Case 4) and circumferentially non-uniform heat-flux condition (Case 3), respectively. The difference of h between downward and upward flows shrinks with the reduction of Reynolds number Re and the increase of heat flux. The overall heat transfer performance of Case 4 is better than that of Case 3 even in downward flow and the flow without gravity, which is the same as that of upward flow, and this discrepancy is becoming more evident with the increase of heat flux.

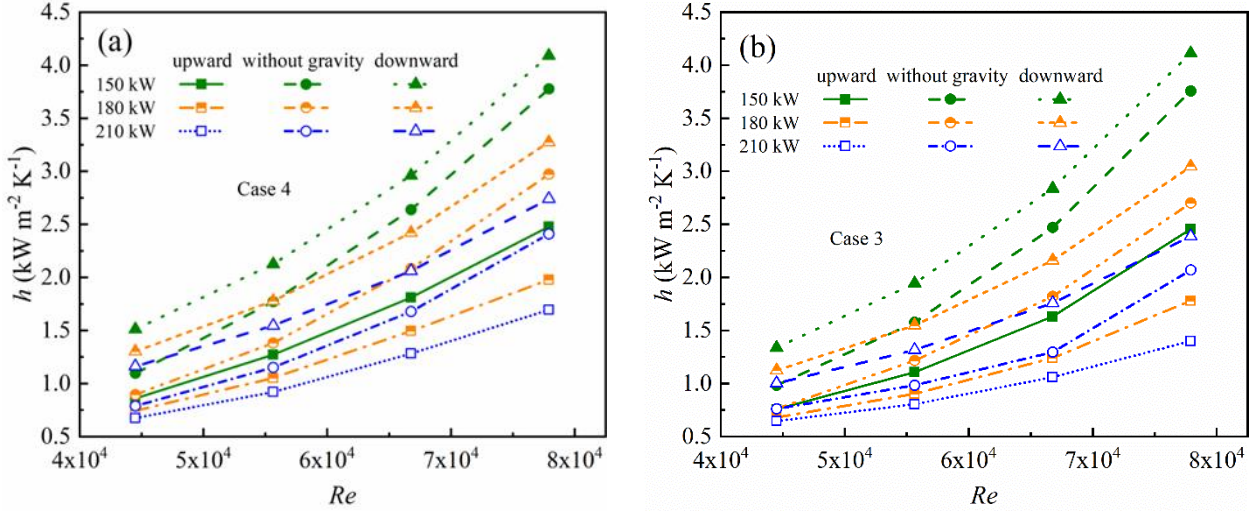


Figure 16. Overall heat transfer coefficient in Case 4 and Case 3 in upward and downward flows as well as the flow without considering gravity at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $q_{ave} = 150$ - 210 kW·m⁻² and $Re = 4.4 \times 10^4$ to 7.8×10^4 .

Figure 17 demonstrates the temperature contours of the outer wall surface in Case 4 and Case 3 at conditions of $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻² for the three flows, where “a” represents the vertical upward flow, “b” represents the flow without gravity and “c” represents the vertical downward flow. The wall with heat flux distribution is called as heated wall. A relatively large wall-temperature non-uniformity in the axial direction is observed in Case 4 and Case 3 especially for upward flow. Under the axially non-uniform heat-flux condition (Case 4), the maximum T_w on both heated and adiabatic wall in vertical upward flow is significantly higher than that in downward flow and the flow without gravity, and the location of maximum T_w is more forward (closer to the inlet) with upward flow. In Case 3, the maximum T_w in the vertical upward flow is obtained near the inlet of the tube, while the highest wall temperature in the vertical downward flow and the flow without gravity appears at the outlet of the tube as shown in Figure 17(b).

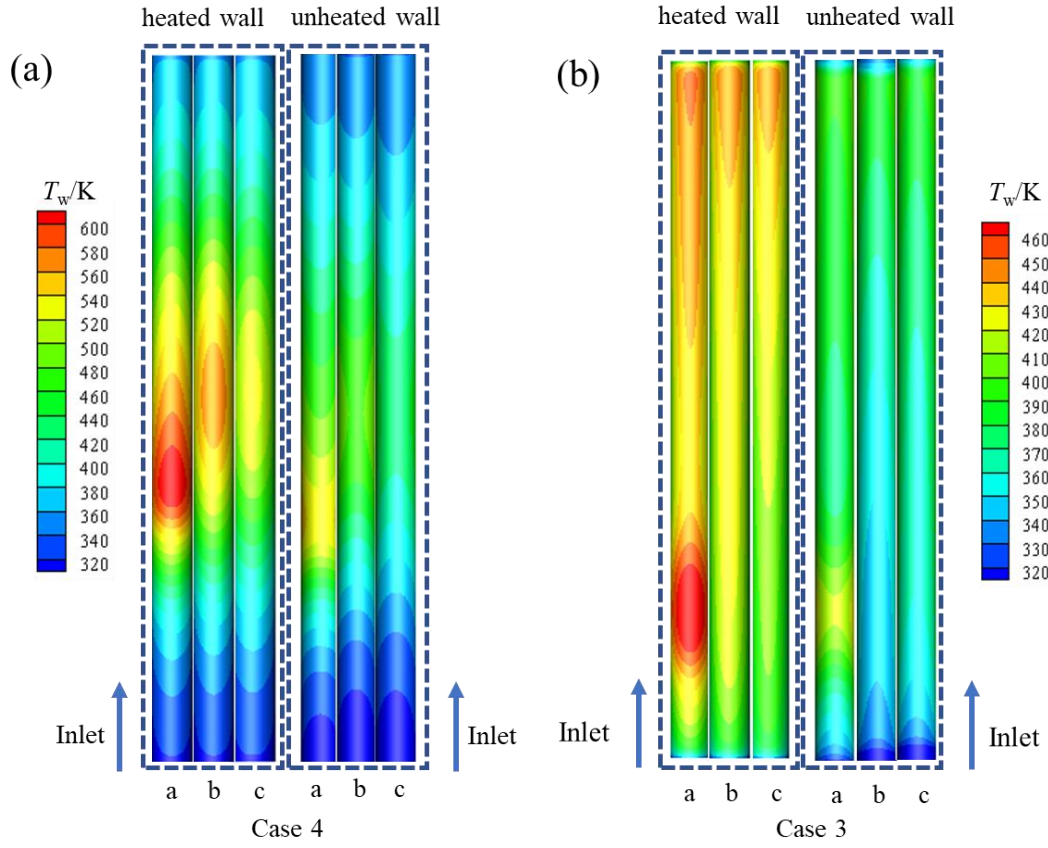
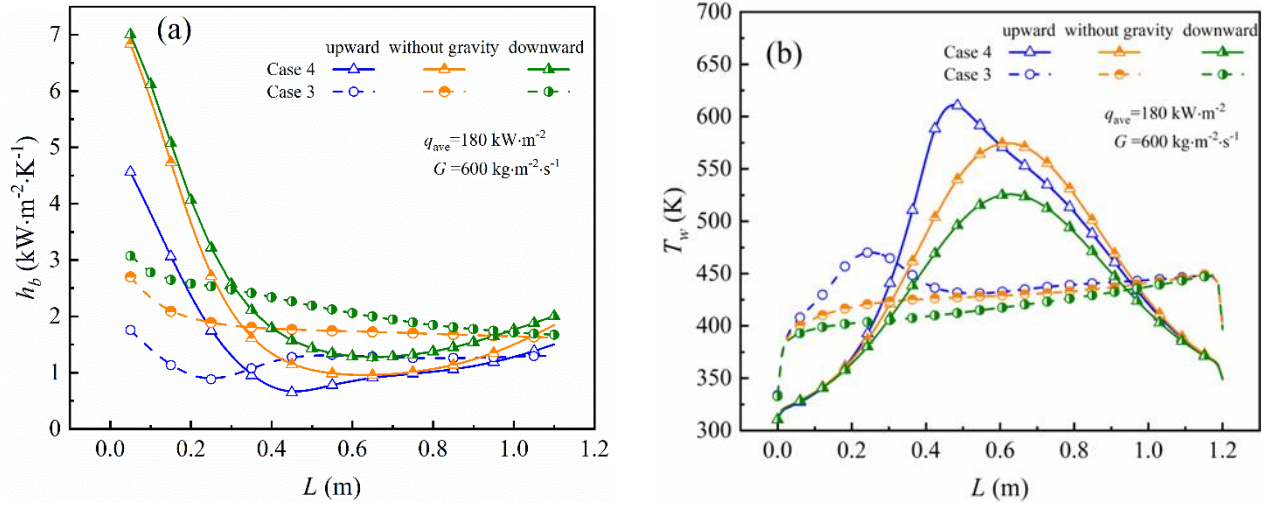


Figure 17. Contours of wall temperature of (a) Case 4 and (b) Case 3 where “a” represents the vertical upward flow, “b” represents the flow without gravity and “c” represents the vertical downward flow at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻².

To further compare the local heat transfer performance in the three flows, the distributions of h_b and T_w at outer wall ($\theta = 0^\circ$) in Case 4 and Case 3 at conditions of $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻² are presented in Figure 18. The h_b in downward flow and the flow without gravity is higher than that in vertical upward flow along the flow direction in Case 4 and Case 3. Under the axially uniform heat-flux condition (Case 3), both h_b and T_w vary along the flow direction without significant fluctuations in downward flow and the flow without gravity. For axially non-uniform heating (Case 4), the minimum value of h_b and the maximum value of T_w both appear in the middle of the tube where the heat flux is the largest in downward flow and the flow without gravity. It can be inferred that heat transfer deterioration does not occur where T_b is near T_{pc} in downward flow and the flow without gravity under Case 4 and Case 3. The maximum T_w in the upward flow is about 85.4 K and 22.6 K higher than that in downward flow in Case 4 and Case 3, respectively, which represents a greater axial temperature gradient in upward flow. The larger the axial temperature gradient of the wall is, the greater the thermal stress and strain become, resulting in an adverse effect on the safety and lifespan of the equipment.



506

507 **Figure 18.** Distributions of (a) local heat transfer coefficient and (b) wall temperature in upward, downward
 508 flows and the flow without gravity at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and
 509 $q_{ave} = 180$ kW·m⁻².

510

511 The distributions of velocity and turbulent kinetic energy on the cross-section of where $T_b/T_{pc} = 1$
 512 (corresponding to $z \approx 367$ mm in Case 4, and $z \approx 275$ mm in Case 3) under the three flows at conditions of
 513 $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻² are shown in Figure 19. Under the condition of upward flow, the
 514 direction of buoyancy force is consistent with the flow direction and there exists a velocity peak in the near-
 515 wall region. In the downward flow, the buoyancy force is opposite to the flow direction, and it can be seen the
 516 velocity in the near-wall region is smaller than that in the flow without gravity. The turbulent kinetic energy is
 517 suppressed in the near-wall region for upward flow, which is not observed for vertical downward flow and the
 518 flow without gravity. As analysed above, the suppression of turbulent kinetic energy may be an important
 519 factor for the heat transfer deterioration.

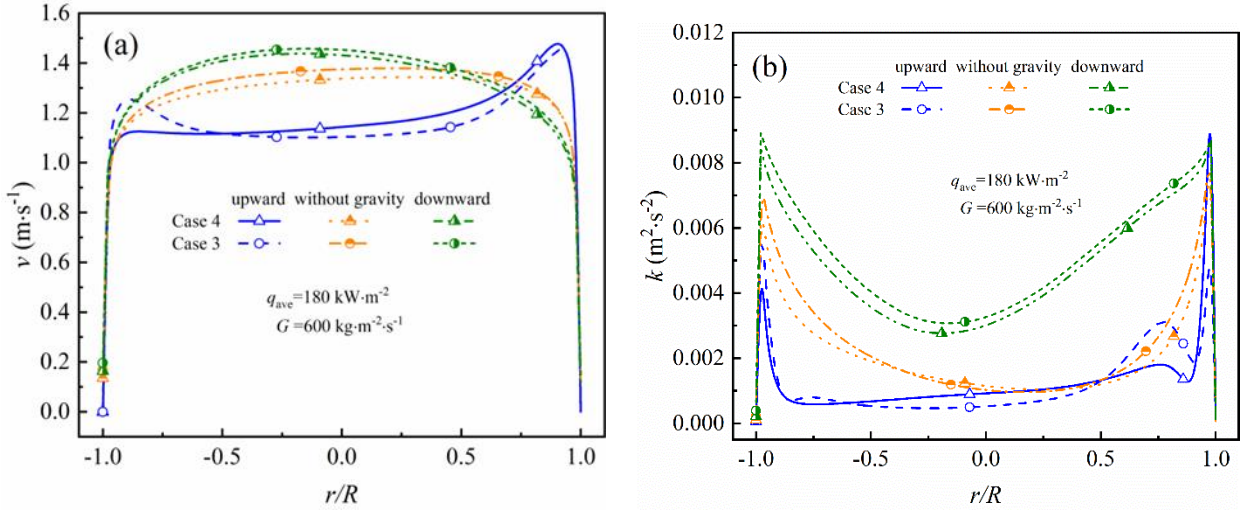


Figure 19. Radial distributions of (a) velocity, (b) turbulent kinetic energy on the cross-section of where $T_b/T_{pc} = 1$ corresponding to $z \approx 367$ mm in Case 4, $z \approx 275$ mm in Case 3 in upward and downward flows as well as the flow without considering gravity at conditions of $T_{in} = 304$ K, $P = 8$ MPa, $G = 600$ kg·m⁻²·s⁻¹ and $q_{ave} = 180$ kW·m⁻².

4. Conclusions

The overall and local heat transfer and flow characteristics of SCO₂ in vertical tube under the conditions of uniform heating, semi-circumferentially uniform heating, semi-circumferentially non-uniform heating and semi-circumferentially and axially non-uniform heating are numerically studied. The effects of heat flux, mass flux, and flow direction on the thermal-hydraulic characteristics of SCO₂ are investigated under conditions of $q_{ave} = 150$ to 210 kW·m⁻² and $G = 400$ to 700 kg·m⁻²·s⁻¹.

The overall heat transfer coefficient of axially non-uniformly heat-flux distribution is 1.2% to 20.5% higher than that of axially uniformly heat-flux distribution under conditions of $q_{ave} = 150$ to 210 kW·m⁻² and $Re = 4.4 \times 10^4$ to 7.8×10^4 . The overall heat transfer coefficient increased under axially non-uniform heat-flux condition, while severe local heat transfer deterioration and steep wall temperature gradient also appear, especially under high heat-flux condition. The T_b of the cross-section where heat transfer deterioration occurs in the axially non-uniformly heated tube is 1.6 K to 3.3 K higher than that of axially uniform heat flux distribution. Increasing mass flux and reducing heat flux could significantly alleviate heat transfer deterioration. The heat transfer deterioration cannot be predicted accurately only from the wall temperature inflection point under axially non-uniform heat-flux condition. Under circumferentially non-uniform heating condition, the heat transfer performance of the heated wall is better than that of the unheated wall.

542 The Bo^* could predict the buoyancy effect under non-uniform heat-flux conditions in the vertical tube. The
543 buoyancy effect changes the cross-sectional velocity distribution, resulting in the difference of the turbulence
544 of the fluid in the near-wall region under different non-uniform heat-flux conditions, which is closely related to
545 the heat transfer performance of SCO_2 . The suppression of turbulent kinetic energy and asymmetric distribution
546 of turbulent production rate leads to severe heat transfer deterioration under axially non-uniform heating
547 condition. The large heat transfer resistance caused by thick momentum and thermal boundary layer near the
548 pseudocritical region should be an important reason for the heat transfer deterioration of SCO_2 under non-
549 uniform heat-flux conditions. A thicker momentum and thermal boundary layer can be observed at the cross-
550 section where more severe heat transfer deterioration occurs under axially non-uniform heating.

551 For axially and circumferentially non-uniform heating, the overall heat transfer coefficient of vertical
552 downward flow is 62.0% to 75.4% higher than that of vertical upward flow, and this discrepancy would
553 increase with the argument of mass flux and the reduction of heat flux. Vertical downward flow is beneficial
554 to effectively eliminate local heat transfer deterioration and reduce axial temperature gradient, especially for
555 axially non-uniform heat-flux condition. The present work and its conclusions could provide new knowledge
556 of the thermohydraulic characteristics of supercritical CO_2 flows as well as practical guidance on the design
557 and optimisation of relevant components.

558

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563

564 **References**

- 565 [1] L.F. Cabeza, A. de Gracia, A.I. Fernández, M.M. Farid, Supercritical CO_2 as heat transfer fluid: A review,
566 Appl. Therm. Eng. 125 (2017) 799-810.
- 567 [2] A.A. Gkountas, A.M. Stamatelos, A.I. Kalfas, Recuperators investigation for high temperature
568 supercritical carbon dioxide power generation cycles, Appl. Therm. Eng. 125 (2017) 1094-1102.
- 569 [3] F. Crespi, G. Gavagnin, D. Sánchez, G.S. Martínez, Supercritical carbon dioxide cycles for power
570 generation: A review, Appl. Energy 195 (2017) 152-183.
- 571 [4] E.R.Casanova, C. Rubio-Maya, J.J. Pacheco-Ibarra, V.M. Ambriz-Díaz, C.E. Romero, X. Wang,

572 Thermodynamic analysis and optimization of supercritical carbon dioxide Brayton cycles for use with
573 low-grade geothermal heat sources, *Energy Convers. Manage.* 216 (2020) 112978.

574 [5] M.J. Li, H.H. Zhu, J.Q. Guo, K. Wang, W.Q. Tao, The development technology and applications of
575 supercritical CO₂ power cycle in nuclear energy, solar energy and other energy industries, *Appl. Therm.*
576 *Eng.* 126 (2017) 255-275.

577 [6] K. Wang, Y.L. He, Thermodynamic analysis and optimization of a molten salt solar power tower integrated
578 with a recompression supercritical CO₂ Brayton cycle based on integrated modeling, *Energy Convers.*
579 *Manage.* 135 (2017) 336-350.

580 [7] J. Xu, E. Sun, M. Li, H. Liu, B. Zhu, Key issues and solution strategies for supercritical carbon dioxide
581 coal fired power plant, *Energy* 157 (2018) 227-246.

582 [8] S. Dilshad, A.R. Kalair, N. Khan, Review of carbon dioxide (CO₂) based heating and cooling technologies:
583 Past, present, and future outlook, *Int. J. Energy Res.* 44 (2019) 1408-1463.

584 [9] J. Xie, D. Liu, H. Yan, G. Xie, S.K.S. Boetcher, A review of heat transfer deterioration of supercritical
585 carbon dioxide flowing in vertical tubes: Heat transfer behaviors, identification methods, critical heat
586 fluxes, and heat transfer correlations, *Int. J. Heat Mass Transf.* 149 (2020) 119233.

587 [10] D.E. Kim, M.H. Kim, Experimental study of the effects of flow acceleration and buoyancy on heat transfer
588 in a supercritical fluid flow in a circular tube, *Nucl. Eng. Des.* 240 (2010) 3336-3349.

589 [11] D.E. Kim, M.H. Kim, Experimental investigation of heat transfer in vertical upward and downward
590 supercritical CO₂ flow in a circular tube, *Int. J. Heat Fluid Flow* 32 (2011) 176-191.

591 [12] X. Lei, J. Zhang, L. Gou, Q. Zhang, H. Li, Experimental study on convection heat transfer of supercritical
592 CO₂ in small upward channels, *Energy* 176 (2019) 119-130.

593 [13] J.D. Jackson, W.B. Hall, Influences of buoyancy on heat transfer to fluids flowing in vertical tubes under
594 turbulent conditions, in: S. Kakac, D.B. Spalding, *Turbulent Forced Convection in Channals and Bundles*,
595 Hemisphere, New York, 1979.

596 [14] N. Kline, F. Feuerstein, S. Tavoularis, Onset of heat transfer deterioration in vertical pipe flows of CO₂ at
597 supercritical pressures, *Int. J. Heat Mass Transf.* 118 (2018) 1056-1068.

598 [15] M.M. Ehsan, Z. Guan, A.Y. Klimenko, A comprehensive review on heat transfer and pressure drop
599 characteristics and correlations with supercritical CO₂ under heating and cooling applications, *Renew.*
600 *Sust. Energ. Rev.* 92 (2018) 658-675.

601 [16] J. Xie, D. Liu, H. Yan, G. Xie, S.K.S. Boetcher, A review of heat transfer deterioration of supercritical

602 carbon dioxide flowing in vertical tubes: Heat transfer behaviors, identification methods, critical heat
603 fluxes, and heat transfer correlations, *Int. J. Heat Mass Transf.* 149 (2020) 119233.

604 [17] M.R. Xiang, J.F. Guo, X.L. Huai, X.Y. Cui, Thermal analysis of supercritical pressure CO₂ in horizontal
605 tubes under cooling condition, *J. Supercrit. Fluids* 130 (2017) 389-398.

606 [18] C. Li, J. Hao, X. Wang, Z. Ge, X. Du, Dual-effect evaluation of heat transfer deterioration of supercritical
607 carbon dioxide in variable cross-section horizontal tubes under heating conditions, *Int. J. Heat Mass Transf.*
608 183 (2022) 122103.

609 [19] J. Wang, Z. Guan, H. Gurgenci, K. Hooman, A. Veeraragavan, X. Kang, Computational investigations of
610 heat transfer to supercritical CO₂ in a large horizontal tube, *Energy Convers. Manage.* 157 (2018) 536-
611 548.

612 [20] J.F. Guo, M.R. Xiang, H.Y. Zhang, X.L. Huai, K.Y. Cheng, X.Y. Cui, Thermal-hydraulic characteristics
613 of supercritical pressure CO₂ in vertical tubes under cooling and heating conditions, *Energy* 170 (2019)
614 1067-1081.

615 [21] C.R. Zhao, Q.F. Liu, Z. Zhang, P.X. Jiang, H.L. Bo, Investigation of buoyancy-enhanced heat transfer of
616 supercritical CO₂ in upward and downward tube flows, *J. Supercrit. Fluids* 138 (2018) 154-166.

617 [22] K. Wang, X. Xu, Y. Wu, C. Liu, C. Dang, Numerical investigation on heat transfer of supercritical CO₂ in
618 heated helically coiled tubes, *J. Supercrit. Fluids* 99 (2015) 112-120.

619 [23] Y. Li, F. Sun, G. Xie, B. Sunden, J. Qin, Numerical investigation on flow and thermal performance of
620 supercritical CO₂ in horizontal cylindrically concaved tubes, *Appl. Therm. Eng.* 153 (2019) 655-668.

621 [24] C.K. Ho, S.S. Khalsa, A Photographic Flux Mapping Method for Concentrating Solar Collectors and
622 Receivers, *J. Sol. Energy.* 134 (2012) 041004.

623 [25] Y.L. He, K. Wang, Y. Qiu, B.C. Du, Q. Liang, S. Du, Review of the solar flux distribution in concentrated
624 solar power: Non-uniform features, challenges, and solutions, *Appl. Therm. Eng.* 149 (2019) 448-474.

625 [26] T. Luu, D. Milani, R. McNaughton, A. Abbas, Dynamic modelling and start-up operation of a solar-
626 assisted recompression supercritical CO₂ Brayton power cycle, *Appl. Energy* 199 (2017) 247-263.

627 [27] Y.H. Fan, G.H. Tang, X.L. Li, D.L. Yang, S.Q. Wang, Correlation evaluation on circumferentially average
628 heat transfer for supercritical carbon dioxide in non-uniform heating vertical tubes, *Energy* 170 (2019)
629 480-496.

630 [28] Y.H. Fan, G.H. Tang, Numerical investigation on heat transfer of supercritical carbon dioxide in a vertical
631 tube under circumferentially non-uniform heating, *Appl. Therm. Eng.* 138 (2018) 354-364.

- [29] H.Y. Zhang, J.F. Guo, X.Y. Cui, X.L. Huai, Heat transfer performance of supercritical pressure CO₂ in a non-uniformly heated horizontal tube, *Int. J. Heat Mass Transf.* 155 (2020) 119748.
- [30] H.Y. Zhang, J.F. Guo, X.Y. Cui, X.L. Huai, Performance analysis of supercritical pressure CO₂ in several enhanced tubes with non-uniform heat flux, *Appl. Therm. Eng.* 180 (2020) 115823.
- [31] Y. Qiu, M.J. Li, Y.L. He, W.Q. Tao, Thermal performance analysis of a parabolic trough solar collector using supercritical CO₂ as heat transfer fluid under non-uniform solar flux, *Appl. Therm. Eng.* 115 (2017) 1255-1265.
- [32] P. Schwarzbzl, R. Pitz-Paal, M. Schmitz, Visual HFLCAL - a software tool for layout and optimisation of heliostat fields, in: *SolarPACES*, Berlin Germany, 2009.
- [33] C. He, Y. Zhao, J. Feng, An improved flux density distribution model for a flat heliostat (iHFLCAL) compared with HFLCAL, *Energy*. 189 (2019) 116239.
- [34] K. Wang, P.S. Jia, Y. Zhang, Z.D. Zhang, T. Wang, C.H. Min, Thermal-fluid-mechanical analysis of tubular solar receiver panels using supercritical CO₂ as heat transfer fluid under non-uniform solar flux distribution, *Sol. Energy* 223 (2021) 72-86.
- [35] Y. Liu, Y. Dong, L.T. Xie, C.Z. Zhang, C. Xu, Heat transfer enhancement of supercritical CO₂ in solar tower receiver by the field synergy principle, *Appl. Therm. Eng.* 212 (2022) 118479.
- [36] Z. Ying, B. He, L. Su, Y. Kuang, D. He, C. Lin, Convective heat transfer of molten salt-based nanofluid in a receiver tube with non-uniform heat flux, *Appl. Therm. Eng.* 181 (2020) 115922.
- [37] X.Z. Wang, Research on heat transfer performance of external tubular receiver using molten salt as HTF for tower solar plant, Harbin Institute of Technology, Harbin, China, 2013.
- [38] Z.J. Zheng, M.J. Li, Y.L. He, Thermal analysis of solar central receiver tube with porous inserts and non-uniform heat flux, *Appl. Energy* 185 (2017) 1152-1161.
- [39] ANSYS CFX 16.0, solver theory guide, ANSYS Inc.2014.
- [40] F.R. Menter, Two-equation eddy-viscosity turbulence models for engineering applications, *AIAA Journal* 32 (1994) 1598-1605.
- [41] Y. Zhang, Y. Yao, Z. Li, G. Tang, Y. Wu, H. Wang, J. Lu, Low-grade heat utilization by supercritical carbon dioxide Rankine cycle: Analysis on the performance of gas heater subjected to heat flux and convective boundary conditions, *Energy Convers. Manage.* 162 (2018) 39-54.
- [42] V.K. Garg, A.A. Ameri, Two-equation turbulence models for prediction of heat transfer on a transonic turbine blade, *Int. J. Heat Fluid Flow* 22 (2001) 593-602.

662 [43] E.W. Lemmon, M.L. Huber, M.O. McLinden, NIST standard reference database 23: reference fluid
663 thermodynamic and transport properties-REFPROP, version 9.0, 2007.

664 [44] L. Zhang, B. Zhu, X. Wu, J. Xu, Heat transfer characteristics of supercritical pressure CO₂ in a vertical
665 smooth tube, *Proceedings of the CSEE* 39 (2019) 4487-4497.

666 [45] A. Urbano, F. Nasuti, Conditions for the occurrence of heat transfer deterioration in light hydrocarbons
667 flows, *Int. J. Heat Mass Transf.* 65 (2013) 599-609.

668 [46] C.R. Zhao, Q.F. Liu, Z. Zhang, P.X. Jiang, H.L. Bo, Investigation of buoyancy-enhanced heat transfer of
669 supercritical CO₂ in upward and downward tube flows, *J. Supercrit. Fluids* 138 (2018) 154-166.

670 [47] J.D. Jackson, M.A. Cotton, B.P. Axcell, Studies of mixed convection in vertical tubes, *Int. J. Heat Fluid*
671 *Flow* 10 (1989) 2-15.

672 [48] X. Liu, X. Xu, C. Liu, J. Ye, H. Li, W. Bai, C. Dang, Numerical study of the effect of buoyancy force
673 and centrifugal force on heat transfer characteristics of supercritical CO₂ in helically coiled tube at
674 various inclination angles, *Appl. Therm. Eng.* 116 (2017) 500-515.

675 [49] M. Bazargan, M. Mohseni, The significance of the buffer zone of boundary layer on convective heat
676 transfer to a vertical turbulent flow of a supercritical fluid, *J. Supercrit. Fluids* 51 (2009) 221-229.

677 [50] H. Tanaka, A. Tsuge, M. Hirata, N. Nishiwaki, Effects of buoyancy and of acceleration owing to thermal-
678 expansion on forced turbulent convection in vertical circular tubes - criteria of effects, velocity and
679 temperature profiles, and reverse transition from turbulent to laminar-flow, *Int. J. Heat Mass Transf.* 16
680 (1973) 1267-1288.

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