

Quantification of Fuel Chemistry Effects on Burning Modes in Turbulent Premixed Flames

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Abstract

The present work quantifies the impact of fuel chemistry on burning modes using premixed dimethyl ether (DME), ethanol (EtOH) and methane flames in a back-to-burnt opposed jet configuration. The study considers equivalence ratios $0 \leq \Phi \leq 1$, resulting in a Damköhler (Da) number range $0.06 \leq Da \leq 5.1$. Multi-scale turbulence ($Re \simeq 19,550$ and $Re_t \simeq 360$) is generated by means of a cross fractal grid and kept constant along with the enthalpy of the hot combustion products ($T_{HCP} = 1700$ K) of the counterflow stream. The mean turbulent rate of strain exceeds the laminar extinction rate for all flames. Simultaneous Mie scattering, OH-PLIF and PIV are used to identify reactants, mixing, weakly reacting, strongly reacting and product fluids. The relative balance between conventional flame propagation and auto-ignition based combustion is highlighted using suitably defined Da numbers and a more rapid transition towards self-sustained (e.g. flamelet type) combustion is observed for DME. The strain rate distribution on the reactant fluid surface for methane remains similar to the (non-reactive) mixing layer ($\Phi = 0$), while DME and EtOH flames gradually detach from the stagnation plane with increasing Φ leading to stabilisation in regions with lower compressive rates of strain. The study further provides information on the conditions leading to burning mode transitions via (i) multi-fluid probabilities, (ii) structural flow field information and turbulence-flame interactions delineated by means of conditional (iii) velocity statistics and (iv) the rate of strain along fluid iso-contours.

Keywords: Turbulent Premixed Flames, Multi-Fluid Statistics, Fuel Effects, Combustion Chemistry, Damköhler Number Scaling

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Nomenclature

Roman Letters

a	Rate of strain [s^{-1}].
\bar{c}	Reaction progress variable [-].
c	Progress variable; Instantaneous conditioning variable [-].
D	Burner nozzle diameter [m].
Da	Conventional Damköhler number [-].
Da_{ai}	Turbulent auto-ignition Damköhler number [-].
$d_{p,x}$	Al_2O_3 particle diameter x% [m].
e	Strain rate tensor [s^{-1}].
f	Rotated strain rate tensor [s^{-1}].
H	Burner nozzle separation [m].
I	Experimental fluorescence signal intensities [-].
I^\ddagger	Reference signal intensity [-].
Ka	Conventional Karlovitz number [-].
$[k]$	Theoretical concentration of species k [$mol\ m^{-3}$].
L_η	Kolmogorov length scale [m].
L_λ	Taylor microscale [m].
L_I	Integral length scale of turbulence [m].
M	Mixing fluid material surface [-].
N	Total number of images [-].
n	Instantaneous image [-].
\hat{n}	Unit vector of the iso-contour normal [-].
\dot{Q}	Heat release rate [$W\ m^{-3}$].
Re	Reynolds number [-].
Re_λ	Reynolds number based on Taylor scales [-].
Re_t	Turbulent Reynolds number [-].
R	Reactant fluid material surface [-].
\mathbf{R}	Rotation matrix [-].
S	Strongly reacting fluid material surface [-].
S_L	Laminar burning velocity [$m\ s^{-1}$].
\hat{s}	Unit vector of the streamline tangent [-].
T	Temperature [K].
T_{ad}	Adiabatic flame temperature [K].
T_{ai}	Auto-ignition temperature [K].
T_{HCP}	Hot combustion product temperature [K].
T_r	Reactant temperature [K].
U	Flow velocity [$m\ s^{-1}$].
\bar{U}	Mean unconditional axial velocity [$m\ s^{-1}$].
\bar{U}_{\dots}	Mean conditional axial velocity [$m\ s^{-1}$].
u	Axial velocity component [$m\ s^{-1}$].
u^*	Leading edge velocity [$m\ s^{-1}$].
$\sqrt{u'^2}$	Unconditional axial velocity fluctuation [$m\ s^{-1}$].
$\sqrt{u'^2_{\dots}}$	Conditional axial velocity fluctuation [$m\ s^{-1}$].
u_{rms}	Root mean square velocity fluctuation [$m\ s^{-1}$].
\dot{V}_{\dots}	Volumetric flow rate [$m^3\ s^{-1}$].
v	Radial velocity component [$m\ s^{-1}$].
$\sqrt{v'^2}$	Unconditional radial velocity fluctuation [$m\ s^{-1}$].
$\sqrt{v'^2_{\dots}}$	Conditional radial velocity fluctuation [$m\ s^{-1}$].
W	Weakly reacting fluid material surface [-].

x	Axial coordinate [m].
x_s	Distance from origin of first thermal alteration [m].
y	Radial coordinate [m].

Greek Letters

β	Material surface iso-contour [-].
δ_f	Laminar fuel consumption layer thickness [m].
δ_t	Distance between the flame edge and the stagnation plane [m].
ε_r	Rate of dissipation in the reactants [$\text{m}^2 \text{s}^{-3}$].
Λ	Threshold value [-].
λ_B	Batchelor scale [m].
λ_D	Mean scalar dissipation layer thickness [m].
λ_{MF}	Multi-fluid spatial resolution [m].
λ_{PIV}	PIV spatial resolution [m].
ν_r	Reactants kinematic viscosity [$\text{m}^2 \text{s}^{-1}$].
Φ	Equivalence ratio [-].
Θ	Angle of rotation [$^\circ$].
τ_c	Chemical timescale [s].
τ_η	Kolmogorov timescale [s].
τ_{ai}	Auto-ignition delay time [s].
τ_I	Integral timescale of turbulence [s].
ω	Vorticity [s^{-1}].

Sub/super-scripts

0	Alignment at the origin; Initial value.
Φ	Dependency on equivalence ratio.
\ddagger	Reference value.
<i>BTB</i>	Back-to-burnt configuration.
<i>b</i>	Bulk flow motion.
<i>d</i>	Total.
<i>FS</i>	Fluid state.
<i>I</i>	Integral scale; Turbulent.
<i>HCP</i>	Hot combustion products.
<i>i, j</i>	Pixel index.
<i>k</i>	Velocity component.
<i>LN</i>	Lower nozzle.
<i>m</i>	Mixing fluid.
<i>NE</i>	Nozzle exit.
<i>n</i>	Instantaneous image; Normal component.
<i>p</i>	Product fluid.
<i>q</i>	Extinction conditions.
<i>r</i>	Reactant fluid.
<i>s</i>	Strongly reacting (flamelet) fluid.
<i>T</i>	Turbulent.
<i>t</i>	Tangential component.
<i>UN</i>	Upper nozzle.
<i>w</i>	Weakly reacting fluid.

1 1. Introduction

2 Novel combustion devices increasingly operate in the absence of distinct
3 flame fronts, e.g. via distributed or flameless combustion modes, to fulfil in-
4 creasingly stringent regulations on NOx and particulate emissions [1]. The
5 approach has potentially significant advantages in terms of reduced emissions
6 and improved fuel efficiency resulting in increased range. The operating con-
7 ditions often exceed conventional extinction criteria and require thermal sup-
8 port, e.g. from exhaust gases [2], for sustained chemical activity. Practical
9 examples that realise stable fuel-lean operation at low Damköhler numbers
10 (Da) include flameless oxidation gas turbine engines [3] and industrial fur-
11 naces [4]. Minamoto et al. [5] investigated the reaction zone structure under
12 flamelet and moderate or intense low-oxygen dilution (MILD) conditions us-
13 ing direct numerical simulation (DNS). The data showed the coexistence of
14 thin and fragmented flamelets with spatially distributed chemical reactions.
15 The broadening or fragmentation of a reaction zone is strongly dependent
16 on the chemical timescale. The flamelet assumption is valid with reasonable
17 accuracy for Karlovitz numbers (Ka) < 10 [6]. A mixture with low reactivity
18 (e.g. low equivalence ratios or high dilution) requires substantially reduced
19 turbulence [7] to move the reaction zone away from a bimodal structure,
20 i.e. the spatial extent of intermediate products becomes statistically relevant.
21 The blending of hot diluents with very fuel rich or lean mixtures can lead
22 to conditions where the chemical timescale dominates the interaction with
23 the turbulent flow [8] and, accordingly, the combustion chemistry assumes a
24 decisive role [5, 6, 9]. Duwig et al. [10] investigated reaction zone broadening
25 of vitiated methane/air jet flame with significant differences for lean ($\Phi =$
26 0.4, 0.8) and rich ($\Phi = 6.0$) conditions observed. A deep penetration of the

27 CH and HCO radicals into the OH layer as well as a CH layer broadening
28 of up to an order of magnitude compared to the laminar thickness was ob-
29 served by Zhou et al. [7, 11–13]. Goh et al. [14] investigated the transition to
30 flameless oxidation of fuel lean premixed JP-10 (exotetrahydrodicyclopent-
31ADIENE) flames using a back-to-burnt (BTB) opposed jet configuration and
32 compared results to corresponding twin flames approaching extinction [15].
33 Hampp and Lindstedt [16, 17] found that high Da counterflow flames detach
34 from the stagnation plane and anchor in low compressive strain regions. Such
35 flames exhibit a clear flamelet-like structure with steep scalar gradients and
36 a distinct dilatation direction. By contrast, low Da combustion precluded
37 conventional flame propagation leading to strong deviations from bimodal
38 structures and OH gradients well below the (theoretical) flamelet limit due
39 to turbulent transport. Reaction progress can also become increasingly de-
40 pendent on external enthalpy sources [18, 19].

41 External enthalpy stabilised combustion exhibits complex turbulence-
42 chemistry interactions that may lead to a competition between auto-ignition
43 related oxidation and conventional flame propagation [17, 20]. While lam-
44 inar burning properties of many hydrocarbon fuels are similar, the auto-
45 ignition delay time can vary by orders of magnitude. Consequently, fuel
46 chemistry effects come to the fore under low Da conditions, yet there is a
47 lack of understanding of the associated turbulence – chemistry interactions.
48 Sabia et al. [21, 22] investigated propane auto-ignition for a wide range of
49 MILD conditions and noted a strong inert gas dilution effect. Fuel lean pre-
50 vaporised acetone, ethanol and n-heptane were investigated by Ye et al. [23]
51 in a MILD combustor with distinct differences in flame stability observed.
52 Wang et al. [24, 25] explored the extinction characteristics of premixed and
53 non-premixed DME and ethanol (EtOH) flames in an opposed jet geometry

54 with DME showing a higher strain resilience than EtOH.

55 The present study quantifies the impact of combustion chemistry on
56 burning mode transitions by the use of DME, EtOH and CH₄ over a wide
57 range of Karlovitz ($3.3 \cdot 10^{-3} \leq Ka^{-1} \leq 0.27$) and Damköhler ($6.0 \cdot 10^{-2} \leq$
58 $Da \leq 5.1$) numbers using an opposed jet configuration. The chemical
59 timescale is varied via the mixture stoichiometry ($0.20 \leq \Phi \leq 1.0$) with
60 the turbulence ($Re_t \simeq 361 \pm 12$) and counterflowing hot combustion prod-
61 ucts ($T_{HCP} = 1700$ K) kept constant. The mixing layer case with $\Phi = 0.0$ is
62 also investigated. The choice of DME and EtOH is based on their relevance
63 as alternative automotive fuels [26, 27], similar laminar flame properties and
64 significantly different auto-ignition characteristics [28] with their relatively
65 well established chemistries [29] an advantage. The methane flames provide a
66 reference point. The impact of the fuel reactivity is investigated via a multi-
67 fluid description [17] by means of simultaneous Mie scattering, hydroxyl –
68 planar laser induced fluorescence (OH-PLIF) and particle image velocimetry
69 (PIV). The analysis extends bimodal descriptions [30] by including a wider
70 range of fluid states (i.e. reactants, products, mixing, weakly and strongly
71 reacting fluids) and has been found particularly useful when delineating low
72 Da combustion [16, 17, 31]. The study further provides information on the
73 conditions leading to burning mode transitions via (i) multi-fluid probabil-
74 ities, (ii) structural flow field information and turbulence-flame interactions
75 delineated by means of conditional (iii) velocity statistics and (iv) the rate
76 of strain along fluid iso-contours. (v) A Da based analysis is performed to
77 delineate conventional and auto-ignition combustion modes.

78 2. Experimental Configuration

79 The current back-to-burnt (BTB) opposed jet configuration has distinct
80 advantages for the systematic investigation of burning mode transitions from
81 flamelet-like structures towards distributed reaction zones. These include (i)
82 relatively well developed turbulence [32, 33]; (ii) separate control of chemical
83 and turbulent timescales [34]; (iii) flame stabilisation related to the intrinsic
84 aerothermochemistry [16, 17, 37]; and (iv) a well controlled burnt gas state
85 can be used to stabilise flames beyond conventional extinction points [17].

86 2.1. Burner Configuration

87 The original opposed jet facility was developed by Geyer et al. [36].
88 Geipel et al. [32] introduced cross fractal grids (CFGs) and Goh et al. [14,
89 15, 35] optimised the nozzle geometry to remove low frequency jet instabil-
90 ities [32, 38]. The use of CFGs [32] induces a turbulent flow field with a
91 multi-scale character [37] and realises a high turbulence to bulk strain ratio
92 while eliminating bulk flow instabilities [35]. The current optimal burner (see
93 supplementary material) is identical to that used by Hampp et al. [16, 17, 39]
94 with the nozzle separation (H) set to one nozzle diameter ($D = 30$ mm).

95 Two separate gas mixing systems provide the reactants, i.e. dried and
96 filtered Howden air, DME (99.9%), CH₄ (99.9%), H₂ (99%) and CO₂ (99%).
97 The gas flow rates are supplied with an uncertainty < 0.8% at full scale [32]
98 and prevapourised ethanol is supplied using the liquid fuel system [14]. Co-
99 flow velocities are set to 30% of the upper nozzle (UN) bulk velocity [32].

100 2.2. Flow Conditions

101 The premixed fuel / air mixtures are injected through the upper nozzle
102 and stabilised against hot combustion products (HCP) emerging from the

Table 1: Experimental Conditions. FBA – Flash Back Arrestor, FSM – Flame Stabilising Mesh; Dil. – Dilution; NL – Nozzle Length; Cross Fractal Grid (CFG) [16].

	UN Conditions Unburnt Reactants		LN Conditions Hot Combustion Products
\dot{V}_{UN}	$7.07 \times 10^{-3} \text{ m}^3 \text{ s}^{-1}$ (293 K)	\dot{V}_{LN}	$3.10 \times 10^{-3} \text{ m}^3 \text{ s}^{-1}$ (293 K)
$U_{b,UN}$	11.2 m s^{-1} (320 K)	$U_{b,LN}$	24.0 m s^{-1} (1700 K)
Fuel	DME, EtOH, CH ₄	Fuel	H ₂
Φ_{UN}	0.0 – 1.0	Φ_{LN}	1.0
T_r	320 K	T_{HCP}	1700 K
Grid	CFG	Grid	FBA and FSM
NL	50 mm	NL	100 mm
Re	$\sim 19,550$	Dil.	22% by volume of CO ₂
Re_t	~ 361		

103 lower nozzle. The flow conditions are listed in Table 1.

104 2.2.1. Upper Nozzle Conditions

105 The CFG is installed 50 mm upstream of the UN exit providing a rela-
106 tively well developed turbulent flow field [32]. The equivalence ratio of the
107 premixed fuel / air mixtures is varied from $0.20 \leq \Phi \leq 1.0$ with the mixing
108 layer ($\Phi = 0.0$) also investigated. The reactants are injected with a constant
109 bulk velocity of $U_b = 11.2 \text{ m s}^{-1}$ leading to bulk and turbulent strain rates
110 of $a_b = 2 \cdot U_b/H \simeq 750 \text{ s}^{-1}$ and $a_T \geq 3200 \text{ s}^{-1}$, respectively [17]. In order
111 to avoid re-condensation of ethanol, reactants are preheated to $T_r = 320 \text{ K}$
112 to raise the vapour pressure well above the maximum fuel mole fraction.
113 The current flow conditions realise a constant $Re_t \simeq 361 \pm 12$ determined
114 based on the integral length scale of turbulence ($L_I = 3.9 \pm 0.2 \text{ mm}$) and
115 velocity fluctuations ($u_{rms} \simeq 1.6 \text{ m s}^{-1}$) measured using hot wire anemom-
116 etry at the UN exit [19]. The u_{rms} is within 10% of that measured with
117 PIV at 1 mm from the UN nozzle exit (for locally axisymmetric turbulence
118 $u_{rms} = 1/3(u' + 2 \cdot v')$, where u' and v' are the axial and radial fluctuations).

119 *2.2.2. Lower Nozzle Conditions*

120 The lower nozzle burnt gas state is obtained using premixed stoichiomet-
121 ric H₂/air flames diluted with 22% CO₂ by volume prior to combustion to
122 control the temperature. The flames are stabilised on a mesh that is installed
123 100 mm upstream of the nozzle exit [16]. The nozzle exit temperature is set
124 to $T_{HCP} = 1700 \pm 3.5$ K, measured via a 50 μm type R thermocouple. The
125 stagnation plane is located in the proximity of the burner centre by matching
126 jet nozzle momenta via a burnt gas velocity of 24 m s⁻¹. The residence time
127 (~ 6 ms) of the burnt gases in the lower nozzle provides hot combustion
128 products in close-to chemical equilibrium with major species concentrations
129 accounting for > 99.8% of the total ($[\text{N}_2] = 3.63$, $[\text{H}_2\text{O}] = 1.91$, $[\text{CO}_2] = 1.56$,
130 $[\text{CO}] = 4.38 \times 10^{-2}$ $[\text{O}_2] = 2.85 \times 10^{-2}$, $[\text{H}_2] = 1.55 \times 10^{-2}$ mol m⁻³). The con-
131 stant thermochemical state provides a constant $[\text{OH}]^\ddagger \approx 8.82 \times 10^{-3}$ mol m⁻³
132 and a clearly detectable fluorescence intensity (I^\ddagger) with a signal to noise ratio
133 (SNR) = 3.5 [17]. The measured I^\ddagger and computed $[\text{OH}]^\ddagger$ provide the refer-
134 ence values used to calibrate the OH intensity segregation technique [17].

135 *2.3. Measurement Setup, PIV and Image Preparation*

136 The diagnostic setup, summarised below, has been discussed by Hampp
137 and Lindstedt [17] including comprehensive uncertainty and error analy-
138 ses. Simultaneous Mie scattering, PIV and OH-PLIF measurements are
139 conducted. Two light sheets (532 nm and 281.7 nm) with a height of 1D and
140 thickness < 0.25 mm and < 0.50 mm, respectively, are superimposed. Mie
141 scattering is recorded using a LaVision ProX 4M camera (2 × 2 binning)
142 equipped with a Tokina AF 100 mm lens (f/2.8) and narrow bandpass filter
143 (3 nm centred at 532 nm). A dichroic beam splitter is used to spectrally
144 separate the OH fluorescence signal from the Mie scattering. OH-PLIF is

145 recorded via a LaVision Imager Intense camera connected to an intensified
146 relay optics (LaVision IRO) unit that is equipped with a 105 mm ultravi-
147 olet lens (f/2.8). A bandpass filter with a transmissivity $> 85\%$ from 300
148 – 320 nm and an optical density > 5 to block the laser lines is used. A
149 temporal separation of 25 μs between the PIV laser pulses is found ideal to
150 minimise spurious vectors. The OH-PLIF images are obtained from the first
151 pulse. Aluminium oxide powder ($d_{p,50} = 0.44 \mu\text{m}$ and $d_{p,90} = 1.7 \mu\text{m}$) is used
152 as a velocity tracer with both nozzles seeded separately. Cross-correlation
153 PIV with decreasing interrogation region size (128×128 to 48×48 with
154 a 75% overlap) is applied using LaVision Davis 8.1. A high accuracy mode
155 with adaptive interrogation window shape modulation is used for the final
156 pass [40]. The velocity field is resolved by 115×88 vectors, providing a spac-
157 ing of 0.30 mm and spatial resolution of $\lambda_{PIV} = 0.60 \text{ mm}$ [17]. A control
158 PIV calculation with $\lambda_{PIV} = 0.30 \text{ mm}$ (i.e. 24×24 , 50% overlap) showed
159 differences in the velocity statistics $< 0.5\%$. This suggests a modest impact
160 of the spatial resolution on conditioning based on fields in close proximity. A
161 minimum of 3000 realisations per condition are recorded to achieve statistical
162 independence. Pre-processing steps of the recorded images (i.e. alignment,
163 data reduction, noise reduction, shot-to-shot intensity fluctuations and white
164 image correction) were described by Hampp et al. [17, 39].

165 3. Chemical Timescales and Limiting Conditions

166 The current experimental conditions provide both thermally supported
167 and self-sustained flames [16, 17]. Self-sustained flames at high Da flames
168 detach from the stagnation plane and anchor in low strain regions. With
169 reducing Da , the strain acting on the flame surface increasingly results in

170 local extinction, with global extinction prevented by the turbulent transport
171 of HCP fluid across the stagnation plane [14, 16, 41]. Chemical timescales
172 and laminar flame properties are determined computationally to support the
173 analysis of experimental data in terms of non-dimensional groups (e.g. Da
174 numbers). The latter are determined based on the properties at the upper
175 nozzle exit to provide an indication of the expected regime of turbulence–
176 chemistry interactions, e.g. Da , and degree of local extinction. The reaction
177 mechanism of Park [29] is used in combination with the hydrogen chemistry
178 of Burke et al. [42]. The mechanism was validated against laminar burning
179 velocities [43–45] and species profiles [46, 47] by Park [29] and auto-ignition
180 delay times (τ_{ai}) by Hampp [39] with good agreement. The calculations
181 are further validated here for flame extinction using experimental data from
182 Wang [25] for DME and EtOH and Law et al. [48] for CH_4 with satisfactory
183 agreement as shown in the supplementary material.

184 3.1. Strained Laminar Flame Extinction Points

185 Self-sustained high Da flames in the BTB configuration decouple from
186 the hot combustion products provided by the opposing (lower nozzle) jet
187 and conventional aerothermochemistry criteria apply. A minimum (critical)
188 integrated heat release rate ($\int \dot{Q}_q$) is required for self-sustained flame prop-
189 agation [17]. Values of $\int \dot{Q} < \int \dot{Q}_q$ lead to the extinction of conventional
190 flames in both the twin flame and back-to-burnt configuration. The ther-
191 mochemical state (e.g. peak temperature and peak species concentrations)
192 is well correlated with $\int \dot{Q}$ and segregate self-sustained from thermally sup-
193 ported burning in BTB flames. Well resolved (i.e. > 25 cells across the CH
194 peak) strained laminar counterflow calculations [49] provide the chemical
195 timescales and flame properties as a function of equivalence ratio and fuel

Table 2: Twin flame extinction point conditions for premixed fuel/air twin flames.

Fuel		DME		EtOH		CH ₄	
Φ	–	0.60	1.0	0.60	1.0	0.60	1.0
a_q	s ⁻¹	575	3000	400	2400	550	1900
T_q	K	1515	1759	1495	1751	1531	1837
$[\text{OH}]_q/[\text{OH}]^\ddagger$	–	3.5	5.8	3.0	5.4	3.2	4.5
$\int \dot{Q}_q$	MW m ⁻²	0.23	0.75	0.18	0.68	0.22	0.62

206 type. The extinction strain (a_q) and peak OH concentration at extinction
 207 $[\text{OH}]_q$ are determined in a twin flame configuration with results summarised
 208 in Table 2 along with the peak temperature at extinction (T_q) and $\int \dot{Q}_q$. At
 209 $\Phi = 0.20$ the extinction strain is significantly lower than the bulk strain and
 200 corresponding cases are excluded. The laminar burning velocity (S_L) and
 201 laminar flame thickness (δ_f , i.e. the 5 – 95% fuel consumption layer [50]) are
 202 determined using a BTB configuration (corresponding to the experiment)
 203 with results summarised in Table 3. As the Da is reduced, combustion is
 204 increasingly supported by the hot combustion products.

205 3.2. Flame Parameters and Dimensionless Groups

206 The Re_t is maintained constant in the current study, while the Damköhler
 207 and Karlovitz numbers, see Eq. (1), are varied by means of the chemical
 208 timescale (τ_c) through the equivalence ratio and fuel type.

$$Da = \frac{\tau_I}{\tau_c} ; \quad Ka^{-1} = \frac{\tau_\eta}{\tau_c} \quad (1)$$

209 The integral timescale of turbulence (τ_I) is based on the measured u_{rms}
 210 = 1.6 m s⁻¹ and $L_I = 3.9$ mm, while the chemical timescale is obtained
 211 from the calculations as defined in Eq. (2). Kolmogorov timescales ($\tau_\eta \simeq$
 212 135 μ s) and length scales ($L_\eta = \nu_r^{3/4} \cdot \varepsilon_r^{-1/4} \simeq 48$ μ m) are based on the

213 rate of dissipation ($\varepsilon_r = u_{rms}^3 \cdot L_I^{-1}$) and the kinematic viscosity (ν_r) in the
 214 reactants, see Table 3.

$$\tau_I = \frac{L_I}{u_{rms}}; \quad \tau_c = \frac{\delta_f}{S_L}; \quad \tau_\eta = \sqrt{\frac{\nu_r}{\varepsilon_r}} \quad (2)$$

215 The resulting ranges of Damköhler numbers and Karlovitz numbers are
 216 $6.0 \cdot 10^{-2} \leq Da \leq 5.1$ and $3.3 \cdot 10^{-3} \leq Ka^{-1} \leq 0.27$, respectively. The Taylor
 217 microscale and Reynolds number follow from $L_\lambda = 10^{1/2} \cdot L_\eta^{3/4} \cdot L_I^{1/3}$ and
 218 $Re_\lambda = L_\lambda \cdot u' \cdot \nu_r^{-1}$, respectively, with values listed in Table 3. Conventional
 219 regime diagrams, see Fig. 1, provide an estimate of the expected burning
 220 modes. The conditions cover the range $3.2 \leq u_{rms}/S_L \leq 40$ with the intense
 221 turbulence regime limit $u_{rms}/S_L \simeq 19$ [51].

222 The mean turbulent strain exceeds the laminar extinction strain for all
 223 cases (see Table 2). This suggests that HCP support, which can cause auto-
 224 ignition, is required to sustain combustion beyond the conventional (strained
 225 twin flame) extinction limit. The ignition delay times (τ_{ai}) and the resulting
 226 auto-ignition based $Da_{ai} = \tau_I \cdot \tau_{ai}^{-1}$, evaluated at initial temperatures of
 227 1300, 1500 and 1700 K, are plotted in Fig. 2 and listed in Table 3. The
 228 temperature range approximately covers the onset of high temperature chain-
 229 branching up to that of the HCP. The trend of the auto-ignition delay times
 230 for DME and EtOH (i.e. shorter τ_{ai} for DME compared to EtOH at low and
 231 intermediate initial temperature and vice versa at higher temperatures) was
 232 also observed by Tingas et al. [52, 53]. While approximate, the data suggests
 233 that Da_{ai} is larger than the flame based Da for temperatures above 1500 K
 234 and that conventional (e.g. flamelet-based) combustion can be (partially)
 235 expected for $\Phi \geq 0.60$ for DME and $\Phi \simeq 1.0$ for methane. The crossing
 236 of the Da and Ka^{-1} lines with the Da_{ai} symbols indicate the presence of a

237 burning mode related to auto-ignition within the most energetic and smallest
238 turbulence scales, respectively. A strong fuel dependency is also evident in
239 the burning mode transition. The fuel reactivity is directly related to, and
240 fully consistent with, the corresponding cetane number for DME or RON /
241 MON numbers of EtOH and CH₄ as well as the spontaneous auto-ignition
242 temperatures. DME is a diesel-like fuel with a high cetane number of 55 –
243 60 [54], while EtOH and methane exhibit high octane numbers, i.e. RON
244 / MON of 110 / 90 [55] and > 120 [56], respectively. Spontaneous auto-
245 ignition temperatures vary significantly at 508 [57], 642 [58], 868 K [59] for
246 DME, EtOH and CH₄, respectively.

247 **4. Multi-Fluid Post-Processing**

248 Turbulence – chemistry interactions in low Da flows cannot be quantified
249 by bimodal (i.e. reactants and products) descriptions as intermediate states
250 become statistically significant [5]. Furthermore, auto-ignition based com-
251 bustion may become significant or dominant. Therefore we recently adopted
252 a multi-fluid description that incorporates a wider range (e.g. mixing, weakly
253 and strongly reacting fluids) of states [16, 17]. The methodology, briefly out-
254 lined below, was found instrumental for low Da flows and is here used to
255 quantify fuel chemistry effects on burning mode transitions.

256 The current multi-fluid analysis uses a conventional PIV tracer particle
257 based density segregation technique, e.g. [37, 60, 61], in combination with a
258 threshold based OH intensity band classification. The first threshold is based
259 on experimental observations and the second on well established combustion
260 theories (see below). For the conditions of interest, i.e. fuel lean combus-
261 tion and the presence of OH in regions with $T \geq 1200$ K, a linear relation

Table 3: Summary of turbulent and chemical conditions to derive the turbulent Reynolds, Damköhler and Karlovitz numbers for DME, EtOH and CH₄ at varying Φ and low strain rate ($a = 75 \text{ s}^{-1}$). Sc is the Schmidt number and $\overline{U}_{r,NE}$ is the mean axial reactant velocity 1 mm away from the UN exit. The reactant temperature was $T_r = 320 \text{ K}$ and the turbulence conditions were evaluated within the reactants.

Fuel	–	DME			EtOH			CH ₄				Air
Φ	–	0.20	0.60	1.0	0.20	0.60	1.0	0.20	0.60	0.80	1.0	0.0
S_L	m s^{-1}	0.04	0.21	0.50	0.04	0.17	0.46	0.04	0.15	0.36	0.39	–
δ_f	mm	1.3	0.46	0.24	1.4	0.55	0.26	1.7	0.85	0.49	0.44	–
T_{ad}/T_r	–	2.8	5.5	7.2	2.8	5.5	7.2	2.7	5.3	6.3	7.0	–
τ_c	ms	33	2.2	0.48	36	2.9	0.57	41	5.7	1.6	1.1	–
$\tau_{ai,1700}$	μs	61 \pm 7.0			60 \pm 1.8			196	219	231	242	–
$\tau_{ai,1500}$	μs	173 \pm 9.0			264 \pm 5.4			832	1053	1128	1194	–
$\tau_{ai,1300}$	ms	1.62	1.11	0.93	3.1	2.7	2.5	11.3	15.2	16.2	17.0	–
$\nu_r \times 10^6$	$\text{m}^2 \text{ s}^{-1}$	17.5	17.0	16.5	17.5	17.0	16.5	17.8	17.9	18.0	18.0	17.9
Sc	–	0.72	0.73	0.73	0.72	0.73	0.73	0.74	0.76	0.77	0.78	0.72
u'	m s^{-1}	1.7	1.8	1.8	1.7	1.6	1.9	1.9	1.9	2.0	2.0	1.7
$\overline{U}_{r,NE}$	m s^{-1}	9.4	9.5	10	10	9.7	10	9.4	9.4	10	9.2	10
u_{rms}	m s^{-1}				1.6 \pm 0.1							
L_I	mm				3.9 \pm 0.2							
τ_I	ms				2.4							
L_λ	μm				287 \pm 4							
L_η	μm				48							
τ_η	μs				135 \pm 2							
$\overline{u^*}/u'$	–	2.7	2.9	3.4	2.7	2.7	3.0	2.7	2.7	2.7	2.7	2.7
δ_l/L_I	–	0.42	0.86	1.0	0.17	0.50	0.93	0.23	0.33	0.70	0.52	0.23
Re_t	–	357	367	378	356	367	378	351	349	347	347	349
Re_λ	–	28	30	31	28	27	33	31	30	32	32	27
Da	–	0.08	1.2	5.1	0.08	1.0	4.4	0.06	0.44	1.5	2.1	–
$Da_{ai,1700}$	–	40 \pm 4.6			40 \pm 1.2			12	11	10	10	–
$Da_{ai,1500}$	–	14 \pm 0.7			9.1 \pm 0.2			2.9	2.3	2.1	2.0	–
$Da_{ai,1300}$	–	1.5	2.2	2.6	0.77	0.88	0.97	0.21	0.16	0.15	0.14	–
$Ka^{-1} \times 10^3$	–	4.1	61	281	3.8	47	237	3.3	24	84	123	–

262 is sufficient (uncertainty < 10%) to link relative theoretical OH concentra-
263 tions to the experimentally observed fluorescence intensity and to identify
264 characteristic intensity bands [17, 39, 62]. The fluids states are:

265 **Reactants:** Fresh reactants emerging from the UN that have not undergone
266 any thermal alteration (i.e. no oxidation or mixing processes). The

267 reactants are detected via a conventional PIV tracer particle based
268 density segregation technique, e.g. [37, 60, 61], capable of detecting
269 multiple and fragmented splines.

270 **Mixing fluid:** A fluid state that has been exposed to a thermal change
271 prior the onset of OH producing chemical reactions (e.g. via mixing
272 with HCP). The mixing fluid is detected by the superimposition of the
273 Mie scattering and OH-PLIF images and is identified in regions with
274 low seeding density and no OH signal.

275 **Strongly reacting fluid:** Regions with a high OH signal intensity consis-
276 tent with self-sustained (e.g. flamelet) burning (see Sec. 4.1). Conven-
277 tional extinction criteria apply [41, 63].

278 **Weakly reacting fluid:** A fluid state with modest levels of OH, e.g. ultra
279 lean flames sustained by support from an external enthalpy source
280 and/or combustion products approaching equilibrium (see Sec. 4.1).

281 **Hot combustion products:** The HCP emerge from the lower nozzle in
282 close-to chemical equilibrium at 1700 K and provide a well defined
283 reference state with a constant $[\text{OH}]^\ddagger$ (see Sec. 3.1) and signal intensity
284 (I^\ddagger) that is used to calibrate the OH band segregation. The maximum
285 measured OH signal of the mixing layer case ($\Phi = 0.0$) is used to
286 separate the HCP from fluid elements containing OH originating from
287 combustion with $\Phi > 0$. The limiting threshold $\Lambda_p = [I/I^\ddagger] = 2.0$ is
288 independent of the UN conditions (i.e. Φ , fuel type). The excess (1.0
289 $< I/I^\ddagger < 2.0$) stems from oxidation of residual reactants in the HCP.

290 The density segregation technique, the delineation of the mixing fluid and
291 the definition of the product fluid threshold (Λ_p) are identical to Hampp and

292 Lindstedt [17] and based on experimental observations. The segregation of
 293 the weakly and strongly reacting fluids is extended, as outlined below, to
 294 accommodate the different fuels.

295 4.1. OH Containing Fluid States

296 The thermochemical state at the twin flame extinction point (e.g. $[\text{OH}]_q$
 297 see Table 2) segregates self-sustained from thermally supported burning in
 298 the current BTB configuration [17]. Regions with an OH fluorescence inten-
 299 sity ratio beyond the non-dimensional (extinction) threshold (see Table 2)
 300 defined by Eq. (3) are consequently assigned to the strongly reacting fluid
 301 (i.e. self-sustained burning).

$$\Lambda_{q(\Phi)} = \frac{[\text{OH}]_{q(\Phi)}}{[\text{OH}]^\ddagger} = \frac{I_{q(\Phi)}}{I^\ddagger} \quad \forall \text{ fuels and } \Phi \geq 0.60 \quad (3)$$

302 The weakly reacting fluid (e.g. thermally supported burning) follows as
 303 $\Lambda_p < I/I^\ddagger < \Lambda_q$ and can originate from (i) ignition events, (ii) decaying
 304 OH concentration in combustion products or (iii) chemically active material
 305 that is diluted by the HCP.

306 4.2. Multi-Fluid Fields and Velocity Conditioning

307 Superposition of the segregated Mie scattering image (i.e. reactants and
 308 thermally altered fluid material) and the three intensity bands of the OH
 309 – PLIF image (i.e. hot combustion products, weakly and strongly react-
 310 ing) yield a quinary multi-fluid image as exemplified in Fig. 3. The spatial
 311 resolution of the multi-fluid analysis was determined based on the smallest
 312 resolved line pair of a post-processed image (i.e. same spatial filters and
 313 denoising functions) of a 1951 USAF target to $\lambda_{MF} \simeq 250 \mu\text{m}$, i.e. L_I
 314 / $\lambda_{MF} \simeq 16$ [17]. The mean scalar dissipation layer thickness ($\lambda_D =$

315 $11.2 \cdot L_I Re_t^{-3/4} Sc^{-1/2}$) [64] is resolved ($\lambda_D / \lambda_{MF} \simeq 2.5$), while the Batchelor
 316 scale ($\lambda_B = L_\eta Sc^{-1/2}$) [65] is under-resolved ($\lambda_B / \lambda_{MF} \simeq 0.22$). Values for
 317 the Schmidt number (Sc) are listed in Table 3. The multi-fluid probabilities
 318 were compiled from the statistics of the prevalence of a given fluid state at a
 319 particular physical location. Multi-fluid velocity conditioning (see Eq. (4))
 320 is used to extend the analysis based on bimodal statistics [35, 37, 60, 61, 66].

$$\begin{aligned}
 \bar{U}_{k,FS,i,j} &= \frac{1}{N} \sum_{n=1}^N c_{FS,n,i,j} \cdot U_{k,n,i,j} \quad \forall i, j \\
 (u'u')_{k,FS,i,j} &= \frac{1}{N} \sum_{n=1}^N c_{FS,n,i,j} \cdot (U_{k,n,i,j} - \bar{U}_{k,FS,i,j})^2 \quad \forall i, j \\
 C_{FS,i,j} &= \frac{1}{N} \sum_{n=1}^N c_{FS,n,i,j} \quad \forall i, j \\
 \sum_{FS} C_{FS} &:= 1
 \end{aligned} \tag{4}$$

321 In Eq. (4), $c_{FS,n}$ is the instantaneous conditioning variable, i.e. unity within
 322 the fluid state (FS) and nil outside, k the velocity component, n the instan-
 323 taneous image, N the number of images (3000) with i and j indices.

324 5. Results and Discussion

325 The multi-fluid probability and conditional velocity statistics are aligned
 326 at the first iso-contour (i.e. $x_s = 0$) where a temperature change is detected
 327 by the density segregation technique (see Fig. 3) to eliminate the movement
 328 and rotation (within an envelope defined by L_I) of the stagnation plane [35].
 329 The multi-fluid probabilities (Sec. 5.2), interface (Sec. 5.3) and conditional
 330 velocity statistics (Sec. 5.4) are evaluated along the theoretical stagnation
 331 point streamline (SPS), i.e. $y = 0$ in Fig. 3. The strain analysis in Sec. 5.6
 332 is conditioned on $y = 0 \pm 1/2 L_I$ to include the radial movement of the

333 stagnation point [32].

334 5.1. Detached and Self-Sustaining Flames

335 The leading edge of the \bar{c} scalar field (i.e. $\bar{c} = 0.02$ iso-contour) is deter-
336 mined from the statistics of the location of the instantaneous reaction onset.
337 Thus, for self-sustained flames $\bar{u}^* = \bar{u}|_{\bar{c}=0.02}$ [67] defines the corresponding
338 turbulent burning velocity. However, for thermally supported low Da flames
339 \bar{u}^* corresponds to the velocity at the leading edge of the mixing layer. The
340 $\Phi = 0.0$ case provides the reference for pure mixing in the absence of heat
341 release and is located at $\bar{u}^*/u' = 2.7$. For the current conditions, the HCP
342 influence becomes reduced for DME flames with $\Phi \geq 0.60$ and for stoichio-
343 metric EtOH flames as shown in row of values of \bar{u}^*/u' listed in Table 3. By
344 contrast, for methane the HCP influence remains dominant for all Φ . The
345 above definition [67] can also be used to determine unique (not influenced
346 by HCP support) turbulent burning velocities in the twin flame opposed
347 geometry.

348 The gradual detachment of the flame from the stagnation plane is illus-
349 trated by the thickness of the layer δ_l defined as the distance between the
350 axial position of the leading edge of \bar{c} and the leading edge of the stagnation
351 plane. The latter is determined from the statistics of the location of the
352 HCP iso-contour. The mixing layer for the $\Phi = 0.0$ case has a thickness
353 $\delta_l/L_I = 0.23$. With increasing Φ (i.e. Da) the leading edge of \bar{c} shifts away
354 from the stagnation plane. For DME, δ_l/L_I increases from 0.42 to 1.0 for
355 $\Phi = 0.20$ to 1.0, respectively. By contrast, the EtOH (CH_4) cases show an
356 attenuated detachment of $\delta_l/L_I = 0.17$ (0.23) and 0.93 (0.52) for $\Phi = 0.20$
357 and 1.0, respectively. All values are listed in Table 3.

358 *5.2. Multi-Fluid Probability Statistics*

359 The reactant fluid probability, conditioned on the SPS and aligned at x_s
360 $= 0$, is found to be fuel and equivalence ratio independent and are hence
361 omitted as the DME case has been discussed by Hampp and Lindstedt [17].
362 By contrast, the mixing fluid probability ($P(m)$) has distinct fuel type and
363 Φ related differences as shown in the top row of Fig. 4. The peak probability
364 for low Da flames (i.e. $\Phi \leq 0.60$) is broadly constant for a given fuel at 0.96
365 for DME, 0.81 for CH_4 and 0.64 for EtOH. The reduced value for EtOH
366 is repeatable [39] and discussed below. The residual percentiles at $x_s =$
367 0 exhibit a low OH signal intensity consistent with the HCP fluid state.
368 Mixing layers below the multi-fluid resolution are not considered. The DME
369 mixing fluid peak probability and its spatial extent reduces significantly with
370 increasing Da (i.e. $\Phi = 1.0$) as chemical active fluid states are favoured. The
371 EtOH mixing fluid shows the same trend, but with an attenuated reduction
372 in $P(m)$ with increasing Da . By contrast, the need for thermal support for
373 the CH_4 flames remains with a peak mixing fluid probability $\sim 70\%$ for the
374 stoichiometric case. This is consistent with the iso-contour based analysis
375 relating to self-sustained flames presented in Sec. 5.1.

376 The weakly reacting fluid probability ($P(w)$) is shown in the middle row
377 of Fig. 4 for all fuels. DME and EtOH show similar distributions with peak
378 values increasing from around 10 to 38% with an increase of Φ from 0.20 to
379 0.60. The shape and spatial variation of $P(w)$ for CH_4 differ substantially
380 from the oxygenated fuels, yet the increase in peak value is approximately
381 maintained (5 – 26%). DME and EtOH show a much stronger decay com-
382 pared to CH_4 as indicated by the probability tails. The phenomena was
383 repeatable [31] and was qualitatively also observed by Shen and Sutton [68].
384 For $Da > 1.0$, the peak of $P(w)$ reduces in favour of a gradual transition to

385 the strongly reacting fluid – particular evident for the stoichiometric DME
 386 case. The weakly reacting fluid probability is only modestly reduced for
 387 EtOH and CH₄ indicating the distinct need of thermal support to initiate
 388 and sustain chemical activity under current turbulence conditions.

389 The strongly reacting fluid probabilities $P(s)$ are depicted in the bottom
 390 row of Fig. 4. At $Da \ll 1$ (i.e. $\Phi = 0.20$) self-sustained combustion is
 391 not observed. With Da of $\mathcal{O}(1)$ at $\Phi = 0.60$ peak probabilities of $P(s) \leq$
 392 20% are obtained. The stoichiometric flames show a substantial increase in
 393 the strongly reacting fluid for DME and EtOH and a modest increase for
 394 CH₄, realising peak probabilities of 72, 47 and 20%, respectively. The trend
 395 is consistent with the determined laminar flame extinction points listed in
 396 Table 2. In the absence of conventional flame propagation, DME is the most
 397 likely fuel to support a transition to an auto-ignition related burning mode
 398 due to the higher Da_{ai} at modest temperatures, see Fig. 2 and Table 3.

399 5.3. Multi-Fluid Interface Statistics

400 The multi-fluid delineation also supports a statistical analysis of inter-
 401 faces between adjacent fluid states along the theoretical stagnation point
 402 streamline. In combination with PIV and streamline calculations, three flow
 403 scenarios were defined: (i) the streamline tangent ($\hat{\mathbf{s}} :=$ positive in flow di-
 404 rection) and interface normal ($\hat{\mathbf{n}} :=$ positive from reactants to HCP) are
 405 aligned, i.e. $\hat{\mathbf{s}} \cdot \hat{\mathbf{n}} > 0.31$, (ii) the opposite with $\hat{\mathbf{s}} \cdot \hat{\mathbf{n}} < -0.31$ and (iii) tangen-
 406 tial flow, i.e. $\|\hat{\mathbf{s}} \cdot \hat{\mathbf{n}}\| < 0.31$ (i.e. $72 - 108^\circ$). The three cases correspond to
 407 transport (i) into a fluid state, (ii) out of it and (iii) sheared or aligned flow
 408 of two fluid states. The methodology was described in detail by Hampp and
 409 Lindstedt [17] and a schematic is provided in the supplementary material.

410 Cases with probabilities $> 1\%$ are shown in Fig. 5. The line weight

411 indicates the likelihood of adjacent fluid state pairs, the arrows indicate the
412 three flow scenarios, the numbers show the likelihood of the corresponding
413 flow scenarios in percent based on the total number of interfaces and the
414 dashed lines show insignificant flow paths. Single-headed arrows point in
415 the flow direction and the double-headed arrow indicates tangential flow.
416 The numbers in Fig. 5 are normalised using all interfaces for each case. In
417 the following we discuss (i) the cumulated interface probabilities of a flow
418 path to quantify the reaction progress, (ii) the dominant flow directions to
419 quantify turbulence – chemistry interactions and (iii) the statistical relevance
420 of adjacent fluid state pairs normalised by the number of multi-fluid images.
421 Consequently, the values in (iii) are larger when compared to (i) as every
422 multi-fluid image contains multiple material transitions and thus interfaces.

423 For $Da < 1$ ($\Phi = 0.20$) the majority of the reactant fluid is adjacent to a
424 mixing fluid layer, i.e. DME with 96%, EtOH with 73% and CH_4 with 88%,
425 and HCP accounting for the residual percentiles. This agrees well with the
426 corresponding peak mixing fluid probabilities adjacent to the origin x_s (see
427 Fig. 3) in Sec. 5.2. The proportion of reactant to mixing fluid interfaces is
428 39, 31 and 40% for DME, EtOH and CH_4 (top row) with the corresponding
429 HCP to mixing fluid interfaces responsible for 42, 34 and 44%.

430 The weakly reacting fluid at $Da < 1$ (i.e. $\Phi = 0.20$ for all fuels) is found
431 almost exclusively ($> 90\%$) adjacent to the HCP fluid with cumulated inter-
432 face probabilities of 17, 21 and 9% for DME, EtOH and CH_4 . This is indica-
433 tive of low OH gradients governed by turbulent mixing or thermal support
434 layers. In general, the multi-fluid structure at $\Phi = 0.20$ is broadly fuel type
435 independent, though a reduced probability of finding weakly reacting fluid
436 adjacent to HCP can be noted for methane.

437 The higher reactivity of DME, compared to EtOH and CH_4 , results in

438 some differences between the fuels for the $Da \simeq 1$ ($\Phi = 0.60$) cases in terms of
439 how strongly reacting fluid may be separated from the reactants via thermal
440 support layers (i.e. mixing or HCP fluid). However, the primary path for
441 the weakly reacting fluid is fuel independent and similar to the lower Da (Φ
442 $= 0.20$) flames. Thus, the weakly reacting fluid is dominated by turbulent
443 transport of HCP into the reactants. Secondary paths arise through the
444 mixing fluid and for EtOH also directly from the reactants. This observation
445 suggests rapid thermal decomposition of EtOH when in contact with HCP.
446 The flame computations suggest that the reaction $C_2H_5OH \rightarrow C_2H_5 + OH$
447 accounts for 11.5 ± 0.1 of the consumption of ethanol and the resulting $[OH]$
448 is sufficient for the detection of the weakly reacting fluid state for all Φ .

449 For stoichiometric cases, the likelihood of detectable ($\lambda_{MF} = 250 \mu m$)
450 reactant adjacent to mixing fluid is reduced to $\sim 25\%$ for DME and EtOH
451 with an interface probability of $\sim 8\%$. By contrast, the corresponding fluid
452 state pair probability remains $> 80\%$ for CH_4 (i.e. 27% of all interfaces) il-
453 lustrating the continuous need for thermal support. A strong augmentation
454 of flamelet-like structures is evident for the stoichiometric DME flame. The
455 probabilities of finding strongly reacting fluid directly adjacent to the reac-
456 tant fluid are 57, 24 and 9% (corresponding to 16, 8 and 3% of all interfaces)
457 for DME, EtOH and CH_4 , respectively. The stoichiometric DME and EtOH
458 flames with high Da detach from the stagnation plane and show a prefer-
459 ential flow direction, e.g. dominant flow path from reactants into strongly
460 reacting fluid as depicted in Figs. 5g – 5h, consistent with a flamelet-like
461 structure. By contrast, the flow across weakly reacting fluid interfaces is
462 mixing dominated with no clear direction evident. The cumulative interface
463 probability leading to strongly reacting fluid (i.e. sum of shared interfaces
464 with all other fluid state) further illustrates the delayed transition towards

465 self-sustained burning of EtOH (34%), and in particular CH₄ (18%), as com-
 466 pared to DME (52%). This is consistent with the fuel reactivity (see Table 2)
 467 and Chen et al. [69] who showed reduced ignition delay and elevated strain
 468 resilience of DME compared to CH₄.

469 5.4. Conditional Velocity Statistics

470 The impact of fuel reactivity on the flow field is further analysed using
 471 conditional multi-fluid velocity statistics. The data are conditioned on the
 472 SPS and aligned at $x_s = 0$. The mean conditional velocities and turbulent
 473 fluctuations are normalised by the mean axial reactant fluid velocity at the
 474 nozzle exit ($\overline{U_{r,NE}}$), see Table 3. A minimum of 75 vectors is used for the
 475 statistical analysis with a maximum change of 10% in the fluctuations within
 476 the last 20% of the sample. The reactant exit velocity is defined as negative.

477 5.4.1. Conditional Reactant Fluid Velocity

478 The conditional mean axial reactant fluid velocities ($\overline{U_{0,r}} / \overline{U_{r,NE}}$) are
 479 depicted, along with the axial ($\overline{u'_{0,r}} / \overline{U_{r,NE}}$) and radial ($\overline{v'_{0,r}} / \overline{U_{r,NE}}$) fluctu-
 480 ations in Fig. 6. No distinct differences are observed between the fuels for
 481 $Da \ll 1$. The impact of differences in the combustion chemistry become
 482 evident for the cases with $\Phi = 0.60$. The higher fuel reactivity of DME
 483 promotes an advanced reaction onset and flame detachment from the stag-
 484 nation plane. The mean velocity at $x_s = 0$ (the location of reaction onset) is
 485 in line with the natural reactant flow direction with $\overline{U_{0,r}} / \overline{U_{r,NE}} = -0.06$. By
 486 contrast, $\overline{U_{0,r}} / \overline{U_{r,NE}} \simeq 0.02$ for EtOH and CH₄ indicating a HCP dominated
 487 reaction onset. With increasing Da (i.e. $\Phi = 1.0$), the reaction onset shifts
 488 further upstream towards the UN with elevated $\overline{U_{0,r}} / \overline{U_{r,NE}} = -0.24, -0.20$
 489 and -0.11 for DME, EtOH and CH₄, respectively. The modest difference

490 between DME and EtOH is consistent with the increased strain resilience of
 491 DME and thus a reduced frequency of local extinction. The trends are also
 492 consistent with the normalised mean velocity ($\overline{u^*}/u'$) at the leading flame
 493 edge, shown in Table 3, with no fuel chemistry effect apparent for $Da \ll$
 494 1 where $\overline{u^*} = 2.7$ as for the $\Phi = 0.0$ mixing layer. For the DME case at
 495 $\Phi = 0.60$ $\overline{u^*}/u'$ increases to 2.9, while the EtOH and CH₄ remain mixing
 496 dominated with 2.7. For the stoichiometric cases, a further increase to 3.4
 497 and 3.0 was observed for DME and EtOH. However, the reaction progress of
 498 the CH₄ remains governed by the HCP. This is consistent with the Da and
 499 Da_{ai} analysis. For example, conventional flame propagation for the methane
 500 cases is unlikely (as indicated by the low Da) and the initial temperature
 501 must exceed 1500 K to realise auto-ignition related burning (see Table 3).
 502 By contrast, auto-ignition related burning of DME is feasible for $T \geq 1300$ K
 503 with EtOH an intermediate case. The axial and radial reactant fluid fluctua-
 504 tions are reduced with increasing Φ , consistent with the gradual detachment
 505 of the reaction onset from the stagnation plane discussed in Sec. 5.1.

506 Reactant fluid pockets are observed at $x_s > 0$ with a probability $\leq 7\%$
 507 due to fragmentation or three dimensional effects. The probability is inde-
 508 pendent of the mixture reactivity and fuel type [17, 39]. However, significant
 509 differences in the preferential flow alignment of these pockets are observed.
 510 For the stoichiometric cases with $Da > 1$, the interlayer is highly exothermic
 511 (e.g. strongly reacting fluid as inferred from Sec. 5.3) and induces a distinct
 512 convective direction of the reactant fluid pocket aligned with the main reac-
 513 tant flow. For DME with $\Phi = 1.0$ this results in an acceleration of the pocket
 514 towards the stagnation plane as indicated by the increasingly negative $\overline{U_{0,r}}$
 515 / $\overline{U_{r,NE}}$ from $0 < x_s/L_I < 0.5$, see top right panel of Fig. 6. It further
 516 causes an alignment of $\overline{U_{0,r}} / \overline{U_{r,NE}}$ with the natural reactant flow direction

517 for DME with $\Phi \geq 0.60$ and for EtOH/CH₄ with $\Phi = 1.0$. By contrast, the
 518 interlayer for $Da < 1$ (i.e. $\Phi = 0.20$) is formed by the mixing fluid and is
 519 hence governed by turbulent mixing across the stagnation plane. Reactant
 520 fluid pockets that align with the reactant fluid flow exhibit significantly (>
 521 50%) reduced axial and radial fluctuations.

522 5.4.2. Conditional Weakly Reacting Fluid Velocity

523 The conditional mean axial weakly reacting fluid velocities ($\overline{U_{0,w}} / \overline{U_{r,NE}}$)
 524 and its axial ($\overline{u'_{0,w}} / \overline{U_{r,NE}}$) and radial ($\overline{v'_{0,w}} / \overline{U_{r,NE}}$) fluctuations are de-
 525 picted in Fig. 7. For flames with $Da < 1$ (i.e. $\Phi = 0.20$), the $\overline{U_{0,w}} / \overline{U_{r,NE}}$
 526 are nearly identical for all fuels. The $\overline{U_{0,w}} / \overline{U_{r,NE}}$ of DME at $\Phi = 0.60$
 527 separate from the EtOH and CH₄ cases and is in line with the UN reac-
 528 tant flow at $x_s = 0$, i.e. negative values. Moreover, attenuated axial velocity
 529 fluctuations are evident that can be attributed to enhanced dilatation. The
 530 weakly reacting fluid velocity of the stoichiometric DME and EtOH cases
 531 coincide and show reduced fluctuations compared to CH₄. The delayed sep-
 532 aration of the reaction onset from the stagnation plane, the continuous re-
 533 quirement for substantial HCP support and, consequently, the attenuated
 534 dilatation effect on the weakly reacting fluid for CH₄ mixtures illustrates a
 535 strong influence of the fuel reactivity. The pronounced lower temperature
 536 reactivity [70] of DME causes advanced dilatation that is readily apparent
 537 for $Da \simeq 1$ (i.e. $\Phi = 0.60$). The conditional mean mixing fluid velocity
 538 is similarly governed by the HCP momentum and independent of the fuel
 539 reactivity for $Da \leq 1$. At $Da > 1$, the stoichiometric DME and EtOH cases
 540 are in line with the UN reactant flow direction at $x_s = 0$, while CH₄ flames
 541 remain strongly affected by the HCP stream. The data is provided in the
 542 supplementary material.

543 *5.4.3. Conditional Strongly Reacting Fluid Velocity*

544 The interface statistics presented in Sec. 5.3 suggest that the strongly
545 reacting fluid can result from small HCP blending fractions depending on
546 the reactivity of the mixture. Self-sustained burning is not realised for any
547 fuel with $\Phi = 0.20$. For $\Phi = 0.60$ the conditional strongly reacting fluid mean
548 velocity ($\overline{U_{0,s}} / \overline{U_{r,NE}}$) for DME ($Da = 1.2$) case is distinctly lower compared
549 to EtOH ($Da = 1.0$) and CH₄ ($Da = 0.44$) as depicted in Fig. 8. For DME
550 the value of $\overline{U_{0,s}} / \overline{U_{r,NE}}$ is lower than the corresponding weakly reacting fluid
551 velocity, while EtOH and CH₄ flames show no clear difference. The DME
552 case further exhibits distinctly reduced (50%) axial velocity fluctuations.
553 For the stoichiometric flames with $Da > 1$ the discrepancies between DME
554 and EtOH vanish and similar values of $\overline{U_{0,s}} / \overline{U_{r,NE}}$ are obtained due to
555 increasingly self-sustained flames detaching from the stagnation plane. The
556 finding is consistent with the similar heat release parameters $\tau = (T_{ad} -$
557 $T_r)/T_r = 6.1$ versus 5.9 and conventional burning properties (e.g. $S_L = 0.50$
558 versus 0.46 m s⁻¹) for DME and EtOH at $\Phi = 1.0$, respectively.

559 However, the likelihood of self-sustained combustion differs for DME and
560 EtOH as shown in Fig. 4. The slightly more negative $\overline{U_{0,s}} / \overline{U_{r,NE}}$ for the
561 stoichiometric DME flame can be explained by the differences in $\overline{u^*}/u'$ and
562 the interface statistics, where a significant proportion of the strongly react-
563 ing fluid shares an interface with the reactant fluid. In contrast to these two
564 fuels, the CH₄ flames show a reduced detachment from the stagnation plane
565 and are consequently subjected to a higher rate of strain (discussed below).
566 The strongly reacting fluid velocity shows a reduced dilatation away from
567 the origin (i.e. $x_s/L_I > 0.5$) and is increasingly influenced by the counter-
568 flowing HCP stream. This leads to substantially increased axial and radial

569 fluctuations for CH₄.

570 5.5. Fuel Effects at Similar Damköhler Numbers

571 The DME flame with $\Phi = 0.60$ ($Da = 1.2$) and an additional CH₄ case
572 featuring $\Phi = 0.80$ ($Da = 1.5$) provide a comparison of two flames with sim-
573 ilar and close-to unity Damköhler numbers. However, the two cases feature
574 distinctly different auto-ignition delay times (factor of ~ 6) with the mini-
575 mum for fresh reactants in contact with the HCP at 1700 K determined to
576 be 62 and 371 μs for DME and CH₄, respectively [39]. This ratio increases to
577 ~ 15 at an initial temperature of 1300 K. Comparisons are shown in Figs. 9
578 and 10. Strong similarities in the multi-fluid probabilities and the multi-fluid
579 conditional velocities are evident. The reactant, mixing and strongly react-
580 ing fluid probabilities show a mean deviation of $< 4\%$ in the range from 0
581 $< x_s/L_I < 1.5$. The mean difference in the weakly reacting fluid probability
582 is $\sim 10\%$ and is consistent with the shorter auto-ignition delay of DME.

583 The mean conditional reactant fluid velocities agree well as shown in
584 Fig. 10. By contrast, the lower (i.e. alignment with the UN reactant flow)
585 mixing and weakly reacting fluid velocity for DME again indicates reduced
586 HCP blending fractions consistent with the different Da_{ai} characteristics.
587 The larger negative conditional strongly reacting fluid velocity at $x_s = 0$
588 and enhanced dilatation at $x_s > 0$ for the CH₄ flame is consistent with the
589 higher heat release of the closer to stoichiometric flame.

590 5.6. Conditional Strain Distribution on Material Surfaces

591 The rate of strain and vorticity are calculated from the instantaneous
592 PIV data via Eq. (5). Conditioning on the multi-fluid material surfaces (β)

593 quantifies the actual rate of strain acting on the reaction zone and augments
 594 the estimated turbulent strain at the nozzle exit.

$$e_{ij} = \frac{1}{2} \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \quad \omega_{ij} = \frac{\partial u_i}{\partial x_j} - \frac{\partial u_j}{\partial x_i} \quad f = \mathbf{R} \cdot e \quad (5)$$

595 The strain rate tensor is rotated by Θ , where \mathbf{R} is the rotation matrix and
 596 Θ is the angle between the iso-contour normal and the SPS. This defines
 597 the normal ($a_n = f_{11}$) and tangential ($a_t = f_{22}$) strain components that
 598 are further, along with the vorticity, conditioned on the fluid state material
 599 surfaces (β) [16]. In addition, the total rate of strain ($a_d|\beta = e_{\beta,11} + e_{\beta,22}$) is
 600 evaluated. The determined instantaneous planar rates of strain are - to some
 601 degree - underestimated due to 3D effects [16]. The analysis is conducted
 602 within $\pm L_I/2$ radially away from the SPS to include the movement of the
 603 stagnation point [32]. In the following, only mean values are discussed with
 604 the PDF shapes shown in Figs. 11 and 12 and the *rms* spread provided in
 605 the supplementary material.

606 5.6.1. Strain Distribution along the Reactant Fluid Surfaces

607 The normal strain evaluated along the reactant fluid iso-contour is de-
 608 picted in Fig. 11. The mean normal compressive strain at $\Phi = 0.20$ is fuel in-
 609 dependent with $a_n|R = -1500 \pm 50 \text{ s}^{-1}$ as the reaction onset is dominated by
 610 mixing with HCP. The fuel dependency becomes apparent as Φ and thus Da
 611 increases. The mean $a_n|R$ for DME reduces by 70% from -1550 to -480 s^{-1}
 612 for $0.08 < Da < 5.1$, as shown in the first row of Fig. 11. By contrast, the
 613 corresponding rate of strain for the methane cases ($0.06 < Da < 2.1$) reduces
 614 by 18% towards lower compressive strain, while $a_n|R$ for EtOH experiences
 615 an attenuation of 58% ($0.08 < Da < 4.4$). The reduction of $a_n|R$ is indica-
 616 tive of a gradual detachment of the $\bar{c} = 0.02$ iso-contour from the stagnation

617 plane with increasing Da as discussed in Sec. 5.1 in the context of the flame
 618 stabilisation point. At high Da , the iso-contour is anchored in regions with
 619 an increased axial velocity, but low compressive strain, which is character-
 620 istic for a reaction onset driven by self-sustained burning. The upstream
 621 shift of the reactant fluid material surface further results in reduced levels of
 622 extensive tangential strain and a strong attenuation in conditional vorticity
 623 levels of 50%. By contrast, the reactant fluid surfaces for the CH_4 cases
 624 show no clear spatial separation from the stagnation plane with high $a_n|R$.
 625 The reaction onset in the proximity of the stagnation plane further causes a
 626 modest increase ($< 10\%$) of $a_t|R$ and a modest reduction ($\sim 13\%$) $\omega|R$ with
 627 increasing Φ . The earlier transition of DME and EtOH to strongly dilating
 628 fluid states, compared to CH_4 , is also reflected in the total rate strain. For
 629 $\Phi = 0.20$, $a_d|R = -760 \pm 20 \text{ s}^{-1}$ and is only modestly affected by the fuel
 630 reactivity. By contrast, the reactant fluid iso-contours for the stoichiomet-
 631 ric DME and EtOH cases are situated in regions of extensive total rates of
 632 strain ($a_d|R \simeq 320$ and 190 s^{-1} respectively), while the corresponding CH_4
 633 iso-contour is found in compressive strain regions with $a_d|R = -420 \text{ s}^{-1}$.
 634 It can further be noticed that DME and EtOH show strong similarities,
 635 while CH_4 shows a broadened PDF that is attributed to a pulsating burning
 636 mode that was previously observed at reduced Re_t by Mastorakos et al. [71].
 637 Moreover, while the DME and EtOH cases with $Da > 1$ (in particular Φ
 638 $= 1.0$) show a clear reduction of a_n with increasing Φ (c.f. $\Phi = 0.0$), the
 639 CH_4 cases show strong similarities for all stoichiometries. The rates of strain
 640 conditioned upon the mixing fluid material surface show similar trends with
 641 values provided in the supplementary material.

642 5.6.2. Strain Distribution along Chemically Active Fluid Surfaces

643 The compressive normal strain conditioned on the weakly reacting fluid
644 material surface ($a_n|W$) is relatively fuel and equivalence ratio independent
645 at $Da \lesssim 1$ (i.e. $\Phi \leq 0.60$) with $a_n|W = -1260 \pm 70 \text{ s}^{-1}$. The corresponding
646 PDFs and values are provided in the supplementary material. However, the
647 corresponding total rate of strain shows an attenuated contracting strain
648 with increasing Φ for all fuels due to increased heat release. The conditional
649 vorticity levels are consistently similar to the values found in the proxim-
650 ity of the stagnation plane ($\omega|W = 2500 \pm 250 \text{ s}^{-1}$). The absence of clear
651 fuel dependent trends suggest a strong HCP influence with the accompany-
652 ing modest heat release at high dilution ratios. This is consistent with the
653 findings of Chakraborty and Swaminathan [72] that showed a lack of flow
654 alignment for low Da flows with reduced heat release.

655 The strongly reacting fluid iso-contour is situated in low compressive
656 strain ($a_n|S$) regions, as depicted in Fig. 12, with $a_n|S$ approximately 16%
657 and 38% lower than the corresponding strain acting on the weakly reacting
658 fluid for all fuels $\Phi = 0.60$ and 1.0, respectively. The modestly higher a_n
659 for DME compared to EtOH (i.e. -700 s^{-1} versus -520 s^{-1}) at $\Phi = 1.0$ is
660 consistent with the increased resilience of DME to strain as indicated by the
661 calculated laminar extinction strain of $a_q = 3000 \text{ s}^{-1}$ and 2400 s^{-1} for DME
662 and EtOH at $\Phi = 1.0$, respectively, and also shown by Wang et al. [24]. Self-
663 sustained DME flames can accordingly exist in regions where EtOH flames
664 are extinguished or rely on thermal support to a greater extent. This is
665 accompanied by a reduced $a_n|W$ of the stoichiometric EtOH flame compared
666 to DME. The CH_4 flame remains located in the proximity of the stagnation
667 plane with a compressive strain of -860 s^{-1} . Overall, the mean $a_n|S$ reduce

668 by 30, 50 and 17% with increasing Φ for DME, EtOH and CH₄.

669 Extensive tangential rates of strain remain approximately constant at
670 $a_t|S$ ($670 \pm 110 \text{ s}^{-1}$) for all cases as increases in dilatation are balanced
671 by the detachment from the stagnation plane. A shift from contracting
672 ($a_d|S = -400 \text{ s}^{-1}$) to dilating strain ($175 \pm 5 \text{ s}^{-1}$) total strain is evident with
673 increasing Φ for DME and EtOH. A distinct reduction ($\sim 80\%$) in $a_d|S$
674 is evident for CH₄. The PDF spread for $a_n|S$ and $a_t|S$ at $\Phi = 0.60$ is
675 within 5% of weakly reacting fluid for all fuels. By contrast, the spread
676 for the stoichiometric cases is 15% lower than that of the weakly reacting
677 fluid. The reduction is attributed to an iso-contour governed by conventional
678 flame propagation rather than turbulent mixing. This is consistent with the
679 vorticity ($\omega|S$) reduction of $\sim 15\%$ and $\sim 30\%$ compared to $\omega|W$ at $\Phi =$
680 0.60 and 1.0, respectively. The less pronounced strain reduction for the CH₄
681 cases is caused by the delayed onset of self-sustained flame propagation.

682 **6. Conclusions**

683 The current study has investigated the impact of fuel chemistry on burn-
684 ing mode transitions in turbulent fuel lean to stoichiometric premixed di-
685 methyl ether, ethanol and methane flames. An opposed jet back-to-burnt
686 configuration with a constant burnt gas state and fractal grid generated
687 multi-scale turbulence was combined with a multi-fluid analysis based on si-
688 multaneous Mie scattering, OH-PLIF and PIV. The Damköhler number was
689 varied through the mixture stoichiometry ($0.20 \leq \Phi \leq 1.0$) and fuel type.
690 Based on a conventional combustion regime diagram, the conditions cov-
691 ered a transition from distributed reaction zones to close-to the corrugated
692 flamelet regime with $0.06 \leq Da \leq 5.1$.

693 Multi-fluid probabilities, conditioned on the reaction onset iso-contour,
694 and interface statistics have been used to quantify the impact of combus-
695 tion chemistry with a more rapid transition to chemically active fluid states
696 and self-sustained burning evident for DME due to its lower auto-ignition
697 temperature and higher resilience to strain. This caused combustion to be
698 anchored in regions of higher reactant fluid velocities for Da numbers around
699 unity. By contrast, the transition from supported to a self-sustained flame
700 propagation was delayed for EtOH and distinctly so for CH_4 . The greater
701 strain resilience of DME led to a slightly higher compressive strain along the
702 iso-contours of chemically active fluid material compared to EtOH, while the
703 further reduced reactivity of methane resulted in burning close to the stag-
704 nation plane and hence in regions characterised by high strain and vorticity.

705 The multi-fluid probabilities show that the aligned extent of the reaction
706 zone is typically less than two integral length scales of turbulence. Inside the
707 reaction zone, the probability of finding intermediate fluid states (i.e. beyond
708 bimodal statistics) was found to be strongly fuel and Da number dependent
709 with values exceeding 90% at low Damköhler numbers. A distinct impact of
710 the fuel reactivity and ease to ignition was also observed in the weakly re-
711 acting fluid velocities, characteristic of HCP supported burning, with an ad-
712 vanced reaction onset and stronger pronounced dilatation observed for DME.
713 The corresponding EtOH cases showed attenuated dilatation and a delayed
714 reaction onset. Both phenomena were much more distinct for CH_4 and can
715 partly be attributed to delayed auto-ignition facilitating higher HCP dilution
716 rates. A comparison at identical Lewis number for DME and EtOH, fuels
717 with similar laminar flame burning properties, further highlights the impor-
718 tance of the overall combustion chemistry in turbulent reacting flows. The
719 chemistry differences cause large changes in conditional velocity statistics

720 and the impact is hence not subtle. Consequently, the data are expected to
721 be particularly valuable for the development of models applicable to combus-
722 tion processes that transition to distributed modes under conditions where
723 the reaction progress can be controlled by an enthalpy source.

724 **Acknowledgements**

725 The authors would like to thank Dr Chiping Li, Dr Russ Cummings and
726 Dr Douglas Smith and acknowledge the support of the AFOSR and EOARD
727 under Grant FA9550-17-1-0021. The US Government is authorised to repro-
728 duce and distribute reprints for Governmental purpose notwithstanding any
729 copyright notation thereon. The authors would also like to thank Dr Robert
730 Barlow.

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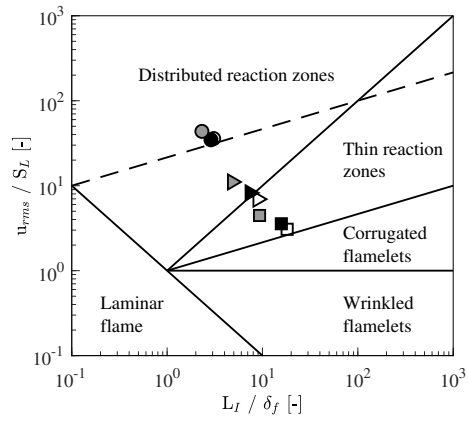


Figure 1: Combustion regime transitions of DME, EtOH and CH₄ visualised in a Borghi diagram for varying Φ . Empty symbols are the DME cases, filled EtOH and grey CH₄. \circ : $\Phi = 0.20$, \triangleright : $\Phi = 0.60$, \square : $\Phi = 1.0$.

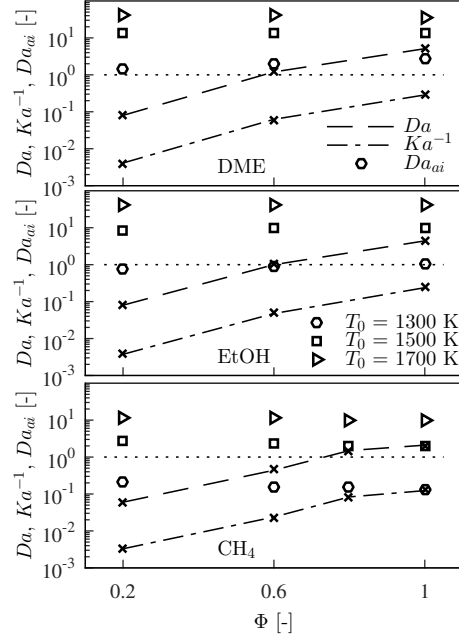


Figure 2: Estimated transition from conventional flame propagation (indicated by means of Da and Ka^{-1}) to an auto-ignition related burning mode (Da_{ai}) for DME (top), EtOH (middle) and CH_4 (bottom). The markers indicate the three Da_{ai} evaluated at initial temperatures of 1300, 1500 and 1700 K.

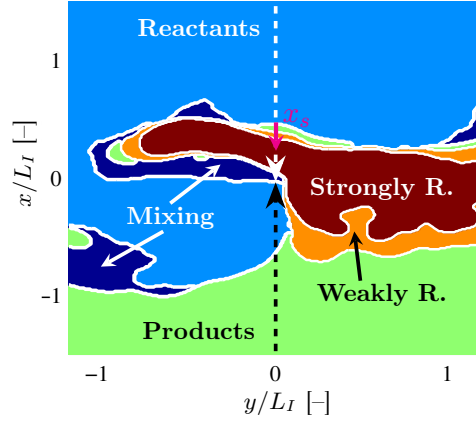


Figure 3: Instantaneous quinary multi-fluid field for DME / air at $\Phi = 0.60$ with $T_{HCP} = 1700$ K truncated around the stagnation point. Vertical white/black arrows show the theoretical stagnation point streamline (SPS). Interfaces are defined by the intersection of the SPS and material surfaces (white iso-contours). Reactants (light blue); Mixing (blue); Weakly reacting (orange); Strongly reacting (red); Products (green). The magenta arrow shows the x_s origin.

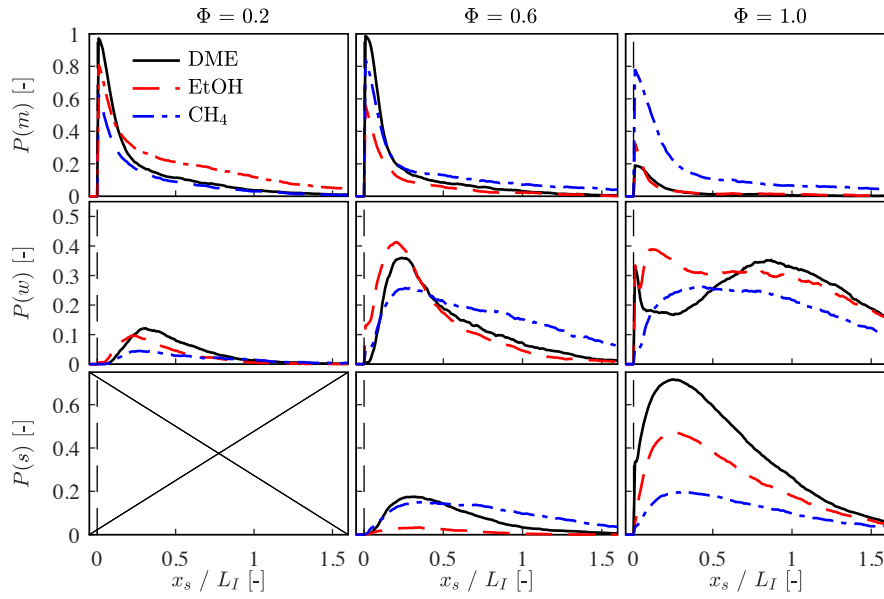


Figure 4: Multi-fluid probability (P) statistics for DME, EtOH and CH_4 combustion with varying Φ along the stagnation point streamline with data aligned at the $x_s = 0$ iso-contour. Top row: Mixing fluid probability ($P(m)$); Middle row: Weakly reacting fluid probability ($P(w)$); Bottom row: Strongly reacting fluid probability ($P(s)$); First column: $\Phi = 0.20$; Second column: $\Phi = 0.60$; Third column: $\Phi = 1.0$.

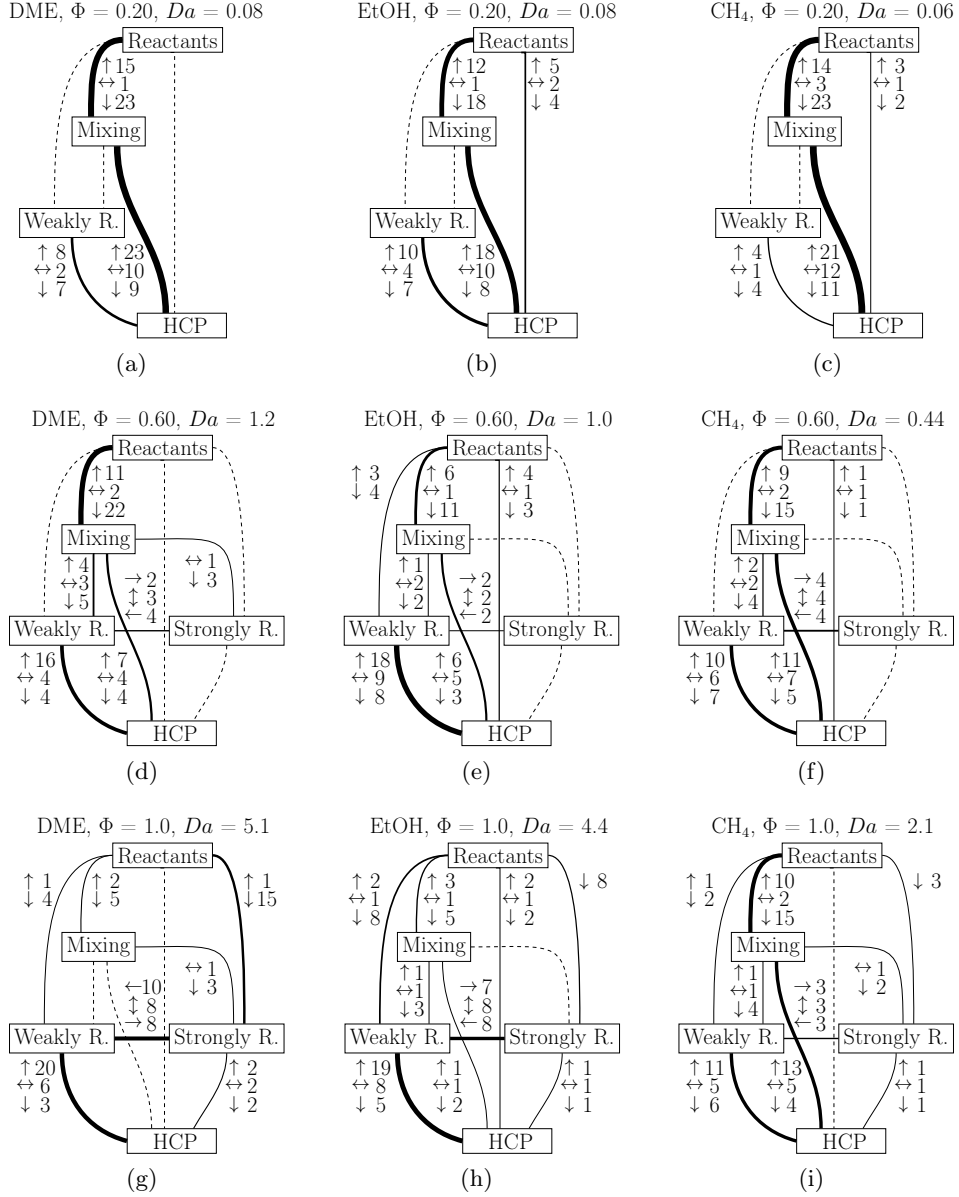


Figure 5: Fuel comparison by means of multi-fluid interface statistics in percent based on the total number of interfaces (N_i) where arrows indicate cases where (i) \uparrow the streamline tangent ($\hat{s} :=$ positive in flow direction) and interface normal ($\hat{n} :=$ positive from reactants to products) are aligned, i.e. $\hat{s} \cdot \hat{n} > 0.31$, (ii) \leftrightarrow nearly tangential flow, i.e. $\|\hat{s} \cdot \hat{n}\| < 0.31$ (i.e. $72 - 108^\circ$) and (iii) \downarrow the opposite of (i) with $\hat{s} \cdot \hat{n} < -0.31$ for all cases.

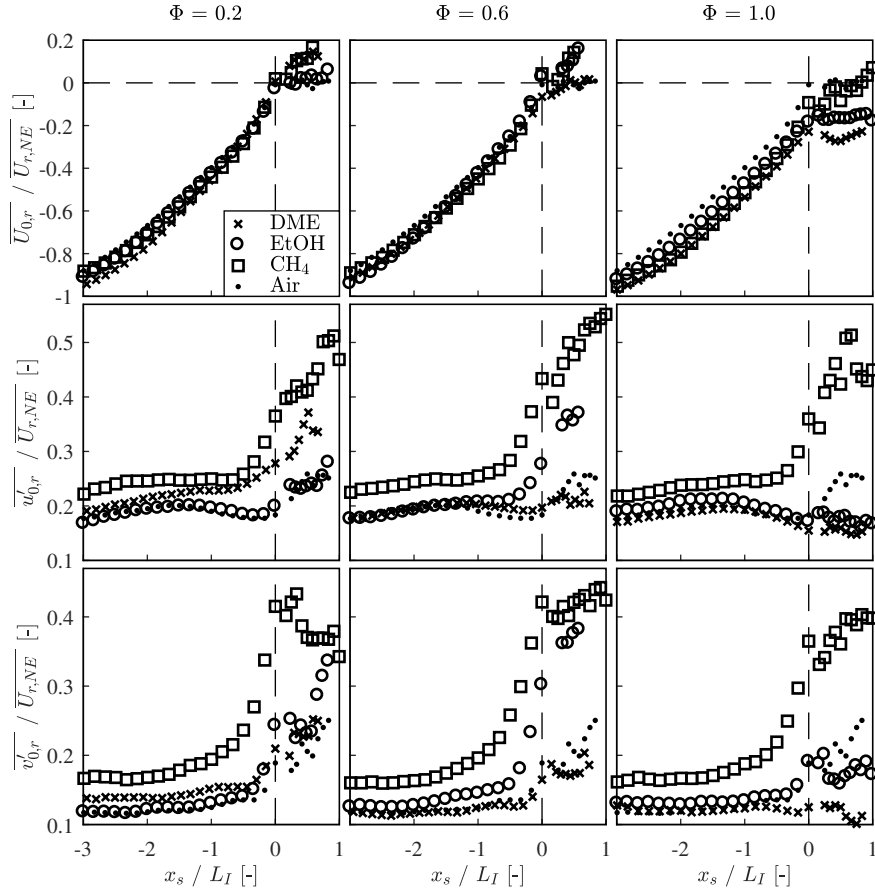


Figure 6: Conditional mean axial reactant fluid velocity and the axial and radial fluctuation for DME, EtOH and CH_4 at varying Φ evaluated along the stagnation point streamline and aligned at $x_s = 0$. The dotted line indicates the mixing case (i.e. air with $\Phi = 0.0$) for reference. Top row: $\overline{U_{0,r}} / \overline{U_{r,NE}}$; Middle row: $\overline{u'_{0,r}} / \overline{U_{r,NE}}$; Bottom row: $\overline{v'_{0,r}} / \overline{U_{r,NE}}$; First column: $\Phi = 0.20$; Second column: $\Phi = 0.60$; Third column: $\Phi = 1.0$.

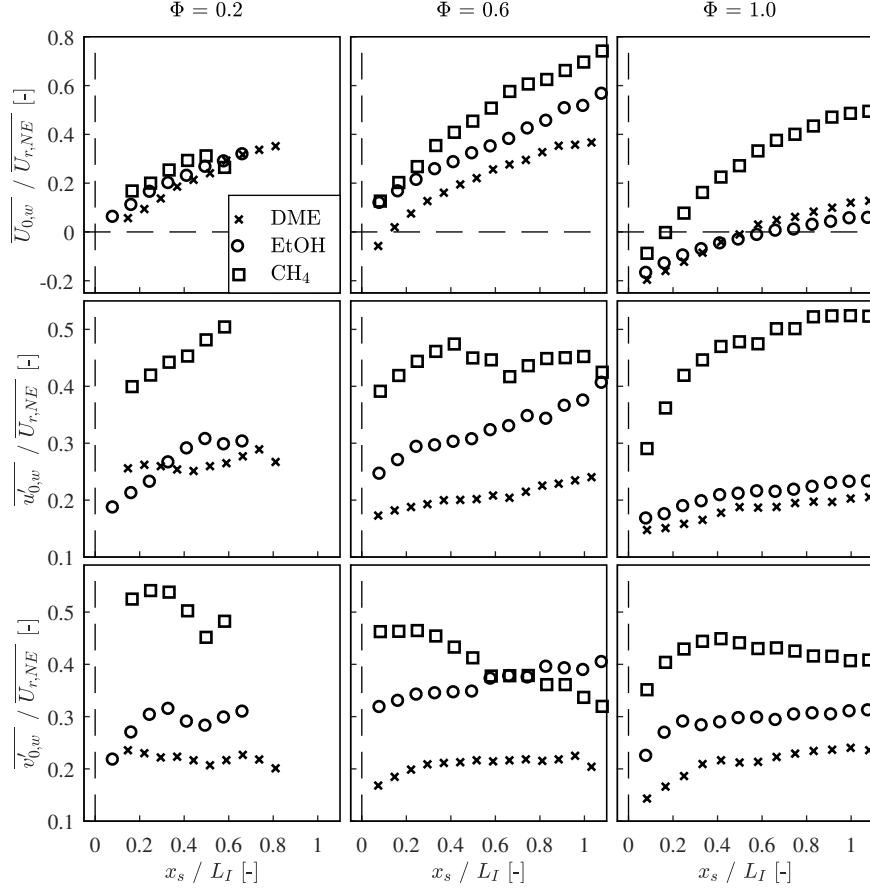


Figure 7: Conditional mean axial weakly reacting fluid velocity and the axial and radial fluctuations for DME, EtOH and CH₄ at varying Φ evaluated along the stagnation point streamline and aligned at $x_s = 0$. Top row: $\overline{U}_{0,w} / \overline{U}_{r,NE}$; Middle row: $\overline{u}'_{0,w} / \overline{U}_{r,NE}$; Bottom row: $\overline{v}'_{0,w} / \overline{U}_{r,NE}$; First column: $\Phi = 0.20$; Second column: $\Phi = 0.60$; Third column: $\Phi = 1.0$.

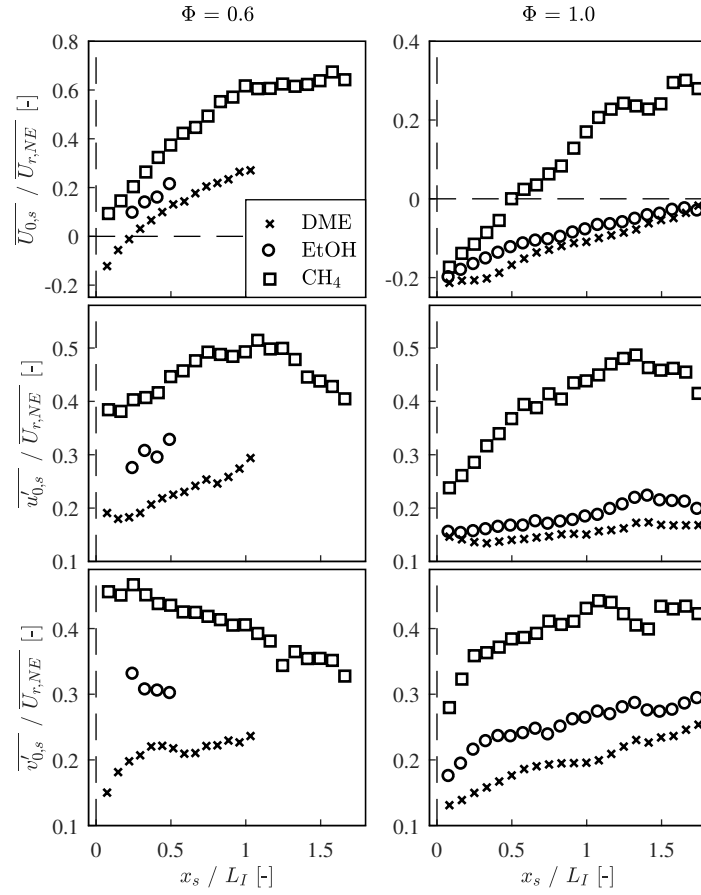


Figure 8: Conditional mean axial strongly reacting fluid velocity and the axial and radial fluctuations for DME, EtOH and CH_4 at varying Φ evaluated along the stagnation point streamline and aligned at $x_s = 0$. Top row: $\overline{U_{0,s}} / \overline{U_{r,NE}}$; Middle row: $\overline{u'_{0,s}} / \overline{U_{r,NE}}$; Bottom row: $\overline{v'_{0,s}} / \overline{U_{r,NE}}$; First column: $\Phi = 0.60$; Second column: $\Phi = 1.0$.

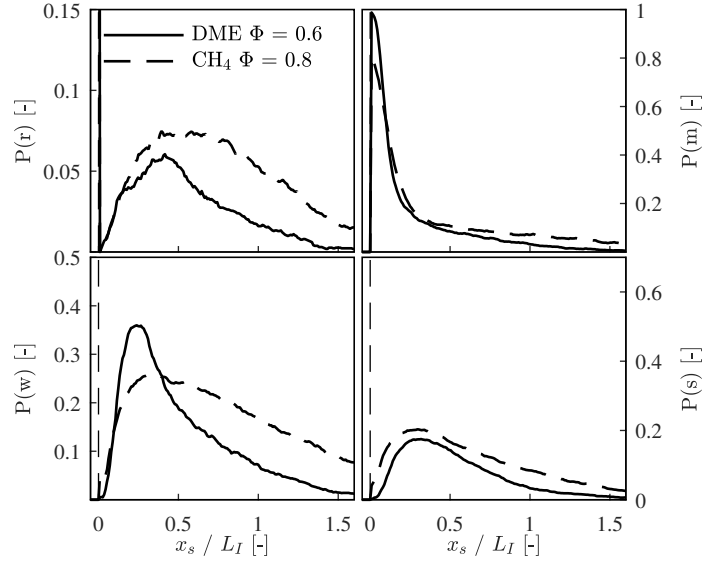


Figure 9: Conditional multi-fluid probabilities for cases with similar Damköhler numbers ($Da = 1.2$ vs. 1.5 ; DME with $\Phi = 0.60$ vs. CH_4 with $\Phi = 0.80$). Top left: Reactant fluid probability $P(r)$; Top right: Mixing fluid probability $P(m)$; Bottom left: Weakly reacting fluid probability $P(w)$; Bottom right: Strongly reactant fluid probability $P(s)$.

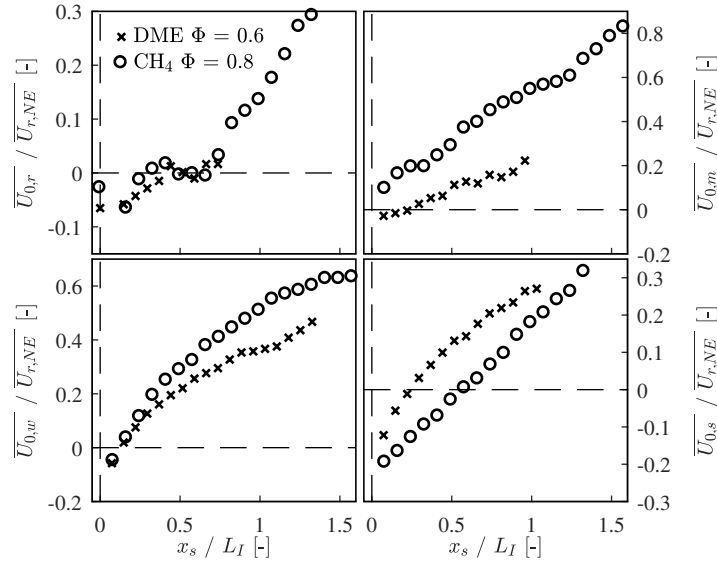


Figure 10: Conditional mean axial fluid velocities for cases with similar Damköhler numbers ($Da = 1.2$ vs. 1.5 ; DME with $\Phi = 0.60$ vs. CH_4 with $\Phi = 0.80$). Top left: Reactant fluid velocity; Top right: Mixing fluid velocity; Bottom left: Weakly reacting fluid velocity; Bottom right: Strongly reactant fluid velocity.

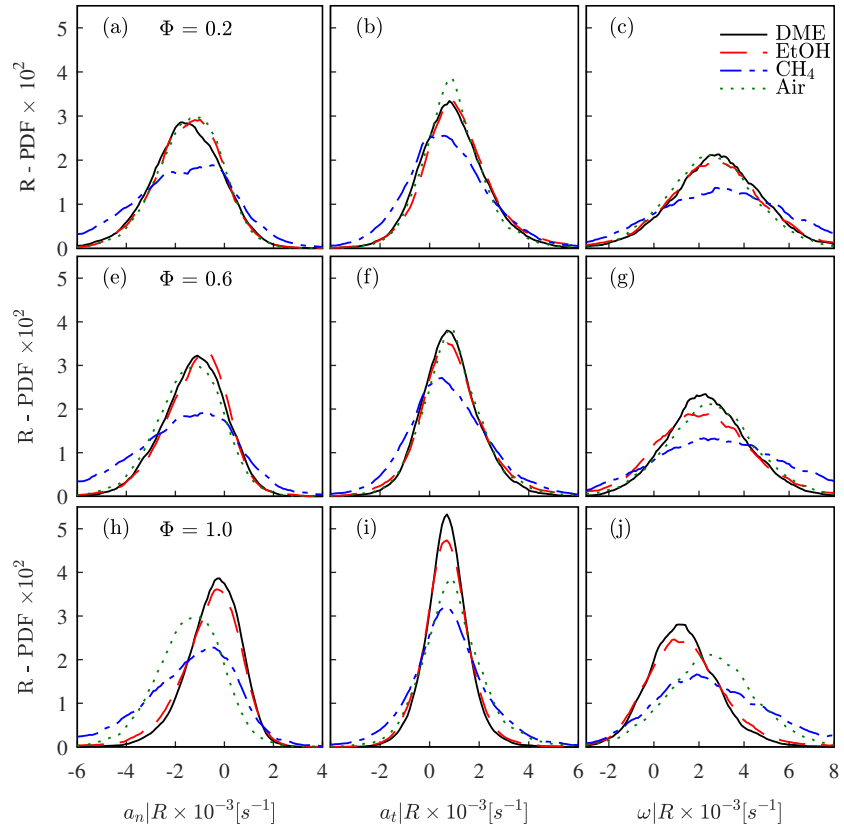


Figure 11: PDF of the rate of strain along the reactant fluid surface (R) for DME, EtOH and CH_4 at varying Φ . The dotted line indicates the mixing case (i.e. air with $\Phi = 0.0$) for reference. First column: Normal strain; Second column: Tangential strain; Third column: Vorticity. First row: $\Phi = 0.20$; Second row: $\Phi = 0.60$; Third row: $\Phi = 1.0$.

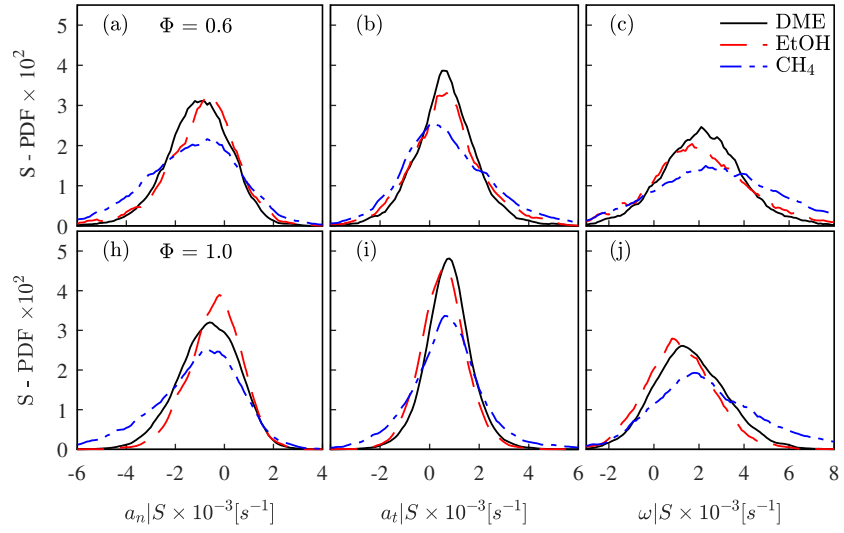


Figure 12: PDF of the rate of strain along the strongly reacting fluid surface (S) for DME, EtOH and CH₄ at varying Φ : First column: Normal strain; Second column: Tangential strain; Third column: Vorticity. First row: $\Phi = 0.60$; Second row: $\Phi = 1.0$.