

# The effects of asteroid layering on ejecta mass-velocity distribution and implications for impact momentum transfer

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## Abstract

Most bodies in the Solar System do not have a homogeneous structure. Understanding the outcome of an impact into regolith layers of different properties is especially important for NASA’s Double Asteroid Redirection Test (DART) and ESA’s Hera missions. Here we used the iSALE shock physics code to simulate the DART impact into three different target scenarios in the strength regime: a homogeneous porous half-space; layered targets with a porous weak layer overlying a stronger bedrock; and targets with exponentially decreasing porosity with depth. For each scenario we determined the sensitivity of crater morphology, ejecta mass-velocity distribution and momentum transferred from the impact for deflection,  $\beta - 1$ , to target properties and structure. We found that for the homogeneous porous half-space, cohesion and porosity play a significant role and the DART impact is expected to produce a  $\beta - 1$  between 1 and 3. In the two-layer target scenario, the

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presence of a less porous, stronger lower layer close to the surface can cause both amplification and reduction of ejected mass and momentum relative to the homogeneous upper-layer case. For the case of DART, the momentum enhancement can change by up to 90%. Impacts into targets with an exponentially decreasing porosity with depth only produced an enhancement in the ejected mass and momentum for sharp decreases in porosity that occur within 6 m of the asteroid surface. Together with measurements of the DART crater by the Hera mission, these results can be used to test the predictive capabilities of numerical models of asteroid deflection.

*Keywords:* Impact cratering, Ejecta, Numerical simulations, Layering, Kinetic impactor

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## 1. Introduction

Impact cratering is a fundamental processes involved in planetary formation. During the excavation stage of crater formation, a large amount of material is ejected ballistically out of the crater as ejecta (Oberbeck, 1975; Housen et al., 1983). Previous laboratory (Housen and Holsapple, 2011) and numerical studies (Jutzi and Michel, 2014; Luther et al., 2018; Raducan et al., 2019) of impact events into homogeneous targets have shown that the speed and mass of ejecta depends sensitively on target material properties, such as cohesive strength, porosity and the coefficient of internal friction. However, most bodies in the Solar System are not homogeneous, as shown by past missions to asteroids, such as the NEAR-Shoemaker (Veeverka et al., 2001), the OSIRIS-REx (Lauretta et al., 2019; Walsh et al., 2012) or the Hayabusa missions (Yano et al., 2006; Watanabe et al., 2019), as well as Earth-based thermal infrared observations (Delbo et al., 2014).

The surface material and the substructure varies from one asteroid to another. While the recent flybys to Ryugu and Bennu revealed a very rocky surface, a large sample of the visited asteroids have been observed to be covered by a layer of fine particles, termed regolith, which overlies a much stronger bedrock substrate (Gundlach and Blum, 2013). Examples of such asteroids include 21 Lutetia (Coradini et al., 2011), 243 Ida, 433 Eros, 951 Gaspra (Sullivan et al., 2002) or 25143 Itokawa (Miyamoto et al., 2007).

The effects of target layering on impact cratering were studied extensively through laboratory experiments conducted by Quaide and Oberbeck (1968). They investigated cratering in analog lunar regolith of different thicknesses, covering a denser, stronger substrate. The study concluded that for a certain

26 size-range of craters, the depth of the regolith has a large influence on the  
27 crater morphology and on the cratering efficiency. For the same impact pa-  
28 rameters, a deeper regolith resulted in larger craters ([Quaide and Oberbeck,](#)  
29 [1968](#)).

30 The presence of a substrate has also been shown to change the mass-  
31 velocity distribution and evolution of ejecta ([Schultz, 1992](#); [Senft and Stew-](#)  
32 [art, 2007](#)), which could have important implications for planetary defence,  
33 e.g. asteroid deflection by a kinetic impactor ([Ahrens and Harris, 1992](#);  
34 [Melosh et al., 1994](#); [Shafer et al., 1994](#)). The outcome of an impact into  
35 regolith layers of different properties and depths is especially important to  
36 understand in the context of space missions such as NASA’s Double Asteroid  
37 Redirection Test (DART) ([Cheng et al., 2016](#); [Cheng et al., 2018](#)) and ESA’s  
38 Hera ([Michel et al., 2018](#)).

39 In our previous work, [Raducan et al. \(2019\)](#), we presented numerical  
40 simulations of strength-dominated impacts on a homogeneous, small-asteroid  
41 surface. We quantified the influence of target cohesion, porosity and internal  
42 friction coefficient on the mass-velocity ejecta distribution produced by an  
43 example artificial impact and used the DART mission ([Michel et al., 2016](#);  
44 [Cheng et al., 2016](#)) as a motivating case study.

45 Here, we extend our previous work to include impacts into asteroid sur-  
46 faces with different internal structures. The paper begins with a brief sum-  
47 mary of the DART and Hera missions, a short description of asteroid de-  
48 flection by momentum transfer and of our current understanding of asteroid  
49 surfaces, followed by a summary of past research of impacts into homogeneous  
50 and layered targets. We then quantify the effects of the target structure on

51 crater morphology, the mass-velocity distribution of the ejected material and  
52 the amount of momentum transferred from the impactor to the asteroid for  
53 deflection. As a layered structure has been proposed for a number of asteroids  
54 observed by spacecraft, here we considered three possible impact scenarios:  
55 a) a homogeneous half-space target, with constant porosity, b) a layered tar-  
56 get with a dense interior, covered by a homogeneously porous regolith layer  
57 and c) a target with a porosity that decreases exponentially with depth.

### 58 *1.1. Kinetic impactor test*

59 To avoid a collision between an asteroid and Earth, the course of the  
60 asteroid could be changed by impacting its surface (Ahrens and Harris,  
61 1992; Melosh et al., 1994). The proposed Double Asteroid Redirection Test  
62 (DART), set to launch in 2021 (Cheng et al., 2018), aims to demonstrate a  
63 controlled deflection of a Near-Earth binary asteroid. The DART mission  
64 plans to impact the smaller component of the 65803 Didymos asteroid sys-  
65 tem, nicknamed Didymoon (Michel et al., 2016), and alter the binary orbit  
66 period of the system (Cheng et al., 2018).

67 In a high velocity impact event, the change in momentum of the asteroid,  
68  $\Delta p$ , can be amplified by the momentum of crater ejecta that exceeds the  
69 escape velocity, enhancing the asteroid deflection (Housen and Holsapple,  
70 2011)

$$\Delta p = \beta m v, \tag{1}$$

71 where  $m$  is the impactor mass,  $v$  is the impactor velocity and  $\beta$  is the momen-  
72 tum enhancement factor—a measure of the efficiency of momentum transfer.  
73  $(\beta - 1)m v$  represents the momentum of the crater ejecta. A value of  $\beta - 1 =$

74 0 implies that the crater ejecta makes no contribution to the total imparted  
75 momentum;  $\beta - 1 > 1$  indicates that the crater ejecta momentum over-  
76 all contribution is larger than the momentum imparted from the impactor  
77 itself.

78 The efficiency of momentum transfer has been found to vary significantly  
79 depending on the target asteroid’s properties and its composition (Holsapple  
80 and Housen, 2012; Jutzi and Michel, 2014; Stickle et al., 2015; Bruck Syal  
81 et al., 2016). Based on the results of experimental impacts in layered targets  
82 (Quaide and Oberbeck, 1968), it is likely that target layering will also affect  
83 the mass and velocity of the ejecta, and therefore,  $\beta$ .

84 To better understand the target properties of Didymoon, ESA is planning  
85 a rendezvous mission, Hera (Michel et al., 2018; Cheng et al., 2018), that has  
86 a 2023 planned launch and will arrive at Didymoon several years after the  
87 DART impact. The Hera spacecraft will provide detailed characterisation of  
88 the Didymoon volume and surface properties, as well as measure the outcome  
89 of the DART impact, including the new binary system orbit and the volume  
90 and morphology of the DART impact crater.

### 91 *1.2. Lunar regolith*

92 Regolith is defined as a layer of weak or unconsolidated material overlying  
93 a stronger substrate (Shoemaker et al., 1963). Most of the lunar regolith is  
94 of impact origin (McKay et al., 1991) and consists of particles smaller than  
95 1 cm in size; however, larger fragments or boulders can also be found on the  
96 surface. The regolith on the Moon was well characterised in many previous  
97 studies (McKay et al., 1991; Wilcox et al., 2005) thanks to the samples re-  
98 turned to Earth from the Apollo missions. The cohesion was measured to

99 be between 0.1 and a few kPa and the porosity was measured to be 50%  
100 (Mitchell, 1974; Colwell et al., 2007), dropping to  $\approx 40\%$  in just a few me-  
101 ters. The regolith covers the entire lunar surface and was estimated to be  
102 4-5 m in depth in the mare areas and 10-15 m in the older highland regions  
103 (McKay et al., 1991).

### 104 *1.3. Regolith on Didymoon*

105 The DART mission’s target, Didymoon, has a diameter of about 160 m  
106 (Michel et al., 2016) and a fast spinning primary close to the critical limit  
107 of stability (Walsh et al., 2012). Moreover, like many near-Earth asteroids,  
108 the double asteroid system is affected by the YORP effect (Jacobson and  
109 Scheeres, 2011).

110 Small asteroids of about 200 m and up to 10 km in size are likely to be  
111 rubble-pile objects (Michel et al., 2001; Richardson et al., 2009), aggregates  
112 held together only by self-gravity or cohesive forces (Richardson et al., 2002),  
113 while fast-spinning asteroids smaller than 150 m can be competent bodies  
114 (Sánchez and Scheeres, 2014). Regardless of the deep internal structure, the  
115 surface of many asteroids is expected to be covered by a layer of regolith  
116 (Sullivan et al., 2002), but its properties can differ from those of the lunar  
117 regolith, as well as from one asteroid to another. The thickness of the regolith  
118 layer and the particle size distribution can depend on the size of the asteroid  
119 or the internal structure (e.g. a porous asteroid would retain more regolith  
120 than a competent body of the same mass).

121 For example, the high escape velocity on asteroid 21 Lutetia,  $\approx 99$  km in  
122 diameter, has allowed a significant amount of the crater ejecta to be retained  
123 on the surface (Murdoch et al., 2015), creating an extensive regolith layer,

124 similar to the lunar regolith, of up to 600 m in depth (Vincent et al., 2012).  
125 Asteroid 433 Eros,  $\approx$  17 km in diameter, has a regolith with particle sizes  
126 ranging from dust grains to large boulders, which extend to depths of up  
127 to several tens of meters (Veveřka et al., 2001; Cheng et al., 2002). Most  
128 interesting of all is the 300 m asteroid NEA (25143) Itokawa. Itokawa is  
129 believed to be a rubble-pile asteroid (Fujiwara et al., 2006), and has two  
130 types of surface terrains: a boulder-rich terrain, with particles larger than 1  
131 cm, and a smooth terrain, with a 2.5 m layer of fine regolith particles covering  
132 a boulder rich substrate (Barnouin-Jha et al., 2008).

133 On the other hand, more recent space missions to top-shaped asteroids—  
134 e.g., JAXA’s Hayabusa2 to the 435 m Ryugu (Watanabe et al., 2019) and  
135 NASA’s OSIRIS-REx to the 268 m Bennu (Lauretta et al., 2019; Walsh et al.,  
136 2019)—have revealed boulder rich, almost regolith-free surfaces. However,  
137 until the arrive of DART at the Didymos system, the surface and internal  
138 structure of Didymoon will not be known. Depending on the formation  
139 mechanisms involved (Durda et al., 2004; Walsh et al., 2008), Didymoon  
140 could be a highly porous fractured body, a regolith-free rubble-pile (like  
141 Ryugu and Bennu); it could have a solid intact interior (e.g. a competent or  
142 monolithic body), or it could have a layered structure (like Lutetia, Eros or  
143 Itokawa) (Scheeres et al., 2010; Sánchez and Scheeres, 2014; Murdoch et al.,  
144 2017). In this paper we investigate impacts into a variety of simple internal  
145 structures, where the target properties vary only with depth, as possible  
146 near-surface structures for Didymoon.



147 *1.4. Impact studies on layered targets*

148 Several previous studies have investigated impacts into layered targets  
149 and the resulting crater formation. [Oberbeck and Quaide \(1967\)](#) studied  
150 fresh craters, with diameters less than 250 m, on the lunar maria and found  
151 four distinct crater morphologies: concentric, flat-bottomed, with a central  
152 mound and bowl shaped. They used laboratory impact experiments to de-  
153 termine the impact conditions responsible for each crater morphology occur-  
154 rence. In their experiments, they set up a simple two-layer target model,  
155 consisting of a loose quartz sand upper layer, overlying a substrate of denser  
156 cohesive quartz sand cemented with epoxy resin. The layered target was  
157 then impacted by various projectiles at velocities between 1 and 7 km/s.  
158 [Oberbeck and Quaide \(1967\)](#) were able to reproduce each of the observed  
159 crater morphologies by varying the thickness of the upper target layer. The  
160 pre-impact layer thickness that produced each crater type was shown to be  
161 influenced by the strength ratio of the target layers ([Quaide and Oberbeck,](#)  
162 [1968](#)), but did not seem to change with variations in the impact velocity, the  
163 angle of impact or projectile type.

164 [Senft and Stewart \(2007\)](#) performed numerical simulations of vertical im-  
165 pacts into basalt two-layer targets on the Moon, where the impact velocity  
166 was 17 km/s and the upper layer was 10 m thick. They varied the projec-  
167 tile radius to produce different crater sizes and replicated the [Quaide and](#)  
168 [Oberbeck \(1968\)](#) crater types. The layered targets had different strength pa-  
169 rameters configurations (e.g. cohesion, coefficient of internal friction), which  
170 were found to strongly influence the ejecta curtain.

171 [Prieur et al. \(2018\)](#) also conducted numerical studies of impact crater  
172 formation in two-layer targets under lunar conditions and investigated the  
173 conditions under which the crater morphologies defined by [Quaide and Ober-](#)  
174 [beck \(1968\)](#) were produced. In agreement with [Quaide and Oberbeck \(1968\)](#)  
175 and [Senft and Stewart \(2007\)](#), they found that the different types of crater  
176 morphology are the result of the variations between the properties in the  
177 two layers. However, they also found that the transitions between the dif-  
178 ferent crater morphologies depend on the mass and velocity of the impactor,  
179 as well as layer-strength variations, which was not observed by [Quaide and](#)  
180 [Oberbeck \(1968\)](#).

## 181 **2. Numerical Model**

182 In this work, we used the iSALE2D shock physics code ([Collins et al.,](#)  
183 [2004](#); [Wünnemann et al., 2006](#)) to numerically simulate impacts into aster-  
184 oid targets with different internal structures. iSALE2D is a multi-material,  
185 multi-rheology extension of the SALE hydrocode ([Amsden et al., 1980](#)), de-  
186 veloped to simulate high-speed impact processes and is similar to the older  
187 SALEB hydrocode ([Ivanov et al., 1997](#); [Ivanov and Artemieva, 2002](#)). To  
188 mimic the material response of an asteroid surface, we used strength models  
189 suitable for impacts into geologic targets ([Collins et al., 2004](#)) and a porosity  
190 compaction model, the  $\epsilon - \alpha$  model ([Wünnemann et al., 2006](#); [Collins et al.,](#)  
191 [2011](#)), both of which are included in iSALE2D.

192 Craters and ejecta distributions produced by impacts simulated with  
193 iSALE have been validated previously against laboratory impact experiments  
194 ([Wünnemann et al., 2016](#); [Luther et al., 2018](#); [Raducan et al., 2019](#)). iSALE

195 has also been benchmarked against other hydrocodes (Pierazzo et al., 2008),  
196 with crater size typically agreeing to within 10%.

### 197 *2.1. Impactor*

198 The impactor parameters were chosen to approximate the DART space-  
199 craft. We assumed a 310 kg porous sphere, with a diameter of 0.84 m,  
200 modelled using the Tillotson equation of state (EOS) and the Johnson-Cook  
201 strength model for aluminium (Johnson and Cook, 1983). The internal struc-  
202 ture of the spacecraft was heavily simplified and we used the  $\epsilon - \alpha$  porosity  
203 model (Wünnemann et al., 2006) to account for the voids in the spacecraft  
204 structure and achieve a bulk density of 1000 kg/m<sup>3</sup> (63% bulk porosity).  
205 The  $\epsilon - \alpha$  model took the nominal input parameters:  $\alpha_0 = 2.7$ ,  $\alpha_x = 1.0$ ,  $\kappa =$   
206  $0.9$ ,  $\chi = 1.0$ . To isolate the influence of target properties, we used the same  
207 impactor properties and an impact velocity of 7 km/s for all simulations.

### 208 *2.2. Target asteroid*

209 As discussed in the previous section, Didymoon’s surface properties or  
210 internal structure are not well understood. To study the target material  
211 response to a possible impact, we numerically simulated impacts into three  
212 distinct target scenarios: (a) a homogeneous porous half-space (Fig. 1a);  
213 (b) a layered target with a porous weak upper layer overlying a stronger  
214 bedrock layer (Fig. 1b); and (c) a target with a porosity that decreased  
215 exponentially with depth (Fig. 1c). For each of these target scenarios we  
216 systematically varied the target material properties (e.g. cohesive strength,  
217 initial porosity – including crush curve, layer thickness or porosity e-folding

218 depth) and determined the crater morphology, the amount of ejected mass  
 219 and the normalised momentum carried away by the ejecta,  $\beta - 1$ .

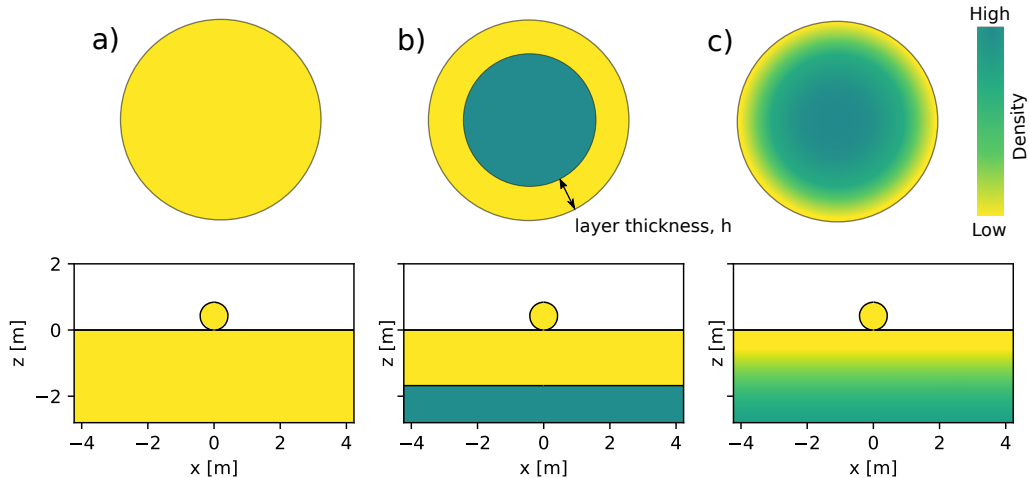


Figure 1: Schematic representation of the asteroid structure (upper row) and the equivalent numerical simulation set-up (lower row), for: a) homogeneous porous half-space; b) layered target consisting of a weak, porous upper layer covering a stronger, less porous substrate; c) target with a continuous porosity gradient.

220 As an S-type silicaceous asteroid system (Michel et al., 2016; Michel et al.,  
 221 2018), the target asteroid material was considered to be made of weak porous  
 222 basalt, which is a reasonable approximation of the compositional structure  
 223 of most asteroids. Therefore, for all impact scenarios, the target was mod-  
 224 elled using the Tillotson EOS for basalt (Tillotson, 1962; Benz and Asphaug,  
 225 1999). The surface gravity was kept constant at  $5 \times 10^{-3} \text{ cm/s}^2$ .

### 226 2.2.1. Homogeneous porous half-space

227 The first scenario investigated DART impacts into a homogeneous porous  
 228 half-space. The shear strength of the target was modelled using a simple

229 pressure-dependent strength model typical of rock-like materials, referred in  
230 this paper as LUND (Lundborg, 1967). The cohesive strength of the damaged  
231 target (at zero pressure),  $Y_{d0}$ , was varied between 0.1 and 100 kPa. The  
232 coefficient of internal friction,  $f$ , which can take a range of values on asteroid  
233 surfaces, was set to 0.6. This value was based on geological analogues and is  
234 the most common for geological materials (Goodman, 1989).

235 The initial porosity of the target,  $\phi_0$ , was varied between 0% and 50%,  
236 and was modelled using the  $\epsilon - \alpha$  model (Wünnemann et al., 2006). The input  
237 parameters for the  $\epsilon - \alpha$  model (Table 1) were derived from mesoscale mod-  
238 elling studies of dynamic compaction of chondritic material (Davison et al.,  
239 2016) and informed by comparison with measured crush curves (distension  
240 as a function of pressure) of analog materials (Fig. 2). We favoured the syn-  
241 thetic dynamic crush curves because they provide a systematic framework  
242 for varying all of the compaction model parameters over the range of target  
243 porosities considered here. Moreover, without crush-curve measurements of  
244 a direct sample of asteroid surface material it is difficult to determine the  
245 best analog for the compaction behaviour of asteroidal materials. Proposed  
246 asteroid regolith analogs, including gypsum (Nakamura et al., 2009), sand  
247 (Hagerty et al., 1993; Housen et al., 2018) and lunar regolith (Stephens and  
248 Lilley, 1970; Ahrens and Cole, 1974), show a range of compaction behavior.  
249 One way to characterise the compaction response is by a “crushing strength”  
250 defined as the pressure required to compact the material by one-half towards  
251 its solid density (Housen et al., 2018). For an initial porosity of 20% and 35%  
252 the crush curves used in this work are broadly consistent with the quasi-static  
253 crush curves of lunar microbreccia and lunar regolith (Fig. 2), and have sim-

254 ilar crushing strengths (Stephens and Lilley, 1970; Ahrens and Cole, 1974).  
255 The crushing strengths in our iSALE simulations were  $\approx 450$ - $650$  MPa for  
256 35% - 20% initial porosity, while the lunar regolith and the lunar microbreccia  
257 have crushing strengths of  $\approx 420$  MPa and 800 MPa, respectively. For  
258 an initial porosity of 35% and 50% the crush curves at low pressures are also  
259 broadly consistent with the crush curves of sand (Hagerty et al., 1993; Housen  
260 et al., 2018) and gypsum (Nakamura et al., 2009), respectively (Fig. 2). How-  
261 ever, both of these earth-sourced analogs have crushing strengths that are  
262 somewhat lower than those of the lunar analogs, perhaps because lunar re-  
263 golith grains are more angular with a higher internal friction angle than their  
264 terrestrial counter parts. Nevertheless, we note that in our simulations only  
265 a very small fraction of the (high velocity) ejecta experiences pressures above  
266 100 MPa, so we do not expect our results to be very sensitive to uncertainty  
267 in the high-pressure portion of the regolith crush curve. A small number of  
268 test simulations showed that adopting a lower crushing strength for a uniform  
269 regolith target changes the  $\beta - 1$  value by less than 4%.

270 The elastic volume strain threshold  $\epsilon_{e0}$  defines the volume strain that  
271 must be exceeded to induce permanent compaction. It can be converted to  
272 an approximate pressure at the onset of crushing by multiplying by the bulk  
273 modulus. As in previous work (Raducan et al., 2019),  $\epsilon_{e0}$  was set propor-  
274 tional to the ratio of the cohesion of the damaged material  $Y_{d0}$  and the bulk  
275 modulus. For simplicity and numerical stability, we set the  $\chi$  parameter (the  
276 elastic wave speed ratio of porous and non-porous material) to 1, implying  
277 that the elastic wave speed in the porous material was equal to that in the  
278 nonporous component. In reality, the elastic wave speed in a porous material

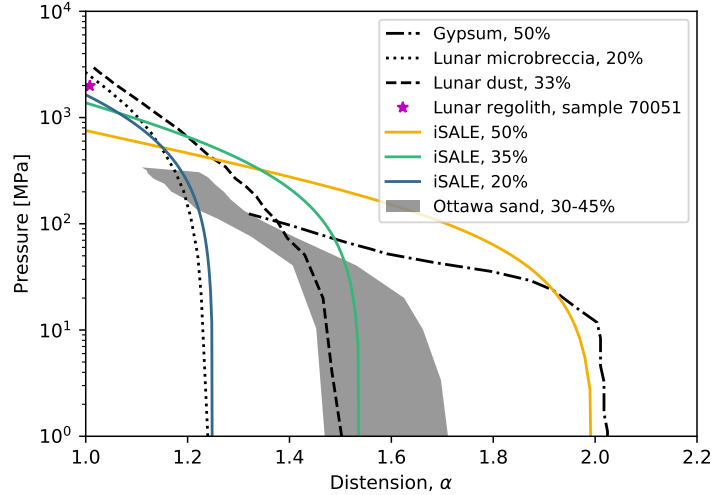


Figure 2: Crush curves used in iSALE for homogeneous targets with 20%, 35% and 50% porosity, compared with experimentally derived, quasi-static crush curves for 50% porous gypsum (Nakamura et al., 2009), 30-45% porous Ottawa sand (shaded area) (Hagerty et al., 1993; Housen et al., 2018), lunar dust and lunar microbreccia (Stephens and Lilley, 1970). Also shown is the result of one shock compaction experiment for lunar regolith (sample 70051) (Ahrens and Cole, 1974). To convert bulk density to distension,  $\alpha$  is estimated as the ratio of reference grain density to compressed bulk density,  $\rho_{s0}/\rho$ .

279 is usually only a small fraction of the speed in the solid component; how-  
 280 ever, sensitivity studies showed that for the scenarios simulated here, the  $\chi$   
 281 parameter has a negligible effect on crater formation, ejected mass or ejection  
 282 speeds. This is because inside the crater the stress wave is still in the  
 283 plastic/shock regime and its speed is dictated primarily by the crush curve  
 284 (and solid equation of state), rather than the elastic wave speed.

Table 1: Material model parameters for simulations of the DART impact. In addition, thermal parameters from [Ivanov et al. \(2010\)](#) were used.

Description	Symbol	Half-space	Regolith	Substrate
Equation of state		Tillotson <sup>a</sup>	Tillotson <sup>a</sup>	Tillotson <sup>a</sup>
Strength model		LUND	LUND	ROCK
Poisson ratio <sup>b</sup>	$\nu$	0.25	0.25	0.25
Intact strength at zero pressure (MPa)	$Y_{i0}$	–	–	1
Damage strength at zero pressure (kPa)	$Y_{d0}$	0.1–100	1	100
Strength at infinite pressure (MPa)	$Y_{inf}$	$10^3$	$10^3$	$10^3$
Coefficient of internal friction (intact material)	$f_i$	–	–	1.0
Coefficient of internal friction (damaged material)	$f$	0.6	0.6	0.6
Porosity model parameters ( $\epsilon - \alpha$ ) <sup>c</sup>				
Initial porosity	$\phi_0$	0–50%	35%/50%	0%/10%
Initial distension**	$\alpha_0$	1.0–2.0	1.54/2.0	1.0/1.11
Distension at transition to power-law	$\alpha_x$	1.0	1.0/1.0	–/1.0
Elastic volumetric strain threshold*	$\epsilon_{e0}$	$-2 \times 10^{-9}$ – $-2 \times 10^{-6}$	$-2 \times 10^{-8}$	–/ $-2 \times 10^{-6}$
Exponential compaction rate**	$\kappa$	0.80–0.98	0.94/0.98	–/0.80
Ratio of porous/nonporous elastic wave speed	$\chi$	1.0	1.00/1.00	–/1.00

<sup>a</sup>[Tillotson \(1962\)](#); <sup>b</sup>[Ivanov et al. \(2010\)](#); <sup>c</sup>[Wünnemann et al. \(2006\)](#).

\*  $\epsilon_{e0}$  varies proportionally with  $Y_{d0}$ .

\*\*  $\alpha_0 = 1/(1 - \phi_0)$  and  $\kappa$  vary with  $\phi_0$ .

### 285 2.2.2. Two-layer targets

286 The second impact scenario considered a simple two-layer target struc-  
287 ture, consisting of a weak, porous upper layer covering a stronger, less porous  
288 substrate. This target scenario is similar to what is observed on the Moon.  
289 To quantify the effects of target layering on the crater morphology and ejecta,  
290 we considered four sets of simulations, with different layer porosity configu-  
291 rations.

292 For all simulation sets, both the upper and the lower target materials  
293 were modelled using the Tillotson EOS for basalt. The upper layer had  
294 a cohesive strength of 1 kPa and was modelled using the LUND strength  
295 model ([Lundborg, 1967](#)), while the lower layer had a cohesive strength of  
296 100 kPa and was modelled using a more complex strength model, in which  
297 the strength is reduced as strain accumulates ([Collins et al., 2004](#)), referred



298 in this paper as the ROCK model. Senft and Stewart (2007) showed that  
299 at least a two-orders-of-magnitude difference between the strength in the  
300 upper layer and the substrate is required in order to produce different crater  
301 morphologies. Both layers had a coefficient of internal friction of the damaged  
302 material  $f = 0.6$ , while the coefficient of internal friction of the intact material  
303 (only in the ROCK model) was set to  $f_i = 1.0$ .

304 We performed four sets of simulations. For each set of simulations we  
305 varied the porosity in the upper and lower layers. In the first two sets, the  
306 upper layer (the regolith) had a nominal porosity of 35% and 50%, respec-  
307 tively, while the lower layer (the substrate) was non-porous (0% porosity).  
308 These two simulations will be referred to as R35-S0 and R50-S0 (Table 3).  
309 The next two sets considered less extreme variation in porosity between the  
310 regolith and substrate. In these two cases the regolith was 35% and 50%  
311 porous, respectively, while the substrate had a porosity of 10%. These sim-  
312 ulation sets were labelled R35-S10 and R50-S10. The full range of input  
313 parameters can be found in Table 1 and a summary of the simulation sets  
314 can be found in Table 3.

315 To investigate the influence of the regolith layer on the crater morphology  
316 and the ejecta produced, for each set of simulations we varied the regolith  
317 layer thickness,  $h$ , between 1 and 20 impactor radii.

Table 2: iSALE input parameters for the layered target.

Parameter	R35-S0	R50-S0	R35-S10	R50-S10
Regolith porosity, $\phi_{r0}$	35%	50%	35%	50%
Substrate porosity, $\phi_{s0}$	0%	0%	10%	10%

318 *2.2.3. Targets with exponentially decreasing porosity*

319 As noted by [Sullivan et al. \(2002\)](#), the transition between regolith and  
320 megaregolith on asteroids might not be as sharp as on the Moon, but instead  
321 might be more gradual. The third target scenario considered a target in  
322 which porosity decreased exponentially with depth. In all cases, strength  
323 was assumed to be independent of porosity and modelled using the LUND  
324 strength model with a fixed cohesive strength and coefficient of friction ( $Y_0$   
325 = 1 kPa and  $f = 0.6$ ). In iSALE, the distension,  $\alpha$ , is defined as the ratio of  
326 solid density to bulk density,  $\rho_s/\rho$ , which is equivalent to  $1/(1 - \phi)$ , where  $\phi$   
327 is the porosity ([Collins et al., 2011](#)). To achieve an exponential decrease in  
328 porosity with depth, the distension as a function of depth,  $\alpha(h)$ , is defined  
329 as

$$\alpha(h) = (\alpha_0 - \alpha_k) \times \exp(h/h_*) + \alpha_k \quad (2)$$

330 where  $\alpha_0$  is the distension at the surface (maximum distension),  $\alpha_k$  is the  
331 minimum distension (asymptotic limit),  $h$  is the depth and  $h_*$  is the e-folding  
332 depth. For this impact scenario, the  $\epsilon - \alpha$  parameters were kept constant  
333 for all initial distensions, including the critical volume strain for the onset of

334 plastic compaction,  $\epsilon_{e0}$ .

335 To investigate impacts into targets with exponentially decreasing porosity,  
 336 we considered two simulation sets. The first set had a surface porosity,  $\phi_0$ , of  
 337 50% and a minimum porosity,  $\phi_k$ , of 0%, while the second set had  $\phi_0 = 35\%$   
 338 and  $\phi_k = 10\%$ . For each set, we varied the e-folding depth,  $0.12 < 1/h_* <$   
 339 1.2.

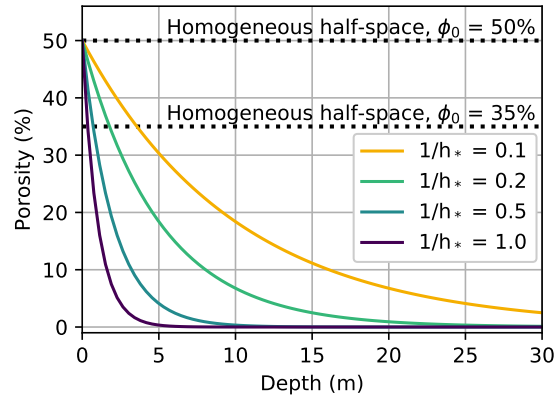


Figure 3: Target porosity as a function of depth for a range of values  $1/h_*$ , between 0.12 and 1.2. The surface porosity is 50% and the minimum porosity is 0%. The porosity in homogeneous half-space cases, at 35% and 50% porosity, are plotted for comparison.

Table 3: iSALE input parameters for the porosity gradient target.

Parameter	S50-K0	S35-K10
Surface porosity, $\phi_0$	50%	35%
Minimum porosity, $\phi_k$	0%	10%

340 *2.3. Regridding and resolution tests*

341 High spatial resolution is required to capture the particles ejected at very  
342 high velocities in the early stages of impact. At the same time, in all of our  
343 simulations the crater grows to become many times larger than the projectile  
344 and cratering occurs over a long time-scale, due to the low gravity and low  
345 strength of the targets investigated. To achieve high spatial resolution at  
346 early times, whilst limiting the computational expense of simulations, we used  
347 the iSALE’s regridding option. Regridding allowed the simulation domain to  
348 be coarsened by a factor of two after predetermined amounts of time. This  
349 method produced results that were comparable to those from simulations  
350 using 40 cells per projectile radius (cpr) resolution, but at an approximately  
351 equivalent cost of a 5 cpr run (Raducan et al., 2019).

352 Lagrangian tracer particles were placed across the computational domain  
353 and their mass and velocity were recorded if they crossed a fixed altitude,  
354 equal to one impactor diameter. Tracers that crossed this line were identified  
355 as ejecta in post-processing if their maximum speed exceeded both the es-  
356 cape velocity ( $\approx 6$  cm/s) and the velocity required to overcome the cohesive  
357 strength of the target. In all cases considered here, the velocity threshold  
358 was 10 cm/s. Due to the very small size of the simulated asteroid, the ejecta  
359 analysis presented here ignored the effect of gravity on the absolute value of  
360 the ejecta velocity at infinity (Housen and Holsapple, 2003). The gravity of  
361 Didymoon, albeit small, would affect the slower ejecta, which introduces an  
362 uncertainty in  $\beta$  of 2 to 4%.

363 The numerical simulations were run until all the ejecta was measured,  
364 however it is important to note that the end of the simulation was not neces-

365 sarily the end of the cratering process. Further crater collapse, which due to  
366 the very low surface gravity occurs over a much longer time-scale than crater  
367 excavation, was not simulated.

### 368 **3. Results of the DART impact**

369 To understand and quantify the effects of target structure on the crater  
370 morphology, amount of ejected mass and ejecta momentum, we divide our  
371 numerical studies into three different impact scenarios.

#### 372 *3.1. DART impact into a homogeneous porous half-space*

373 To investigate the individual effects of target cohesion and porosity on  
374 crater excavation and ejection processes, we first performed a series of simu-  
375 lations of impacts into homogeneous half-space targets. This extended results  
376 presented in [Raducan et al. \(2019\)](#), by providing additional numerical runs  
377 that covered a larger range of target porosities.

378 To investigate the effects of target cohesion we varied  $Y_{d0}$  between 0.1  
379 and 100 kPa, while keeping the porosity set at 20% and all the other impact  
380 parameters constant. We found that the target cohesion had a large influence  
381 on the crater diameter at the preimpact surface level, which varied from 36  
382 m for a 0.1 kPa target, to 9 m at 100 kPa.

383 To investigate the effects of porosity at constant cohesion, we varied the  
384 porosity between 0% and 50%, for cohesions of 1, 10 and 100 kPa. For all  
385 cohesions, the crater diameter decreased by about one third between the  
386 non-porous and 50% porosity target scenarios. However, the relative change  
387 in crater diameter between the 20% and the 50% porosity targets was less  
388 than 2 m (8-14%), which might be difficult to resolve from images taken

389 by a potential post-impact fly-by. This relatively modest change in crater  
390 diameter with porosity is consistent with previous experiments (Housen and  
391 Holsapple, 2003) and numerical simulations (Wünnemann et al., 2006; Prieur  
392 et al., 2017) of impacts in low-to-moderate porosities (up to 50%). A greater  
393 dependence of crater diameter and depth on crater size is expected for higher  
394 porosity targets not considered here (Housen and Holsapple, 2003).

395 By analysing the impact ejecta, we found that an increase in target poros-  
396 ity led to lower ejection velocities, while an increase in target cohesion re-  
397 sulted in lower ejection speeds only for the last ejecta to leave the crater.  
398 This is consistent with the results of recent numerical simulations of ejecta  
399 behaviour in gravity-dominated impacts (Luther et al., 2018).

400 By integrating the mass-velocity ejecta distribution, we determined the  
401 momentum transferred to the target and carried away by the ejecta,  $\beta - 1$   
402 1. Our simulation results suggest that  $\beta - 1$  is also very sensitive to the  
403 material properties investigated, with the target cohesion having the greatest  
404 influence. Figure 4 shows the the ratio of ejected momentum to impactor  
405 momentum  $\beta - 1$ , as a function of target porosity, for four different cohesions.  
406 As the cohesion is decreased,  $\beta - 1$  increases;  $\beta - 1$  also increases as the initial  
407 porosity of the asteroid surface decreases. For a target with 20% porosity,  
408 which is the current best estimate for the Didymos asteroid system (Michel  
409 et al., 2016),  $\beta - 1$  was determined to range between 1 for a cohesion of 100  
410 kPa and approximately 3 for a cohesion of 0.1 kPa.

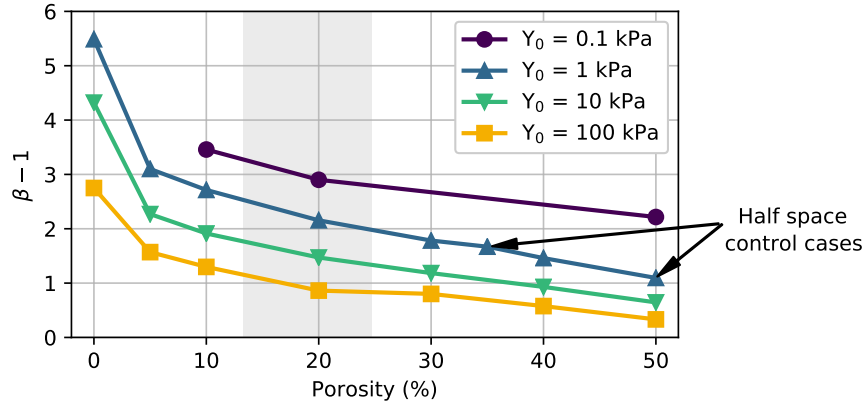


Figure 4: Total ejected momentum in the  $z$  direction ( $\beta - 1$ ) for four different cohesions and different porosities of the target. Adapted from Raducan et al. (2019).

411 Our results are broadly consistent with previous impact simulations where  
 412 targets with similar cohesive strengths were used. For example, simulations  
 413 by Bruck Syal et al. (2016) suggested a 10 km/s impact into a 1 kPa, 20%  
 414 porous target would produce a  $\beta - 1 \approx 3$ , while impacts into much stronger  
 415 targets, of a few MPa, have been shown to produce  $\beta - 1$  between 0.1 and  
 416 1 (Jutzi and Michel, 2014; Stickle et al., 2015; Cheng et al., 2016)). As  
 417 discussed in Raducan et al. (2019), these results emphasise the importance  
 418 of cohesion in determining the  $\beta$  value and the outcome of asteroid deflection.

419

### 420 3.2. DART impact into a two-layer target

421 We performed four sets of numerical simulations of impacts into layered  
 422 Didymoon-like targets, where each set had a different porosity configuration  
 423 in the upper and lower layers, as summarised in Table 3. For each simula-  
 424 tion set we varied the thickness of the regolith,  $h$ , from 0.5 to 20 times the

425 impactor radius,  $a$ .

### 426 3.2.1. Crater morphology

427 Varying the upper-layer thickness resulted in craters with different mor-  
428 phologies. We identified the same four types of crater morphologies as re-  
429 ported by [Quaide and Oberbeck \(1968\)](#): concentric craters, flat-floor craters,  
430 central-mound craters and bowl shaped craters. Examples of crater profiles  
431 of these morphologies, from our iSALE simulations, are illustrated in Fig. 5.  
432 The crater profiles are recorded at the end of the crater growth process ( $\approx$   
433 10 s); however, the material velocity in the rim area is non-zero and directed  
434 inwards at this time. This suggests that further collapse of rim material  
435 might occur, but over a long timescale due to the low gravity of the body.  
436 In Fig. 5, the material with a particle velocity pointing downwards and that  
437 might collapse is highlighted in red.

438 For very thin regolith layers,  $h/a < 4$ , cratering deformed both the up-  
439 per and the lower layers, to produce concentric craters with both an in-  
440 ner and outer crater rim. As the regolith layer was increased, in the range  
441  $4 < h/a < 10$ , the same impact conditions produced a crater that only formed  
442 into the upper layer and exhibited a flat floor. For regolith layer thicknesses  
443  $h/a$  between 10 and 16, the impact produced central-mound craters, while  
444 for very thick regolith layers,  $h/a > 16$ , the substrate had no influence on  
445 the cratering process. In this case, the crater took the form of a simple bowl  
446 shaped crater, similar to the craters formed in the equivalent homogeneous  
447 target simulation.

448 In our study, the transition threshold between these crater morphologies  
449 did not appear to be influenced by the range of porosities investigated. How-



450 ever, [Prieur et al. \(2018\)](#) noted the the transition threshold between crater  
 451 morphologies, for a given impact into a layered target, depended on the im-  
 452 pactor’s mass and velocity.

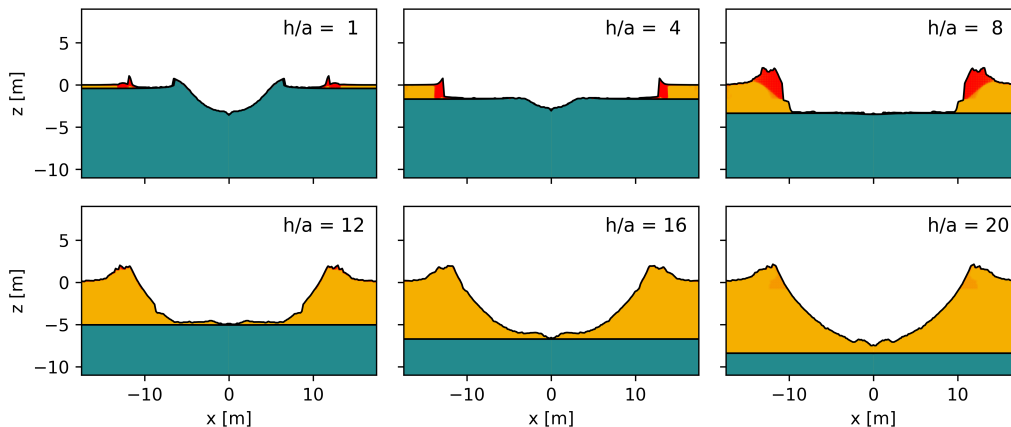


Figure 5: Example of crater profiles with distinct morphologies: concentric crater ( $h/a = 1$ ), flat-bottomed ( $h/a = 8$ ), central-mound ( $h/a = 12$ ) and bowl shaped ( $h/a = 20$ ). The upper layer had a pre-impact porosity of 35%, while the substrate was non-porous. Crater rim material highlighted in red is expected to undergo late-stage collapse.

### 453 3.2.2. Ejected mass

454 We have shown in the previous section that DART impacts into two-layer  
 455 targets produce different crater morphologies, each of which had a different  
 456 cratering efficiency. For impacts into homogeneous targets, the crater size  
 457 is strongly correlated with the amount of ejecta produced, and a similar  
 458 behaviour is expected for impacts into layered targets.

459 Fig. 6 shows the total ejected mass,  $M$ , as a function of the upper layer  
 460 thickness, normalised by the impactor radius,  $h/a$ . The ejected mass for each  
 461 simulation is normalised by the amount of ejected mass from an equivalent

462 impact into a half-space regolith target, with no lower layer,  $M_{hs}$ . R35-S0  
463 and R35-S10 results are normalised by the total mass ejected from an impact  
464 into a 35% porosity homogeneous half-space target, while R50-S0 and R50-  
465 S10 results are normalised by the mass ejected from an impact into 50%  
466 porosity half-space target.

467 For impacts into regolith layers with a thickness greater than  $\approx 18 h/a$ ,  
468 the total ejected mass converged to the mass ejected in the equivalent homo-  
469 geneous regolith scenario. As the crater only formed in the upper layer, the  
470 influence of the substrate on the excavation flow was not significant. How-  
471 ever, as the substrate got closer to the surface, up to 20% more mass was  
472 ejected than in the porous homogeneous regolith cases. This amplification in  
473 ejected mass is a consequence of the difference in the mechanical impedance  
474 (the product of density and wave speed) between the upper and lower layers.  
475 The impedance contrast results in a strong reflection of the shock wave at the  
476 boundary, and only partial transmission of energy into the substrate. As a  
477 result, less of the impactor kinetic energy was transmitted into the substrate  
478 and more was retained within the shallow subsurface, which amplified the  
479 ejection speeds in the upper layer.

480 On the other hand, for thin regolith layers ( $h/a$  less than 4; concentric  
481 craters) the ejected mass is less than in the homogeneous upper-layer case,  
482 despite the greater density of the material in the lower layer. At these regolith  
483 thicknesses, the excavation flow is strongly impeded at depth by the higher  
484 strength in the lower layer, but less impeded in the weaker regolith. The  
485 large strength difference between the layers more than compensates for the  
486 higher density of the substrate material and results in a reduction in total

487 ejected mass relative to the homogeneous upper-layer case.

488 For intermediate regolith layer thicknesses of about  $4 < h/a < 10$ , the  
 489 total ejected mass can be higher or lower than in the homogeneous upper-  
 490 layer case, depending on the size of the porosity contrast between the layers.  
 491 In this regime, both effects— increased resistance to flow in the substrate,  
 492 and reflection of kinetic energy back into the regolith—are in competition.  
 493 For the scenarios with a nonporous substrate, maximum ejecta mass ampli-  
 494 fication (20-30%) occurs at a regolith thickness of  $h/a \approx 6$ . A small amount  
 495 (10%) of porosity in the substrate material is sufficient to negate most of the  
 496 amplification of ejection from shock wave reflection.

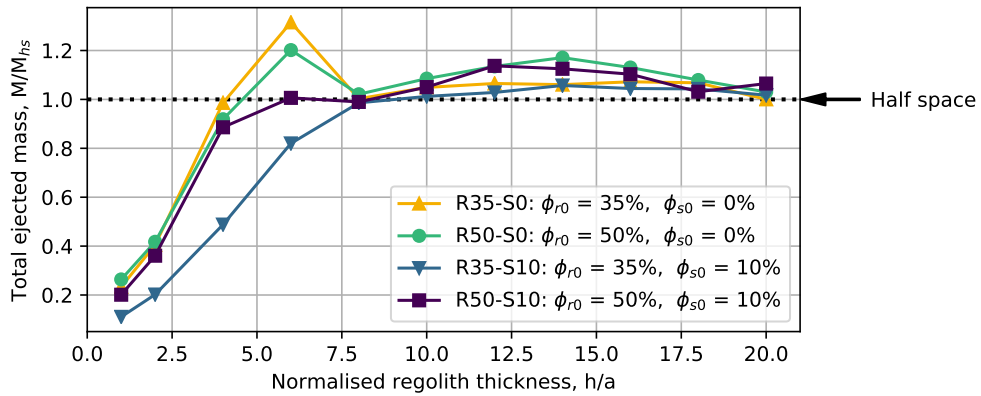


Figure 6: Total ejected mass,  $M$ , normalised by the mass ejected in the half-space case,  $M_{hs}$ , for impacts into layered targets with different regolith-substrate porosities configurations: R35-333S0, R50-S0, R35-S10 and R50-S1 (see Table 3 for details). The total mass is plotted as a function of regolith thickness  $h$  normalised by the impactor radius,  $a$ .

### 497 3.2.3. Ejected momentum

498 Fig. 7a shows the cumulative mass ejected at speeds higher than velocity  
 499  $v$ ,  $M(> v)$ , normalised by the mass of the impactor,  $m$ , as a function of ejection

500 tion velocity,  $v$ , normalised by the impactor velocity,  $U$ . The cases illustrated  
501 are for a 35% porosity upper layer and a nonporous substrate (R35-S0). The  
502 ejecta mass-velocity distributions were integrated to determine the cumula-  
503 tive, vertically ejected momentum,  $p_{ej(z)}/mU = (\beta - 1)$ , as a function of  
504 ejection speed in the vertical direction  $v_z$  (Fig. 7b).

505 An impact into a 35% porous homogeneous regolith target produced a  
506 normalised total ejecta momentum,  $\beta - 1$ , of about 1.75 (y-intercept of the  
507 dotted line in Fig. 7b). The effects of layer thickness on ejected momentum  
508 are more complex than on the ejected mass, as layering influences both the  
509 velocity and the mass of ejecta. For a regolith layer thickness  $h/a = 14$ ,  
510  $\beta - 1$  is  $\approx 10\%$  greater than in the regolith half-space case. At this regolith  
511 thickness, the top of the substrate is sufficiently close to the surface for the  
512 reflected shock wave to significantly increase ejection speeds towards the end  
513 of crater growth. This results in the amplification of ejection speeds and  
514 momentum of the slowest ejecta that is last to leave the crater.

515 For regolith thicknesses  $4 < h/a < 10$ , where the impact formed flat-  
516 floored craters, the shock wave reflects off the regolith-substrate boundary  
517 earlier in the cratering process, because the boundary is closer to the sur-  
518 face. Earlier shock wave reflection likewise results in amplification of ejection  
519 speeds earlier in the cratering process, such that even the fastest ejecta are  
520 given an extra acceleration from the reflected shock wave. For  $h/a \approx 6$ ,  $\beta - 1$   
521 is increased to over 2.

522 For thin regolith layers,  $h/a < 6$ , a number of competing effects are  
523 observed. On the one hand, as regolith thickness is decreased the reflected  
524 shock wave occurs earlier and earlier, tending to produce an even larger

525 amplification of ejection speeds. On the other hand, the high strength of  
 526 the substrate also acts to retard the excavation flow at depth, which reduces  
 527 ejection speeds. Finally, as the regolith layer gets very thin,  $h/a < 1$ , dense,  
 528 lower layer material becomes part of the ejecta mass. Both the mass and  
 529 the speed of the fastest ejecta, relative to the porous half-space case, were  
 530 increased by the high density and the low shock wave attenuation of the  
 531 lower layer. The highest  $\beta - 1$  value was observed in the thinnest regolith  
 532 layer case simulated, with  $h/a = 0.5$ .

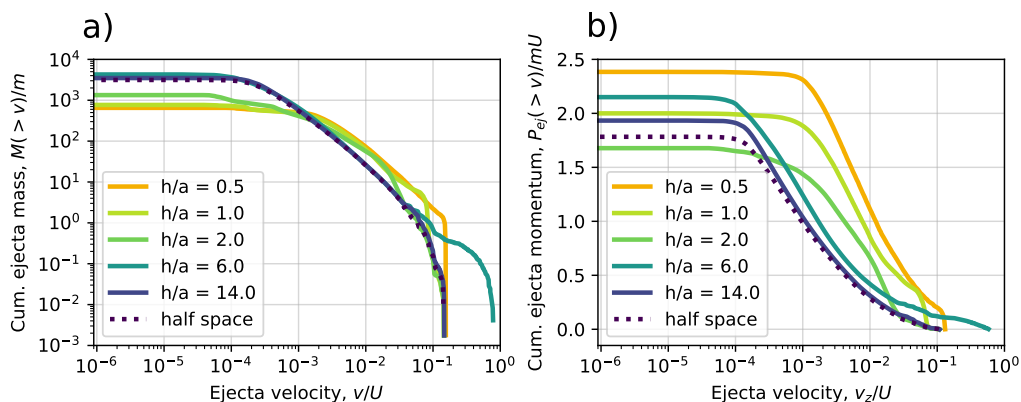


Figure 7: Ejecta distribution from impacts into layered target with a 35% porosity regolith of varying thicknesses ( $0.5 < h/a < 14$ ), overlying a non-porous substrate. a) Cumulative ejected mass,  $M(>v)$ , normalised by the mass of the impactor,  $m$ , as a function of normalised ejection velocity,  $v/U$ . b) Total ejected momentum,  $\beta - 1$ , as a function of the normalised vertical component of the velocity,  $v_z/U$ .

533 Fig. 8 shows the momentum carried away by the ejecta,  $\beta - 1$ , as a function  
 534 of regolith thickness,  $h/a$ , for the four layering scenarios investigated here:  
 535 R35-S0, R50-S0, R35-S10 and R50-S10. For thick regolith layers, the trend  
 536 in all cases is well behaved, while for thinner regolith layers,  $h/a < 10$ , the

537 behaviour is more complex. In this range,  $\beta - 1$  is very sensitive to the  
 538 material properties of the two layers. The figure also shows that when the  
 539 substrate has a small porosity, any amplification in ejected momentum is  
 540 reduced or even negated for thin-layer scenarios.

541 Compared with the uniform half-space cases, we found that the optimum  
 542 layer thickness that produces the maximum ejected momentum depends on  
 543 the strength and density ratios of the two layers, and it can be different to  
 544 the layer thickness that results in the greatest total ejected mass.

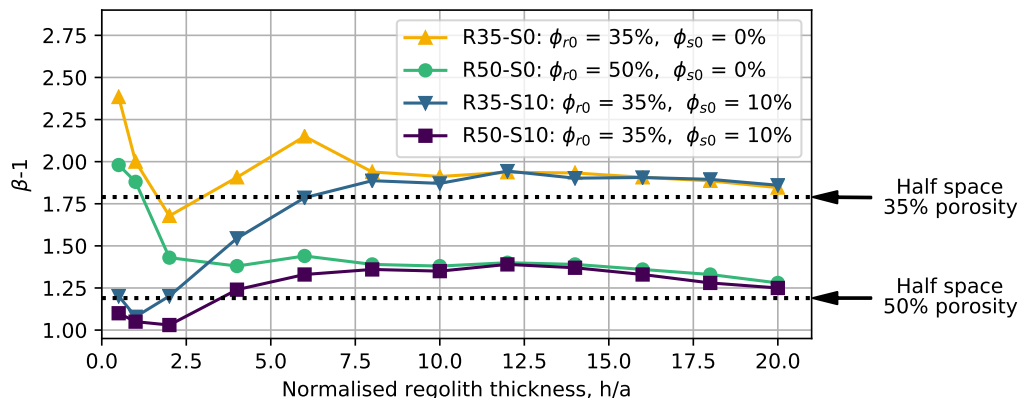


Figure 8: Normalised ejected momentum,  $\beta-1$ , from impact simulations into four different configurations of layered targets, R35-S0, R50-S0, R35-S0 and R50-S10 (Table 3).  $\beta-1$  is plotted as a function of target regolith thickness,  $h$ , normalised by the impactor radius,  $a$ .  $\beta-1$  from impacts into 35% and 50% porous half-space are plotted for comparison (dotted horizontal lines).

545 In the R35-S0 impact scenario, the most mass was ejected at  $h/a = 6$ ,  
 546 and the largest  $\beta - 1$  amplification was at  $h/a < 0.5$ , with a 30% amplifi-  
 547 cation, and at  $h/a = 6$ , with a 20% amplification. On the other hand, in  
 548 the R50-S0 impact scenario, the most mass was ejected at  $h/a = 6$ , while

549  $\beta - 1$  was only 10% larger than in the regolith half-space case. This suggests  
 550 that the additional mass is ejected at very low velocities and it does not  
 551 add significantly to the total momentum of the ejecta. The greatest ejected  
 552 momentum occurred at  $h/a = 0.5$ , even though the ejected mass in this case  
 553 was 70% lower than the homogeneous regolith case.

554 *3.3. DART impact into targets with an exponentially decreasing porosity with*  
 555 *depth*

556 The last simulated scenario considered DART impacts into asteroid tar-  
 557 gets with a porosity that decreased exponentially with depth. We numerically  
 558 simulated impacts into two target configurations, one with a surface porosity  
 559 of  $\phi_0 = 50\%$  and a minimum porosity of  $\phi_k = 0\%$  (S50-K0), and one with  
 560 surface porosity of  $\phi_0 = 35\%$  and a minimum porosity of  $\phi_k = 10\%$  (S35-  
 561 K10). For each target configuration we varied the porosity e-folding depth,  
 562  $0.1 < 1/h_* < 1.0$ .

563 All the impacts simulated in this case formed simple bowl-shaped craters,  
 564 with radii that remained almost constant with varying e-folding depth,  $h_*$ .  
 565 The crater radius was  $\approx 10$  m in the S50-K0 case and  $\approx 10.5$  m in the S35-K10  
 566 case, which is comparable to the radius of craters formed in the equivalent  
 567 half-space targets, with 50% and 35% porosity, respectively.

568 Fig. 9a shows the total ejected mass for the two simulated scenarios, S50-  
 569 K0 and S35-K10, as a function of normalised porosity e-folding depth,  $h_*/a$ .  
 570 A smaller e-folding depth implies a more rapid decrease in porosity with  
 571 depth below the surface (Fig. 9). In both cases, the ejected mass is normalised  
 572 by the total ejected mass in their corresponding minimum porosity half-space  
 573 scenario. Fig. 9b shows the total momentum carried away by the ejecta in the

574  $z$ -direction, normalised by the impactor momentum,  $\beta - 1$ , as a function of  
 575 normalised porosity e-folding depth,  $h_*/a$ .

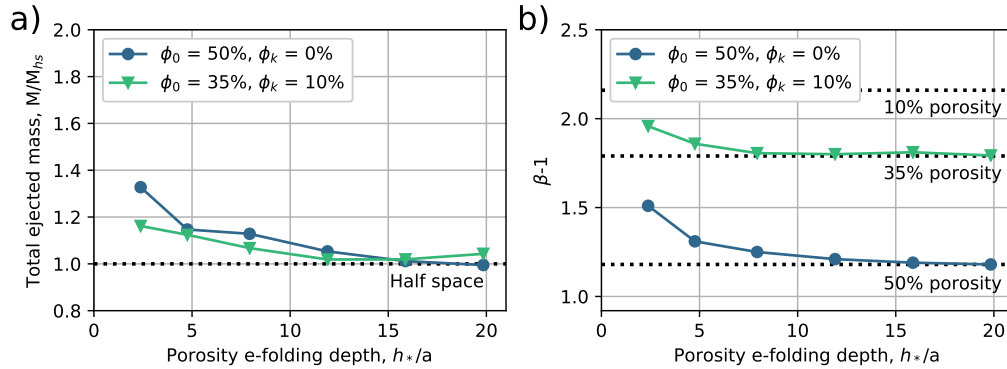


Figure 9: Numerical simulations results for impacts into targets with a continuous porosity gradient. a) Total ejected mass, as a function of normalised porosity e-folding depth,  $h_*/a$ . The mass ejected from the  $\phi_0=50\%$ ,  $\phi_k = 0\%$  case was normalised by the 50% half-space scenario and the mass from  $\phi_0=35\%$ ,  $\phi_k = 10\%$  was normalised by the 35% half-space scenario. b) Normalised ejected momentum,  $\beta - 1$  as a function of normalised porosity e-folding depth. The dotted horizontal lines represent the  $\beta - 1$  in the 10%, 35% and 50% porosity half-space.

576 Both the ejected mass and the ejected momentum graphs follow the same  
 577 trend. For large porosity e-folding depths,  $h_*/a > 12$ , there was no significant  
 578 increase in the ejected mass or the ejected momentum, compared to the  
 579 equivalent half-space scenario. On the other hand, as porosity e-folding depth  
 580 decreased,  $h_*/a \ll 12$ , more mass was ejected as the density of the ejected  
 581 subsurface material increased. Similarly, the denser ejected mass also added  
 582 to the amount of ejected momentum,  $\beta - 1$ , causing an amplification of up  
 583 to 25%.



584 **4. Discussion**

585 *4.1. Comparing ejecta from impacts into layered targets with impacts into*  
586 *targets with a porosity gradient*

587 To allow for a meaningful comparison between two impacts, where one  
588 is in a layered target and one is in a homogeneous target with exponentially  
589 increasing density, we compared the ejecta mass and momentum transfer  
590 from impacts into targets with equivalent mass per unit area. We integrated  
591 the porosity-target depth function for each simulated target and found that  
592 a layered target of thickness  $h/a$  has the same mass per unit area as a target  
593 with a porosity e-folding gradient of  $h_*/a$ . For example, a target with a  
594 surface porosity  $\phi_0 = 50\%$ , minimum porosity  $\phi_k = 0\%$  and an e-folding  
595 depth of  $h_*/a = 20$  had the same integrated area as a target with a  $20a$  thick  
596 50% porous regolith layer overlying a non-porous substrate (Fig. 10).

597 Fig. 11a shows the total ejected mass from the R50-S0 layering scenario  
598 (50% porosity upper layer over a non-porous substrate), compared with the  
599 S50-K0 porosity gradient scenario (50% surface porosity and 0% minimum  
600 porosity). The ejected mass in the layered targets is plotted as a function of  
601 regolith layer thickness,  $h/a$ , while the ejected mass from the exponentially-  
602 decreasing porosity targets is plotted as a function of porosity e-folding depth,  
603  $h_*/a$ . Similarly, Fig. 11c shows the normalised ejected momentum,  $\beta - 1$   
604 for the R50-S0 layering scenario, compared with the S50-K0 exponentially-  
605 decreasing porosity scenario.

606 Impacts into both the layered target and the exponentially-decreasing  
607 porosity target scenarios display an amplification in the amount of ejected  
608 momentum,  $\beta - 1$ , compared to the homogeneous 50% half-space case. In

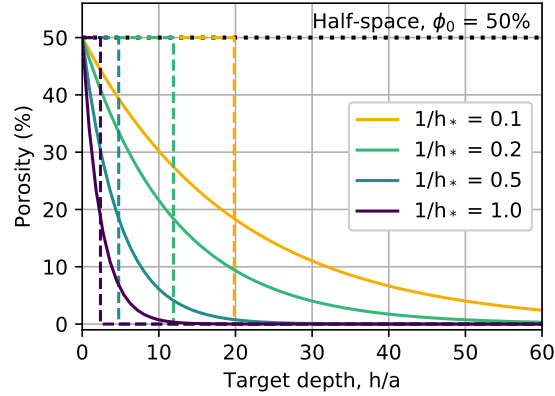


Figure 10: Target porosity as a function of depth for a range of e-folding depths,  $0.1 < 1/h_* < 1.0$ . The surface porosity is 50% and the minimum porosity is 0%. The equivalent integrated area from a layered target porosity-depth function is shown in same colour and dashed lines. The homogeneous half-space case at 50% porosity is plotted for comparison (dotted line).

609 both cases,  $\beta - 1$  has an ascending trend with decreasing regolith thickness,  
 610 and with values of up to 60% larger than in the half-space case (Fig. 11c).  
 611 However, the mechanism responsible for the amplification differs between  
 612 the two scenarios. For the exponentially-decreasing porosity scenarios, the  
 613 amplification in  $\beta - 1$  is caused by the ejection of denser subsurface material.  
 614 In the layered scenario, the amount of ejected mass is limited by the high  
 615 strength of the substrate, which impedes the excavation flow (Fig. 11a). The  
 616 amplification in  $\beta - 1$  is instead caused by the ejecta acceleration due to the  
 617 reflection of the shock wave.

618 Fig. 11b and d show the normalised total ejected mass and the normalised  
 619 ejected momentum from the R35-S10 layering scenario (35% porosity upper  
 620 layer over a 10% porous substrate), compared with the S35-K10 porosity

621 gradient scenario (35% surface porosity and 10% minimum porosity).

622 For  $h/a < 10$ , impacts into the S35-K10 target scenario produced similar  
623 trends for  $\beta - 1$  as in the S50-K0 target scenario, with an increased amplifica-  
624 tion in the amount of ejected momentum towards sharper porosity gradients,  
625 compared to an impact into a 35% porous half-space. At the same time, for  
626 the same  $h/a$ , impacts into R35-S10 produced smaller  $\beta - 1$ . This is caused  
627 by the difference in strength: in the layered case, the lower layer is denser,  
628 but also stronger; in the porosity gradient case, the deeper material is denser,  
629 but not stronger.

630 The work presented here is just a first step in understanding the effects of  
631 layering on small-body crater formation. Future work is needed to examine  
632 the effect of a continuous increase in strength with depth commensurate with  
633 the decrease in porosity. Impact angle is also expected to play a significant  
634 role in the ejecta distribution and momentum transfer. While numerical  
635 simulations presented were restricted to vertical impact only, future studies  
636 should also investigate and quantify the role of the impact angle on the ejecta  
637 distribution of impacts into homogeneous and layered asteroid targets.

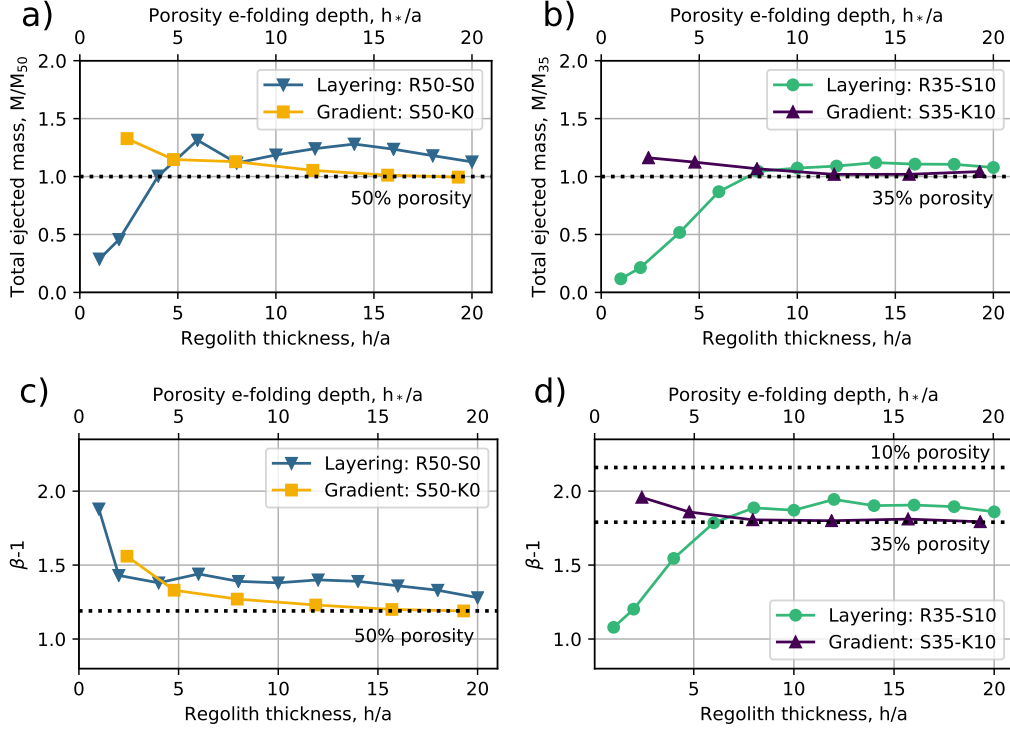


Figure 11: a) Total ejected mass,  $M$ , normalised by the ejected mass in the 50% porosity half-space scenario,  $M_{50}$ , as a function of regolith layer thickness,  $h/a$ , for R50-S0 and S50-K0 target scenarios. b) Total ejected mass normalised by the mass ejected in the 35% half-space scenario,  $M/M_{35}$ , as a function of regolith layer thickness,  $h/a$ , for R35-S10 and S35-K10 target scenarios. c) Normalised ejected momentum,  $\beta - 1$ , as a function of regolith thickness,  $h/a$ , for R50-S0 and S50-K0 target scenarios. d)  $\beta - 1$ , as a function of regolith thickness, for R35-S10 and S35-K10 target scenarios. In all figures, the upper label shows the equivalent porosity e-folding depth,  $h_*/a$ . The equivalent ejected mass and  $\beta - 1$  for the porous half-space scenarios, are plotted for comparison (dotted lines).

638 *4.2. Implications for the DART mission*

639 The DART mission will be the first to produce a controlled deflection  
640 of a near-Earth asteroid. DART will provide a great opportunity to test  
641 our capability to deflect a potentially hazardous asteroid and to calibrate  
642 numerical simulations of m-scale impacts on asteroids.

643 The numerical simulations presented here used impactor properties ap-  
644 propriate for the DART spacecraft and target scenarios consistent with our  
645 current understanding of DART’s target, Didymoon. In all of the target sce-  
646 narios simulated, the deflection caused by the DART impact was amplified  
647 by the escaping crater ejecta; i.e.,  $\beta - 1 > 0$ .

648 In this work we have shown that  $\beta - 1$  can vary significantly depending  
649 on the target asteroid’s properties and structure. If the DART spacecraft  
650 impacts a homogeneous porous target, we found that  $\beta - 1$  can have values  
651 between 0.5 and 3.5, depending on the target cohesion and porosity. On the  
652 other hand, if the DART spacecraft impacts a layered target, the deflection  
653 can be both amplified by up to 30% or reduced by up to 60%, compared  
654 with the equivalent homogeneous upper layer case. If the transition between  
655 the surface material and the substrate is gradual, then  $\beta - 1$  can be ampli-  
656 fied by up to 30% compared with the case of a homogeneous high-porosity  
657 subsurface.

658 Our results also show that some of our simulated impacts into different  
659 targets produced very similar  $\beta - 1$  values, yet very different crater sizes  
660 and morphologies. For example, the same deflection amplification ( $\beta - 1 \approx$   
661 1.9) was found when impacting homogeneous targets with cohesions and  
662 porosities of: 100 kPa and few percent porosity, 10 kPa and 20% porosity,

663 1 kPa and  $\approx 30\%$  porosity or 0.1 kPa and  $\approx 50\%$  porosity. Meanwhile, the  
664 same impacts produced craters that had radii of 5 m, 15 m, 23 m and 37 m,  
665 respectively.

666 The same amplification,  $\beta - 1 \approx 1.9$ , was also produced in different  
667 layered-target scenarios: An impact into a 50% porous regolith layer, one  
668 impactor radius thick, overlying a non-porous substrate produced a concen-  
669 tric crater with a radius of 4 m, while an impact into a regolith layer eight  
670 impactor radii thick, overlying a non-porous or few percent ( $< 10\%$ ) porous  
671 substrate, produced a flat-floor crater, with a radius of  $\approx 10$  m.  $\beta - 1 \approx 1.9$   
672 was also observed in the simulations of impacts into targets in which porosity  
673 decreased exponentially with depth from a porosity of 35% at the surface to a  
674 minimum subsurface porosity of 10% at a depth of  $\approx 1-2$  m, which produced  
675 bowl shaped craters with radii of about 10 m.

676 Therefore, before the arrival of Hera at Didymoon, which will be able to  
677 evaluate the outcome of the DART impact in great detail, measurements of  
678  $\beta-1$  alone will not be sufficient for the purpose of numerical model validation.  
679 Data collected by the Hera spacecraft will be used to construct a shape model  
680 of Didymoon and constrain its bulk density, as well as setting approximate  
681 limits on the surface cohesion by surveying the asteroid surface roughness  
682 and performing a CubeSat bouncing experiment (Küppers et al., 2019). In  
683 addition, high-resolution images of the DART impact crater will determine  
684 the crater size and morphology.

685 Validation of numerical models of impact deflection against a controlled  
686 impact scenario with a measured  $\beta - 1$  value is not only important for plan-  
687 etary defence, but also for inferring the subsurface structure of small aster-

688 oids. The DART crater morphology can provide insight into the surface and  
689 subsurface properties of Didymoon, whilst helping our understanding of the  
690 origins and formation processes of small near-Earth binary asteroids.

## 691 5. Conclusions

692 We conducted numerical simulations of the DART impact using the iSALE  
693 shock physics code. We investigated various different impact scenarios in the  
694 strength regime, to determine the sensitivity of crater morphology, ejecta  
695 mass-velocity distribution and momentum enhancement,  $\beta$ , to target prop-  
696 erties and structure. Targets with three different structures were investi-  
697 gated: homogeneous porous half-space, layered targets with a porous weak  
698 upper layer overlying a stronger bedrock layer and targets with porosity that  
699 decreased exponentially with depth.

700 For impacts into homogeneous porous half-space, we found that cohe-  
701 sion has the greatest influence on crater diameter, while both cohesion and  
702 porosity have a major influence on the ejected mass and subsequently, the  
703 amount of momentum transferred from the impact for deflection. For the  
704 DART impact, the expected deflection was calculated  $\beta - 1$  between 1 and  
705 3, implying a momentum multiplication factor of between 2 and 4.

706 For impacts into layered targets, we found that that the presence of a  
707 less porous, stronger lower layer within six impactor diameters of the surface  
708 can influence the crater morphology and the mass velocity distribution of  
709 ejecta. We showed that there is an amplification in the amount of ejected  
710 mass, caused by the proximity of the stronger substrate to the surface and  
711 that the change in mass and velocity of the material ejected is dependent on

712 the thickness of the upper layer. However, this amplification does not always  
713 result in an enhancement of ejected momentum relative to the homogeneous  
714 upper-layer case. Both amplification and reduction of ejected mass and mo-  
715 mentum relative to the homogeneous upper-layer case can occur in two-layer  
716 targets. This result has important implications for asteroid deflection. For  
717 the case of DART, momentum enhancement can change by up to 90%.

718       Impacts into targets with an exponentially decreasing porosity with depth  
719 produced an enhancement in the ejected mass and ejected momentum, but  
720 the effects were only considerable for relatively sharp decreases in porosity  
721 that occur within 6 m of the asteroid surface.

722       By measuring the DART crater, the Hera mission will provide vital data  
723 to test the predictive capabilities of numerical simulations of asteroid deflec-  
724 tion.



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<sup>1010</sup> **Appendix A. Appendix**

<sup>1011</sup> Additional supporting information (tables, model outcomes) will be pro-  
<sup>1012</sup> vided at the time of the publication as a DOI.